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# UNSTEADY TRANSONIC WIND TUNNEL TEST ON A SEMISPAN STRAKED DELTA WING MODEL OSCILLATING IN PITCH

Part 2: Selected Data Points for Harmonic Oscillation

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**National Aerospace Laboratory (NLR)** 

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106

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A wind tunnel investigation		investigate the unstead	dy aerodynamic aspects of
transonic high incidence flo	was conducted in 1992 to	ving model This test	was designed to show how
low speed vortex type flows	evolve into complicated s	shock vortex interacting	of flows at transonic speeds.
Requirements for this test w	vere based on a low speed	test conducted in 1986	on a full span model in the
NLR Low Speed Tunnel. T	he transpire model was a	semispan version of th	e low speed model with
some modifications. It was	acvinged with a three cor	nnonent semisnen hele	nce to measure total wing
some modifications. It was	equipped with a timee-cor	mponent semispan baia	receives and 15 vertical
loads, seven rows of high re	sponse pressure transduce	The model was see	pressures and 15 vertical
accelerometers to measure i	nodel motion and vibratio	is. The model was osc	cillated sinusoidally in pitch
at various amplitudes and fr	equencies for mean mode	incidences varying in	on 4 to 40 . In addition,
maneuver type transient mo	tions of the model were tes	sted with amplitudes of	16° and 30° total rotation at
various starting angles. The	test was conducted in the	NLR HS1 in the Macr	range of 0.225 to 0.90 with
some preliminary vapor scre	een flow visualization data	a taken at M=0.6 and 0	.9. This part of the report
presents selected data points	s for harmonic oscillations	<b>3.</b>	
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#### **FOREWORD**

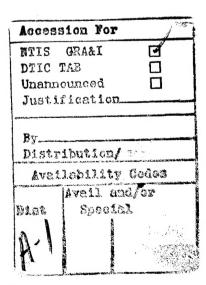
This report summarizes the results of an investigation into transonic high incidence unsteady aerodynamics. Transonic wind tunnel tests were conducted for a semispan straked delta wing model oscillating in pitch at high incidences.

This test was conducted under a cooperative program of research between the Lockheed Fort Worth Company (LFWC), Fort Worth, Texas, USA (Formerly the Fort Worth Division of General Dynamics until 28 February 1993) and the National Aerospace Laboratory (NLR), Amsterdam, The Netherlands. The test was conducted in April/May 1992. The models and corresponding support system were designed at NLR under an earlier separate program with funding from the Fort Worth Division of General Dynamics (now LFWC) and NLR. The fabrication of models, test preparation, wind tunnel test and reporting were performed at NLR under a subcontract from LFWC. This work was funded under Air Force contract F33657-84-C-0247 (CCP 4551) for the Aeronautical Systems Center, Wright-Patterson Air Force Base, Ohio. The work was administered by Mr. F. Zapata of the F-16 SPO (ASC/YPEF) and Mr. L. J. Huttsell of the Flight Dynamics Directorate of Wright Laboratory (WL/FIBG) Wright-Patterson Air Force Base, Ohio under work unit 24010292.

The program manager was Dr. A. M. Cunningham, Jr. at LFWC. The principal investigators were Dr. Cunningham at LFWC, Mr. R. G. den Boer for the wind tunnel test programs at NLR. Mr. den Boer was assisted by the following NLR specialists: C. D. G. Dogger, E. G. M. Geurts, A. P. Retel and R. J. Zwaan. The authors would like to acknowledge the following person who contributed substantially in the project:

Mr. E. W. M. Slijkerman for the design of the wind tunnel model and support; Messrs. T. Horsman and F. Hofman for the fabrication and instrumentation of the model; and Messrs. O. van Teunenbroek and A. van der Kamp for their contribution in the software development.

The test program for the straked delta wing is documented in three separate reports. This report (Part 2) contains selected test points for harmonic oscillation. Part 1 contains a description of the model, test setup, data acquisition, and data processing. Part 3 contains selected data for simulated maneuvers. Parts 1 and 3 are published as WL-TR-94-3094 and WL-TR-94-3096, respectively. An overview of this test program is also published as WL-TR-94-3017.



# TABLE OF CONTENTS

		Page
APPENDIX A	Prints and Plots of the Mean and First Five Harmonic Components of the Unsteady Data of Data Points 151, 358, 375, 593 and 605	1
APPENDIX B	Plots of Normal Force and Moment Coefficients Versus Incidence	102

#### **SUMMARY**

An unsteady transonic wind tunnel test was conducted for a highly instrumented semispan simple straked delta wing model. Harmonic pitch oscillation as well as maneuver simulation were performed.

Appendix A of this report presents the mean and first five harmonic components of the unsteady data of data points 151,358,375,593 and 605 of the test on the model oscillating harmonically in pitch in tabulated and plotted form. These data are also available on a floppy disk. The rationale for selecting these data points is discussed in Section 7 of Part 1 (WL-TR-94-3094). Table 4 in Part 1 provides a convenient cross-reference of conditions and data point numbers. For nondimensionalization, symbols and definitions the reader is referred to Part 1 (see the section starting on page ix of Part 1).

As a reference, Appendix B presents plots of normal force and moment coefficients versus incidence. In these plots, results of a large number of data points have been collected. Data of the SiS configuration WITHOUT filler plate are only presented in the Figures B.1 to B.5; all other figures contain only data of the SiS configuration WITH filler plate.

APPENDIX A Prints and Plots of the Mean and First Five Harmonic Components of the Unsteady Data of Data Points 151, 358, 375, 593 and 605

Unsteady Transonic Delta Program

BALANCE	LOADS	aerodynamic coefficients		aero	angular defl	ections [deg]		
position	comp.	Zero	Re 1	Im 1	inertia [%]	Zero	Re 1	Im 1
main	CN Cm Cl	1.09156 0.08135 37659	2.14309 0.29350 50772	0.61275 04218 26108	2364.58 201.95 625.90	056 063	037 004	0.002

ACCELE	ERATIONS				vibration	mode		
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	75.286 35.066 28.761	2.197 3.471 -178.363	1	2.878	1.790	7.946
21 22	-138.6 -46.6	-116.9 -116.9	24.535	16.071	2	28.034	1.208	8.353
12 13 21 22 23 31 32 33 41	121.4 -74.6 -10.6	-116.9 -189.9 -189.9	24.104 8.681	-167.130 18.730	3	45.540	2.749	7.223
33 41 42 43	141.4 29.4 89.4	-189.9 -304.9 -304.9	26.302 3.384 17.520	-168.691 -172.471 -178.576 -178.495	4	73.118	0.588	8.675
43 51 52 53	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	27.168 15.733 22.863 29.896	-178.495 -166.775 -163.986 -164.250	5	89.904	1.179	8.758

Unsteady Transonic Delta Program

PRESSURES	section		00.65 mm 09.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
101 102 103 104	2.00 5.00 10.00 15.00	-1.572 -1.621 -1.557 -1.987	0.899 1.512 2.213 4.878	972 885 -1.058 -1.166
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 82.50 85.00 90.00 95.00 10.00 40.00 60.00 80.00	-1.117878775756723663615574521463 0.509 0.336 0.164	-2.346 -2.855 -2.952 -3.090 -2.848 -2.535 -2.426 -2.331 -2.240 -2.179 0.889 1.010 0.889 0.158	-1.420 -1.111 989 913 752 393 177 009 0.280 0.472 0.180 0.241 0.292 0.295

PRESSURES	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
201 202 203 204 205 206 207 208 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	913 918 928 894 879 773 7668 596 556 556 556 0.601 0.495 0.330 0.101	1.992 2.009 2.178 2.310 1.119 2.697 2.012 0.686 331 -1.636 -1.674 -1.878 -2.116 0.765 0.879 0.741 0.452 012	914 -1.005 -1.034948 -1.667942827814827696637540389 0.217 0.259 0.273 0.227

PRESSURES section 3				194.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	523 528 522 513 506 513 528 560 537 538 554	1.506 1.477 1.463 1.353 0.453 0.903 0.339 051 248 468 692 931	653 653 702 698 -1.029 636 536 440 390 466 594 676

PRESSURES section 4				.44.42 mm 195.32 mm
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00	300 308 305 315 324 349 378 389 399 396 405	0.666 0.628 0.525 0.469 0.416 201 586 880 -1.025 -1.078	550 552 516 523 491 433 326 235 221 248 280

Unsteady Transonic Delta Program

PRESSURES section 5 b = 82.70 mm x =-269.60 mm				
nr.	y/b [%]	Cp 0	ReCp 1	ImCp 1
501 502 503 504 505 506 507 508 509	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	440 574 835 -1.298 -1.541 -1.686 -1.645 -1.419 -1.124	-1.518 -2.557 -4.215 -5.928 -6.029 -5.240 -4.189 -3.535 -3.077 -2.990	267 119 0.156 0.427 0.374 0.138 141 326 290 316

PRESSURES section 6				233.73 mm -60.62 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.568 -1.771 -1.731 -1.556 -1.258 -1.233 -1.404 -1.965 -2.647 -1.874 -1.621	-3.997 -4.141 -4.658 -5.500 -5.123 -6.098 -6.762 -4.665 4.103 3.100 1.512	031 364 575 593 664 728 944 -1.308 -1.041 957 885

PRESSURES section 7			17.90 mm .00.71 mm
y/b [%]	Cp 0	ReCp 1	ImCp 1
22.71 28.21 33.72 39.26 44.69 50.03 55.28 60.46 65.56 70.59 75.54 80.42 85.22 90.19 94.60	036 312 7312 756 756 700 773 689 613 528 443 315	0.673 0.592 - 192 -2.468 -3.140 -3.090 -2.560 -957 0.686 0.802 0.672 0.339 0.305 0.340	516 066 0.370 0.079 631 913 945 960 814 668 517 513 513 513
	y/b [%] 22.71 28.21 33.26 44.69 55.28 60.46 65.56 70.59 75.54 80.42 85.22 90.19	y/b Cp 0 [%]  22.71036 28.21312 33.72778 39.26891 44.69870 50.03756 55.28700 60.46752 65.56773 70.59689 75.54613 80.42528 85.22443 90.19374	X = 1   Y/b

SECTION COEFFICIENTS						
section	comp.	Zero	Re 1	Im 1		
1	CN_u CN_1 CN_t CM_u Cm_u	0.976 0.319 1.295 120 026	1.357 0.627 1.984 803 065	0.770 0.261 1.031 061 072		
2	CE_11 CN_11 CN_12 CN_12 CE_E	146 0.733 0.295 1.029 140 017	868 426 0.482 0.056 342 024 366	133 0.854 0.250 1.104 152 062 214		
3	CN_u Cm_u	0.507 118	118 199	0.604		
4	CN_u Cm_u	0.341	0.412	0.370 060		
5	CN_u Cl_u	0.992	3.683	0.051		
6	CN_u	1.531	3.071	0.489		
7	Cl_u CN_u Cl_u	745 0.464 266	-1.060 0.271 140	355 0.512 282		

test conditions

Simple Strake configuration

alpha = 22.109 deg | Q = 6.690 kPa Mach = 0.225 | Ptot = 195.256 kPa Re\*10^-6 = 7.982 | Ttot = 291.828 K

dalpha = 8.342 deg freq = 5.700 Hz k = 0.192 harm = 2

BALANCE LOADS aerodynamic coefficients		aero	angular defl	ections [deg]				
position	comp.	Zero	Re 2	Im 2	inertia [%]	Zero	Re 2	Im 2
main	CN Cm Cl	1.09156 0.08135 37659	51877 00052 0.19050	~.13902 00582 0.05409	****** 769.37 *****	056 063	0.001 0.005	0.001 0.001

ACCELERATIONS					vibration	mode			
nr	[mm]	[mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]	
11 12	-425.6 -215.6	-12.0 -12.0	3.224 3.505	-136.668 -147.962	1	2.878	0.297	0.072	
11 11 12 12 12 12 12 12 12 13 13 14 14 15 15 15 15 15 15 15 15 15 15 15 15 15	167.4 -138.6 -46.6	-12.0 -116.9 -116.9	1.397 0.447	40.046 -125.017	2	28.034	0.047	0.035	
23 31	121.4 -74.6 -10.6	-116.9 -189.9 -189.9	0.486 0.292	115.696 -143.511	3	45.540	0.050	0.044	
33 41	141.4 29.4	-189.9 -304.9	0.703 0.339 0.275	107.313 88.281 80.138	4	73.118	0.048	0.008	
42 43 51	89.4 152.4 85.0	-304.9 -304.9 -374.9	0.431 0.366	84.757 98.551					
52 53	121.4 157.4	-374.9 -374.9	0.189 0.248	142.441 -160.389	5	89.904	0.172	0.074	

Unsteady Transonic Delta Program

PRESSURE	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
101 102 103 104	2.00 5.00 10.00 15.00	-1.572 -1.621 -1.557 -1.987	1.204 1.484 1.130 1.105	0.148 0.474 0.068 267
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 90.00 95.00 10.00 20.00 40.00 80.00	-1.1178787757757757663615574521463 0.619 0.509 0.336 0.247 0.164	2.337 1.891 1.610 1.754 1.805 1.747 1.619 1.338 0.973 076 079 072 065 044	2.246 1.462 0.950 0.853 1.214 1.427 1.388 1.267 0.915 0.461 089 062 034 034

PRESSURE	PRESSURES section 2 $c = 246.21 \text{ mm}$ y = -273.97  mm							
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2				
201 202 203 204 205 206 207 208 2109 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 18.00 30.00 40.00 50.00 60.00 70.00 82.50 85.00 90.00 95.00 20.00 40.00 80.00	- 913 - 918 - 929 - 879 - 879 - 843 - 779 - 7668 - 556 - 5556 - 5556 0 601 0 495 0 0 224 0 101	0.344 0.411 0.577 0.453 0.172 0.189 0.176 0.299 0.609 0.717 0.743 0.858 1.008 082 082 082	0.130 0.131 0.233 0.112 153 350 515 341 0.190 0.274 0.326 0.437 0.550 091 067 031 002 0.058				

PRESSURES	section		194.13 mm 336.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
301 302 303 304 305 306 307 308 309 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	523 528 522 513 506 513 528 560 537 538 554 549	0.199 0.164 0.132 0.093 072 166 508 704 650 459 245 0.020	100 119 149 182 094 400 589 674 610 549 464 294

PRESSURES section 4 $c = 144.42 \text{ mm}$ y = -395.32  mm					
nr.	x/c [%]	Cp 0	ReCp 2	ImCp 2	
401 402 403 404 405 406 407 408 409 410 411	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	300 308 305 315 324 349 378 389 399 396 405	0.043 0.003 060 095 122 265 415 571 627 551 436 340	316 331 337 360 376 451 513 544 449 362 311 236	

Unsteady Transonic Delta Program

PRESSURES	Section		82.70 mm 269.60 mm	
nr. up	y/b [%]	Cp 0	ReCp 2	ImCp 2
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	440 574 835 -1.298 -1.541 -1.686 -1.645 -1.419 -1.124 -1.123	0.139 0.108 0.155 0.478 0.615 0.539 0.323 0.238 0.330 0.283	0.012 108 144 0.013 0.104 0.091 0.000 016 0.038 0.026

PRESSURES	section	n 6		233.73 mm -60.62 mm
nr.	y/b [%]	Cp 0	ReCp 2	ImCp 2
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.568 -1.771 -1.731 -1.556 -1.233 -1.404 -1.965 -2.647 -1.874 -1.621	0.706 0.784 0.906 1.081 0.605 0.952 1.730 3.126 2.191 1.282 1.484	0.146 0.217 0.339 0.380 163 0.078 1.006 2.817 0.829 472 0.474

PRESSURES	section		117.90 mm 100.71 mm	
nr.	y/b [%]	Cp 0	ReCp 2	ImCp 2
701 702 703 704 705 109 706 707 208 708 709 307 711 405	22.71 28.21 33.726 44.69 55.03 55.28 60.46 65.56 70.59 75.54 80.42 85.29 90.19 94.60	036 312 778 891 870 756 702 773 689 613 528 443 315	267 058 1.294 2.418 2.029 1.550 1.311 0.827 0.160 419 558 508 330 220 122	448350 1.098 2.245 1.505 0.853 0.688 0.193515761693589376

SECTION O	COEFFICIE	INTS		
section	comp.	Zero	Re 2	Im 2
1	CN_u	0.976 0.319	-1.595 064	982 044
	CN_1 CN_t	1.295	-1.659	-1.026
	Cm_u	120	0.391	0.289
	Cm_l Cm_t	026 146	0.012	0.006 0.295
2	CN_u	0.733	422	0.038
	CN_1	0.295 1.029	021 443	013 0.025
	CN_t Cm_u	140	0.156	0.025
	Cm_1	017 157	012 0.144	013 0.012
3	Cm_t CN_u	0.507	0.251	0.405
	Cm_u	118	084	125
4	CN_u Cm_u	0.341	0.354	0.386
5	CN_u	0.992	285	0.013
6	Cl_u CN_u	553 1.531	0.164	0.004
	Cl_u	745	0.658	0.223
7	CN_u Cl_u	0.464	335 0.138	025 023
		.200	3.130	

test conditions

alpha = 22.109 deg | Q = 6.690 kPa
Mach = 0.225 | Ptot = 195.256 kPa
Re\*10^-6 = 7.982 | Ttot = 291.828 K

dalpha = 8.342 deg freq = 5.700 Hz k = 0.192 harm = 3

BALANCE	BALANCE LOADS aerodynamic coefficients		aero	angular defl	ections [deg]			
position	comp.	Zero	Re 3	Im 3	inertia [%]	Zero	Re 3	Im 3
main	CN Cm Cl	1.09156 0.08135 37659	03052 00290 0.00361	0.02867 02506 02478	945.44 365.75 584.27	056 063	0.000	0.002

ACCEL	ERATIONS				vibration mode			
nr	× [mm]	y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	2.109 1.296 0.917	-94.531 -117.082 77.704	1	2.878	0.031	0.025
12 13 21 22 23 31 32 33 34 44 42 43 55 55 55 55	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.501 0.714	-92.601 93.022	2	28.034	0.013	0.023
31 32	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	0.233	-78.844 100.266	3	45.540	0.012	0.024
41 42	29.4 89.4 152.4	-304.9 -304.9 -304.9	0.205 0.471 0.539	160.524 82.926 92.783	4	73.118	0.012	0.020
51 52	85.0 121.4 157.4	-374.9 -374.9 -374.9	0.418 0.668 1.060	86.025 88.596 92.720	5	89.904	0.035	0.044

PRESSURES	section		00.65 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
101 102 103 104	2.00 5.00 10.00 15.00	-1.572 -1.621 -1.557 -1.987	0.036 0.109 093 210	0.499 0.207 0.293 148
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 90.00 95.00 10.00 40.00 60.00 80.00	-1.117878775756723663574521463 0.619 0.336 0.247 0.164	0.041 215 185 161 182 0.082 0.276 0.615 006 002 0.001 0.001	0.347 0.449 0.436 0.466 0.138 134 328 661 855 0.000 009 015 027

PRESSURES	section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 70.00 79.00 82.50 90.00 95.00 90.00 95.00 90.00 95.00 80.00 80.00	913 918 929 894 879 773 773 7668 566 5569 5552 0.6011 0.4330 0.224	095 123 136 097 118 160 115 035 075 0.000 0.042 0.065 0.140 0.207 0.001 0.004 0.004	186165202269021511956941483 0.059 0.081 0.062 0.002065 0.001005013024054

PRESSURES section 3			c = 194.13 mm y =-336.06 mm		
nr. up	x/c [%]	Cp 0	ReCp 3	ImCp 3	
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 79.00 90.00	523 528 522 513 506 513 528 560 537 538 554	0.009 0.019 0.040 0.036 072 0.034 0.004 043 035 018 018 089	111 143 174 203 054 386 519 385 0.077 0.174 0.243	

PRESSURES	section		144.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 3	ImCp 3
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 90.00	300 308 305 315 315 324 349 378 389 399 396 405	0.228 0.226 0.210 0.224 0.227 0.193 0.125 0.012 064 053 013 015	198225251279294318281192 0.003 0.169 0.251 0.307

Unsteady Transonic Delta Program

PRESSURE	S section		82.70 mm 269.60 mm	
nr. up	y/b [%]	Cp 0	ReCp 3	ImCp 3
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	440 574 835 -1.298 -1.541 -1.686 -1.645 -1.419 -1.124 -1.123	009 0.002 0.015 0.029 0.036 0.052 0.080 0.101 0.036	033 0.032 0.131 0.200 0.172 0.156 0.212 0.266 0.132 0.065

PRESSURES section 6				233.73 mm -60.62 mm
nr.	y/b [%]	Cp 0	ReCp 3	ImCp 3
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.568 -1.771 -1.731 -1.556 -1.258 -1.233 -1.404 -1.965 -2.647 -1.874 -1.621	0.032 0.069 0.084 0.013 071 281 417 0.005 0.645 051	0.086 0.179 0.229 0.250 0.517 0.659 0.717 0.240 0.098 0.998

PRESSURES section 7				17.90 mm 00.71 mm
nr.	[%]	Cp 0	ReCp 3	ImCp 3
701 702 703 704 705 109 706 707 208 709 708 709 710 711 405	22.71 28.21 33.72 39.26 44.69 50.03 60.46 65.56 65.56 75.54 80.42 85.22 90.19 94.60	036 312 778 891 750 750 753 689 613 443 315	0.130 0.278 0.513 0.131 215 161 038 0.082 035 045 0.004 0.081 0.195	289306475 0.357 0.875 0.430176627941851653519379311294

SECTION COEFFICIENTS						
section	comp.	Zero	Re 3	Im 3		
2	CN_u CN_1 CN_t CN_t CN_l CN_l CN_l CN_l CN_l CN_l CN_l	0.976 0.319 1.295 120 026 146 0.733 0.295 1.029 140 017	017 0.002 015 0.050 002 0.048 0.045 0.010 0.055 0.014 005	140 032 172 025 0.016 009 0.345 024 0.321 059 0.012		
3 4 5 6 7	Cm_1 Cm_t Cm_u CN_u CN_u CN_u CN_u CN_u CN_u CN_u C1_u C1_u	0157 0.507 118 0.341 087 0.992 553 1.531 745 0.464 266	0.009 0.007 023 073 011 028 0.019 020 0.013 089 0.036	047 0.113 0.019 0.055 0.051 112 0.069 268 0.172 0.271 163		

Simple Strake configuration

test conditions

alpha = 22.109 deg | Q = 6.690 kPa
Mach = 0.225 | Ptot = 195.256 kPa
Re\*10^-6 = 7.982 | Ttot = 291.828 K

dalpha = 8.342 deg freq = 5.700 Hz k = 0.192 harm = 4

BALANCE	LOADS	aerodynamic coefficients		aero	angular defl	ections [deg]		
position	comp.	Zero	Re 4	Im 4	inertia [%]	Zero	Re 4	Im 4
main	CN Cm C1	1.09156 0.08135 37659	06407 0.00515 0.02299	0.03904 00362 01616	6885.83 370.80 2665.55	056 063	0.000	0.000

ACCELE	RATIONS				vibration	n mode		
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	0.585 0.379 0.481	-12.457 -5.627 157.183	1	2.878	0.004	0.005
21	-138.6 -46.6	-116.9 -116.9 -116.9	0.282	-30.727 140.103	2	28.034	0.003	0.007
31 32	121.4 -74.6 -10.6	-189.9 -189.9	0.071	44.611	3	45.540	0.006	0.005
13 21 22 23 31 32 33 41 42 43 51 52	141.4 29.4 89.4	-189.9 -304.9 -304.9	0.335 0.174 0.257	143.290 -122.030 135.162	4	73.118	0.007	0.008
43 51 52	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	0.295 0.126 0.290 0.488	143.568 111.927 148.646 146.496	5	89.904	0.018	0.015

Unsteady Transonic Delta Program

PRESSURES	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
101 102 103 104	2.00 5.00 10.00 15.00	-1.572 -1.621 -1.557 -1.987	096 0.084 0.054 0.047	227 341 249 624
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 80.00	-1.117 878 775 756 723 663 574 521 463 0.619 0.509 0.336 0.247	0.381 0.547 0.627 0.545 0.604 1.020 0.933 0.569 0.014 003 009 016 027 062	233 180 244 315 339 354 361 357 338 243 0.002 0.005 0.005 0.009

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 50.00 60.00 79.00 82.50 90.00 90.00 95.00 10.00 20.00 40.00 80.00	- 913 - 918 - 929 - 894 - 879 - 773 - 773 - 720 - 668 - 566 - 556 - 5569 - 556 0 0.495 0 0.224 0 101	0.092 0.071 0.125 0.067 0.001 016 0.074 0.021 211 0.054 0.165 0.229 0.350 0.430 008 013 017 019 006	114 106 171 070 024 0.108 045 141 130 189 248 278 344 404 0.007 0.004 026

PRESSURES section 3				.94.13 mm 36.06 mm
nr. up	x/c [%]	Ср 0	ReCp 4	ImCp 4
301 302 303 304 305 306 307 308 309 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	523 528 522 513 506 513 528 560 537 538 554	002 005 0.004 003 0.044 008 054 249 326 285 193 040	0.196 0.196 0.200 0.171 0.017 0.052 0.004 0.140 0.315 0.348 0.250

PRESSURES	S section		144.42 mm 395.32 mm	
nr.	x/c [%]	Cp 0	ReCp 4	ImCp 4
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	300 308 305 315 315 324 349 378 389 399 396 405	0.069 0.062 0.049 0.050 0.041 049 149 263 321 282 201 120	072 082 088 093 097 111 100 073 0.031 0.113 0.093 0.048

DPN = 151

Unsteady Transonic Delta Program

PRESSURES section 5 $b = 82.70 \text{ mm}$ x = -269.60  mm				
y/b [%]	Cp 0	ReCp 4	ImCp 4	
6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	440 574 835 -1.298 -1.541 -1.686 -1.645 -1.419 -1.124	0.015 0.012 0.031 0.030 0.001 032 037 005 0.000	001 021 040 031 0.011 0.053 0.049 009 016 0.007	
	y/b [%] 6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72	y/b [%]  6.62440 20.43574 34.05835 47.67 -1.298 54.49 -1.541 61.29 -1.686 68.10 -1.645 74.91 -1.419 81.72 -1.124	X = -2   Y/b   Cp 0   ReCp 4   Recp 4	

PRESSURES	S section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 4	ImCp 4
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.568 -1.771 -1.731 -1.558 -1.233 -1.404 -1.965 -2.647 -1.874 -1.621	017 001 0.007 0.029 004 0.051 0.210 0.211 626 0.137 0.084	0.054 005 060 063 087 074 054 044 602 465 341

PRESSURES	S section		17.90 mm .00.71 mm	
nr.	y/b [%]	Cp 0	ReCp 4	ImCp 4
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.72 44.69 50.03 55.28 60.46 65.56 70.59 75.54 285.22 90.19 94.60	036 312 732 871 871 756 700 755 773 689 613 528 443 374 315	0.115 0.028 085 0.624 0.589 0.545 0.438 0.168 055 055 054 032 0.009 0.041	030 0.127 0.321 037 187 385 380 381 141 0.179 0.110 0.004 0.000 033 097

SECTION C	SECTION COEFFICIENTS							
section	comp.	Zero	Re 4	Im 4				
1	CN_u CN_1 CN_t	0.976 0.319 1.295	441 029 470	0.313 0.010 0.323				
	Cm_u	120	0.161	074 004				
	Cm_l Cm_t	026 146	0.175	078				
2	CN_u	0.733	062	0.133				
	CN_I	0.295	012	007				
	CN_t Cm_u	1.029	074 0.033	0.127				
	Cm_1	017	0.003	0.005				
	Cm_t	157	0.036	056				
3	CN_u	0.507 118	0.113	158 0.048				
4	CM_u CN_u	0.341	0.132	0.020				
-	Cm_u	087	058	0.015				
5	CN_u	0.992	005 001	0.005				
6	Cl_u CN_u	553 1.531	006	0.093				
	Cl_u	745	0.008	091				
7	CN_u Cl_u	0.464	146 0.058	0.053				
	u	200						

BALANCE	LOADS	aerodynamic	erodynamic coefficients		aero	angular deflections [deg]		
position	comp.	Zero	Re 5	Im 5	inertia [%]	Zero	Re 5	Im 5
main	CN Cm C1	1.09156 0.08135 37659	0.02322 00367 01128	00886 00113 0.00566	1111.01 110.07 583.13	056 063	0.000	0.000

ACCEL	ACCELERATIONS			vibration mode				
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13 21 22 23 31 32 33	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	0.924 0.778 0.645	92.656 63.782 -122.223	1	2.878	0.006	0.004
21 22 23	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.311	67.978 -106.055	2	28.034	0.000	0.004
31 32 33	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	0.107	55.643 -104.762	3	45.540	0.002	0.004
41 42 43 51 52	29.4 89.4 152.4	-304.9 -304.9 -304.9	0.169 0.314 0.398	-80.975 -119.600 -114.170	4	73.118	0.004	0.004
51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	0.327 0.316 0.303	-110.443 -96.919 -71.002	5	89.904	0.016	0.005

Unsteady Transonic Delta Program

DPN = 151

PRESSURES	section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
101 102 103 104	2.00 5.00 10.00 15.00	-1.572 -1.621 -1.557 -1.987	0.095 048 005 0.193	0.175 0.120 0.113 0.177
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 80.00	-1.117 878 775 756 723 663 615 574 521 463 0.619 0.509 0.336 0.247 0.164	0.260 0.159 0.094 0.028 0.070 058 207 313 517 644 003 002 003	0.078 0.089 0.076 0.091 015 230 655 655 005 005 006

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
201 202 203 204 205 206 207 208 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	- 913 - 918 - 929 - 894 - 879 - 843 - 773 - 7720 - 668 - 596 - 556 - 556 - 556 0 601 0 495 0 330 0 1224 0 101	133 104 126 196 0.001 196 362 372 044 063 156 259 001 0.000 005 001	0.047 0.053 0.093 0.099 0.065 0.090 0.115 0.014 019 0.165 0.142 0.094 013 140 005 007 008 008

PRESSURES	section		194.13 mm 336.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	523 528 522 513 506 513 528 560 537 538 554	014 0.009 0.041 0.069 0.018 0.090 0.051 090 034 0.105 0.189 0.179	0.080 0.071 0.068 0.062 015 021 094 096 019 0.023 0.070 0.086

PRESSURES section 4				144.42 mm 395.32 mm
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	300 308 305 315 315 324 349 378 389 399 396 405	0.088 0.086 0.072 0.070 0.064 0.031 0.014 0.008 0.005 0.083 0.143 0.168	110 116 124 138 151 159 136 076 0.057 0.149 0.129

Unsteady Transonic Delta Program

PRESSURES section 5				82.70 mm 269.60 mm
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	440 574 835 -1.298 -1.541 -1.686 -1.645 -1.419 -1.124	029 033 024 010 0.007 0.036 0.066 0.090 0.042	0.003 0.004 0.009 0.013 0.013 0.009 0.004 0.002 0.006 0.009

PRESSURE	PRESSURES section 6			233.73 mm -60.62 mm
nr. up	y/b [%]	СрО	ReCp 5	ImCp 5
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.568 -1.771 -1.731 -1.556 -1.258 -1.233 -1.404 -1.965 -2.647 -1.874 -1.621	060 066 069 079 030 109 129 0.580 0.092 048	013 011 006 019 0.034 0.036 0.030 0.026 0.549 0.205

nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
701 702 703 704 705 109 706 707 208 708 708 709 710 405	22.71 28.21 33.72 39.26 44.69 50.03 55.28 60.46 65.56 70.59 75.54 85.22 90.19 94.60	036 312 778 891 756 700 752 773 689 613 528 443 315	0.139 0.172 0.086 200 0.142 0.028 054 188 398 194 0.036 0.051 0.084 0.094	0.064 0.087 017 081 0.132 0.006 055 011 0.014 025 089 094 073 115

SECTION COEFFICIENTS						
section	comp.	Zero	Re 5	Im 5		
2	CN_u CN_1 CN_t CN_t Cm_u Cm_1 Cm_t CM_u CN_1 CN_1 CN_1 CN_1 CM_u CM_1 CM_U CM_1	0.976 0.319 1.295 120 026 146 0.733 0.295 1.029 140 017 157	0.008 012 003 056 0.006 050 0.210 020 0.191 057 0.011 046	0.035 011 0.024 067 0.004 062 052 017 069 0.005 0.007 0.013		
4 5 6 7	CM_u CM_u CM_u CN_u C1_u CN_u C1_u C1_u C1_u C1_u	118 0.341 087 0.992 553 1.531 745 0.464 266	0.031 072 0.027 004 0.010 0.034 008 023 003	0.007 0.020 0.026 007 0.004 050 0.045 0.010 024		

test conditions

Simple Strake configuration

alpha = 10.031 deg | Q = 17.267 kPa Mach = 0.601 | Ptqt = 87.220 kPa Re\*10^-6 = 8.024 | Ttot = 300.608 K

dalpha = 4.167 deg freq = 5.700 Hz k = 0.073 harm = 1

BALANCE	LOADS aerodynamic coefficients		aero	angular deflections [deg]				
position	comp.	Zero	Re 1	Im 1	inertia [%]	Zero	Re 1	Im 1
main	CN Cm Cl	0.56090 0.02249 22050	3.05619 0.18782 -1.03618	0.20800 01030 09754	8386.65 330.62 2944.68	049 105	031 030	0.001 004

ACCEL	ERATIONS				vibration mode			
nr	× [mm]	[mm] A	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13 21 22 23 31 32 33 41 42 43	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	39.541 20.007 15.211	1.436 1.583 -178.370	1	2.878	0.100	4.129
21 22 23	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	12.074	15.552 -164.672	2	28.034	0.040	3.877
31 32	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	3.708 7.532	13.106	3	45.540	0.177	2.324
41 42	29.4 89.4 152.4	-304.9 -304.9 -304.9	1.513 7.343 12.448	-177.722 -174.598 -178.089	4	73.118	0.658	3.977
51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	8.560 12.107 15.893	-164.652 -162.969 -162.982	5	89.904	0.375	4.530

Unsteady Transonic Delta Program

PRESSURES section 1				300.65 mm 209.06 mm
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
101 102 103 104	2.00 5.00 10.00 15.00	-1.319 -1.226 -1.189 -1.179	-2.035 -4.247 -6.699 -8.611	312 300 249 245
105 106 107 108 110 1110 1111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 80.00	511 404 351 284 221 157 119 092 048 006 0.363 0.243 0.114 0.092 0.118	-4.355 -2.552 -1.865 -1.139 925 925 881 850 794 713 1.795 1.542 1.133 0.689 0.206	074 050 092 112 092 057 049 041 024 003 0.100 0.121 0.137 0.134

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 70.00 79.00 82.50 90.00 90.00 95.00 10.00 40.00 40.00 80.00	-1.019962887831823747653557439361278258229130 0.362 0.248 0.135 0.105	2.406 1.152 835 -2.090 325 -3.625 -4.196 -3.631 -3.047 -3.195 -3.027 -2.824 1.668 1.461 0.952 0.508 078	515 516 504 360 0.646 469 346 161 0.019 0.123 0.109 0.126 0.1241 0.120 0.133 0.130

PRESSURES section 3				.94.13 mm 36.06 mm
nr.	x/c [%]	Cp 0	ReCp 1	ImCp 1
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00	783 755 700 650 618 554 504 472 414 372 331 273	3.432 2.702 1.364 0.214 258 -1.283 -1.685 -2.054 -2.227 -2.404 -2.601 -3.038	510 521 531 485 462 401 373 357 311 257 220 128

PRESSURES section 4				144.42 mm 395.32 mm
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	634 608 544 514 494 413 376 346 316 288 260	3.367 2.572 1.312 0.390 031 998 -1.308 -1.376 -1.341 -1.348 -1.445 -1.812	318326309306299271253250257271244212

Unsteady Transonic Delta Program

PRESSURES section 5				82.70 mm 269.60 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	136 134 148 215 304 443 574 580 449 443	-1.187 -1.213 -1.544 -3.026 -4.609 -6.082 -5.482 -2.432 -2.730	056 059 042 0.074 0.196 0.309 0.232 141 132

PRESSURES section 6				233.73 mm -60.62 mm
nr.	y/b [%]	Cp 0	ReCp 1	ImCp 1
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	674 772 580 505 419 382 464 -1.280 -1.369 -1.269	-5.291 -3.838 -2.881 -2.407 -1.709 -1.201 -3.702 -15.463 -11.865 -8.247	0.210 180 400 147 208 132 164 310 167 172 300

PRESSURE	PRESSURES section 7			17.90 mm .00.71 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.72 39.26 44.69 50.03 55.28 60.46 70.59 75.54 80.42 85.22 85.22 89.19 94.60	2214012758279284317421557532504501494494	595 -2.434 -1.528 -1.246 -1.219 -1.139 -2.141 -3.682 -4.196 -2.202 -1.685 -1.192 -740031	071 0.154 040 0.008 089 112 036 0.187 161 304 395 373 363 346 299

SECTION COEFFICIENTS						
section	comp.	Zero	Re 1	Im 1		
1	CN_u CN_1 CN_t Cm_u Cm_1	0.481 0.155 0.636 008 012	2.834 0.927 3.761 142 059	0.111 0.122 0.233 007 031		
2	Cm_t CN_u CN_l CN_t Cm_u Cm_l	020 0.545 0.158 0.703 058 011	201 2.666 0.738 3.404 941 0.007	038 0.094 0.125 0.219 0.021 030		
3	Cm_t CN_u Cm_u	0.462	1.411	0.334		
4	CN_u Cm_u	0.380	0.770 495	0.263		
5	CN_u C1_u	0.279	2.660 -1.557	011 0.010		
6	CN_u	0.710	5.106	0.046		
7	Cl_u CN_u Cl_u	0.353	1.510	0.132		

test conditions

Simple Strake configuration

 alpha
 =
 10.031 deg
 Q
 =
 17.267 kPa

 Mach
 =
 0.601
 Ptot
 =
 87.220 kPa

 Re\*10^-6
 =
 8.024
 Ttot
 =
 300.608 K

dalpha = 4.167 deg freq = 5.700 Hz k = 0.073 harm = 2

BALANCE	LOADS	aerodynamic	coefficients		aero angular		eflections [deg]		
position	comp.	Zero	Re 2	Im 2	inertia [%]	Zero	Re 2	Im 2	
main	CN Cm Cl	0.56090 0.02249 22050	03536 00129 0.03330	05546 00012 0.06695	****** 273.49 *****	049 105	0.000	0.000	

ACCELI	ERATIONS				vibration mode			
nr	× [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	0.353 0.206 0.118	-160.915 -151.976 16.915	1	2.878	0.008	0.009
21 22 23 31	-138.6 -46.6	-116.9 -116.9	0.274	-132.925 -40.210	2	28.034	0.025	0.012
23 31 32 33	121.4 -74.6 -10.6	-116.9 -189.9 -189.9	1.073	-109.743	3	45.540	0.169	0.034
41 42	141.4 29.4 89.4	-189.9 -304.9 -304.9	0.532 0.169 0.811	-81.113 139.140 13.336	4	73.118	0.090	0.055
43 51 52 53	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	0.426 0.385 0.117 0.743	-48.292 6.730 42.225 40.562	5	89.904	0.117	0.074

Unsteady Transonic Delta Program

DPN = 358

PRESSURES section 1				00.65 mm
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
101 102 103 104	2.00 5.00 10.00 15.00	-1.319 -1.226 -1.189 -1.179	230 0.862 1.738 2.258	365 1.589 2.993 3.791
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 40.00 60.00 80.00	511404351284221157119092048006 0.363 0.243 0.114 0.092 0.118	-1.450 -1.510 706 262 123 111 138 150 163 042 042 023 039	-2.485 -2.384 -1.150 468 252 226 259 264 276 112 067 033 028 058

PRESSURES	section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 90.00 90.00 20.00 40.00 80.00	-1.019 962 887 831 823 747 653 361 278 258 259 190 131 0.362 0.248 0.135 0.105 0.094	318 0.117 0.855 1.408 0.348 1.502 1.017 0.366 0.011 208 189 174 171 191 047 047 049 025	404 0.442 1.711 2.005124 2.352 1.511 0.412105469452439437448129080052020039

PRESSURES section 3				.94.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 30.00 40.00 50.00 70.00 79.00 90.00	783 755 700 650 618 554 504 472 414 372 331 273	-1.053 843 385 027 0.109 0.309 0.342 0.411 0.444 0.451 0.467	-1.547 -1.074 191 0.405 0.608 0.789 0.718 0.729 0.682 0.621 0.586 0.618

PRESSURES	section		.44.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	634 608 544 514 494 413 376 346 316 288 260	-1.136 862 429 153 052 0.055 0.105 0.130 0.166 0.185 0.325	-1.873 -1.293 469 0.064 0.256 0.446 0.390 0.342 0.297 0.283 0.290 0.480

Unsteady Transonic Delta Program

PRESSURES section 5			b = x =-3	82.70 mm 269.60 mm
nr.	y/b [%]	Cp 0	ReCp 2	ImCp 2
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	136 134 148 215 304 443 574 580 449 443	031 048 144 399 435 059 0.608 0.508 184 0.004	056 082 215 564 628 163 0.691 0.643 181 0.016
	1			

PRESSURES section 6 b = 233.73 mm $\times$ = -60.62 mm						
nr. up	y/b [%]	0 qO	ReCp 2	ImCp 2		
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	674 772 580 505 419 382 464 -1.280 -1.269 -1.269	0.179 0.288 035 0.082 044 415 -1.567 796 1.857 1.771 0.862	0.094 0.275 003 0.061 103 683 -2.971 -1.678 3.311 3.055 1.589		

PRESSURES	section		117.90 mm 100.71 mm	
nr. up	y/b [%]	Cp 0	ReCp 2	ImCp 2
701 702 703 704 705 109 706 707 208 708 708 307 710 710 405	22.71 28.21 33.72 44.69 55.03 66.46 65.56 65.59 75.54 85.22 90.19 94.60	- 221 - 401 - 271 - 258 - 279 - 284 - 317 - 421 - 557 - 532 - 525 - 504 - 501 - 494	119285322236150262771697 0.366 0.773 0.683 0.342 0.104 0.040052	1264245243533004681.250 -1.123 0.412 1.147 1.107 0.718 0.494 0.490 0.256

SECTION COEFFICIENTS						
section	comp.	Zero	Re 2	Im 2		
1	CN_u CN_1 CN_t CM_u Cm_u Cm_1	0.481 0.155 0.636 008 012	0.138 036 0.102 121 0.007	0.221 057 0.164 207 0.010		
2	Cm_t CN_u CN_1 CN_t Cm_u Cm_1	020 0.545 0.158 0.703 058 011	114 365 029 394 014 0.001	197 457 058 515 063 0.006		
3	Cm_t CN_u Cm_u	069 0.462 066	012 246 0.154	057 468 0.199		
4	CN_u	0.380	015	163		
5	Cm_u CN_u Cl_u	059 0.279 170	0.080 0.041 002	0.134 0.079 013		
6	CN_u	0.710	237	302		
7	Cl_u CN_u Cl_u	375 0.353 202	0.171 0.065 0.006	0.270 0.034 0.065		

Simple Strake configuration

#### Unsteady Transonic Delta Program

test conditions

alpha = 10.031 deg | Q = 17.267 kPa

Mach = 0.601 | Ptot = 87.220 kPa

Re\*10^-6 = 8.024 | Ttot = 300.608 K

dalpha = 4.167 deg freq = 5.700 Hz k = 0.073 harm = 3

BALANCE	LOADS	aerodynamic	coefficients		aero	angular deflections [deg]		
position	comp.	Zero	Re 3	Im 3	inertia [%]	Zero	Re 3	Im 3
main	CN Cm Cl	0.56090 0.02249 22050	0.00506 00026 0.00167	0.06681 00438 04107	5556.22 233.71 3521.94	049 105	0.000	0.000

ACCELERATIONS				vibration mode				
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	1.059 0.546 0.400	-46.982 -47.599 133.650	1	2.878	0.001	0.012
21 22 23	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.384	-47.902 142.625	2	28.034	0.003	0.013
12 13 21 22 23 31 32 33	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	0.132	122.581	3	45.540	0.029	0.013
41	29.4 89.4 152.4	-304.9 -304.9 -304.9	0.786 0.745 1.717	81.488 83.673 85.779	4	73.118	0.022	0.037
41 42 43 51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	0.493 0.563 0.826	100.300 121.695 121.791	5	89.904	0.029	0.028

Unsteady Transonic Delta Program

PRESSURES	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
101 102 103 104	2.00 5.00 10.00 15.00	-1.319 -1.226 -1.189 -1.179	386 0.449 0.394 870	0.586 720 757 1.177
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 90.00 95.00 10.00 20.00 40.00 60.00	511 404 351 284 221 157 119 092 048 006 0.363 0.243 0.114 0.092 0.118	0.331 0.676 0.218 0.037 005 0.000 0.005 0.009 0.021 0.033 0.004 0.004 0.003 001	536 -1.184324020 0.041 0.035 0.023 0.014036 0.004 0.007 0.011 0.020 0.024

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 50.00 60.00 79.00 82.50 90.00 10.00 20.00 40.00 60.00 80.00	-1.019962887831923747653557361278258229131 0.362 0.248 0.135 0.105	- 362 0.122 0.716 0.465 - 119 - 180 - 496 - 498 - 388 - 219 - 207 - 193 - 172 - 139 0.009 - 001 - 006 - 018	0.362 362 -1.350 -1.545 077 181 0.447 0.711 0.760 0.684 0.645 0.625 0.055 0.013 0.027 0.035 0.069

PRESSURES	section	n 3		194.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 3	ImCp 3
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	783 755 700 650 618 554 504 472 414 372 331 273	331 047 0.440 0.662 0.681 - 0.454 0.222 0.054 067 160 211 242	0.351 014 759 -1.162 -1.259 -1.084 766 510 247 030 0.115 0.230

PRESSURE	S section		144.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 3	ImCp 3
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	634 608 544 514 494 413 376 346 316 288 260	553 233 0.161 0.373 0.417 0.295 0.151 0.066 0.012 026 055 047	0.651 0.253 297 611 685 520 292 134 029 0.030 0.071 0.009

Unsteady Transonic Delta Program

DPN = 358

PRESSURES section 5 $b = 82.70 \text{ mm}$ x = -269.60  mm						
nr. up	y/b [%]	Cp 0	ReCp 3	ImCp 3		
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	136 134 148 215 304 443 574 580 449 443	0.003 0.002 0.012 003 130 071 0.073 030	020 021 041 0.004 0.165 0.312 0.082 318 0.115 007		

PRESSURES	section		233.73 mm -60.62 mm	
nr. up	[%] A\p	Cp 0	ReCp 3	ImCp 3
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	674 772 580 505 419 382 464 -1.280 -1.369 -1.269	052 029 010 0031 0.157 0.568 751 -1.189 0.315 0.449	0.090 070 009 016 0.007 272 769 0.838 1.586 620 720

PRESSURES	section		17.90 mm 00.71 mm	
nr.	y/b [%]	Cp 0	ReCp 3	ImCp 3
701 702 703 704 705 109 706 707 208 708 708 708 709 307 710 711 405	22.71 28.21 33.72 44.69 50.03 55.28 60.46 65.56 70.59 75.54 80.42 90.19 94.60	221 401 271 258 279 284 317 421 557 532 504 501 494 494	0.023 0.044 0.100 005 011 0.037 0.110 167 498 290 0.030 0.222 0.340 0.458 0.417	061 035 140 0.081 0.084 020 119 0.699 0.711 0.093 571 771 812 685

SECTION O	COEFFICIE	INTS		
section	comp.	Zero	Re 3	Im 3
1	CN_u CN_1 CN_t CM_u Cm_u	0.481 0.155 0.636 008 012	075 0.001 074 0.025 0.000	0.144 0.015 0.159 031 006
2	Cm_t CN_u CN_l CN_t Cm_u Cm_l	020 0.545 0.158 0.703 058 011	0.025 0.195 005 0.190 080 0.004	037 261 0.036 225 0.195
3	Cm_t CN_u Cm_u	069 0.462 066	076 083 045	0.180 0.394 0.005
4	CN_u	0.380	066	0.149
5	Cm_u CN_u Cl_u	059 0.279 170	001 0.014 010	005 013 0.010
6	CN_u	0.710	0.005	0.031
7	Cl_u CN_u Cl_u	375 0.353 202	0.017 055 0.049	056 0.140 124

test conditions					Simple Strake configuration
alpha Mach Re*10^-6	=	10.031 deg 0.601 8.024	Q Ptot Ttot	= 17.267 kPa = 87.220 kPa = 300.608 K	
dalpha freq k harm	= = = = = =				

BALANCE	LOADS	aerodynamic	coefficients		aero	angular deflections [deg]		
position	comp.	Zero	Re 4	Im 4	inertia [%]	Zero	Re 4	Im 4
main	CN Cm Cl	0.56090 0.02249 22050	0.00225 00160 00200	00531 0.00065 0.00498	****** 6563.78 *****	049 105	0.000	0.000

ACCELERATIONS					vibration mode			
nr	x [mm]	[ <i>ww</i> ] A	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	0.140 0.091 0.047	158.308 171.214 4.091	1	2.878	0.001	0.001
21 22 23	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.038	-124.861 -28.023	2	28.034	0.001	0.001
23 31 32 33	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	0.109 0.159 0.695	-44.621 -44.882 -46.110	3	45.540	0.006	0.001
41 42 43	29.4 89.4 152.4	-304.9 -304.9 -304.9	0.184 0.734 0.243	17.826 -77.567 0.613	4	73.118	0.023	0.009
51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	0.132 0.336	-25.674 -11.359	5	89.904	0.004	0.004

DPN = 358

Unsteady Transonic Delta Program

PRESSURES	S section		00.65 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
101 102 103 104	2.00 5.00 10.00 15.00	-1.319 -1.226 -1.189 -1.179	0.288 627 025 0.192	0.007 064 0.044 0.051
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 85.00 90.00 10.00 20.00 40.00 80.00	511 404 351 284 221 157 119 092 048 006 0.363 0.243 0.114 0.092 0.118	304 178 365 182 048 005 008 004 007 008 007 008 001	005 0.040 0.042 0.001 019 017 014 010 009 0.001 0.001 0.001 002

PRESSURES section 2 $c = 246.21 \text{ mm}$ y = -273.97  mm									
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4					
201 202 203 204 205 206 207 208 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 10.00 20.00 40.00 80.00	-1.019 962 887 831 747 653 557 361 278 258 258 131 0.362 0.105 0.105	0.930 0.306 341 137 0.004 0.908 0.976 0.480 0.102 224 220 214 206 015 015 014 001	0.009 056 073 006 0.065 0.023 017 043 020 011 009 004 001 0.003 0.000 0.000 002 002 002					

PRESSURES	section		94.13 mm 36.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	783 755 700 650 618 554 504 472 414 372 331 273	0.352 024 696 977 971 433 082 0.129 0.264 0.333 0.360 0.381	0.060 002 055 035 0.002 0.111 0.120 0.091 0.058 0.021 005 043

PRESSURES	section		.44.42 mm 95.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	634 608 544 514 494 413 376 346 316 288 260	0.518 0.162 296 530 329 196 105 029 0.016 0.045	0.090 0.009 053 051 032 0.083 0.113 0.094 0.020 0.021

Unsteady Transonic Delta Program

PRESSURES	S section		82.70 mm 269.60 mm	
nr. up	y/b [%]	Cp 0	ReCp 4	ImCp 4
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	136 134 148 215 304 443 574 580 449 443	0.001 0.000 006 042 040 0.047 0.116 021 0.025 0.005	0.001 0.001 0.002 0.008 0.004 017 021 0.019 014

PRESSURE	S section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 4	ImCp 4
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	674 772 580 505 419 382 464 -1.280 -1.369 -1.269	0.053 0.022 007 0.024 0.035 007 275 0.265 0.189 0.476 627	016 0.005 006 002 0.012 0.007 016 008 0.037 0.092 064

PRESSURES	S section		117.90 mm 100.71 mm	
nr. up	[#] A\p	Cp 0	ReCp 4	ImCp 4
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22. 71 28. 21 33. 726 44. 69 50. 03 65. 28 60. 46 70. 59 75. 54 85. 22 90. 19 94. 60	2214012758279284317457532525504501494	005 060 023 133 148 1845 720 0.480 0.806 0.468 082 579 530	003 0.009 0.016 0.018 0.006 0.001 0.023 0.047 017 0.006 0.048 0.120 0.114 0.030

SECTION COEFFICIENTS									
section	comp.	Zero	Re 4	Im 4					
1	CN_u CN_1 CN_t Cm_u Cm 1	0.481 0.155 0.636 008	0.112 004 0.107 020 0.000	008 0.000 008 001 0.000					
2	Cm_t CN_u CN_1 CN_t Cm_u Cm_1	020 0.545 0.158 0.703 058 011	019 217 007 224 009 0.000	001 0.006 002 0.004 003 0.001					
3	Cm_t CN_u Cm_u	069 0.462 066	009 0.025 0.094	002 032 0.003					
4 5	CN_u Cm_u CN u	0.380 059 0.279	0.104 0.003 005	041 0.012 0.001					
6	CN_u Cl_u Cl_u CN_u Cl_u	0.279 170 0.710 375 0.353 202	0.005 0.005 0.025 044 0.101 073	001 0.009 004 018 0.013					

Unsteady Transonic Delta Program

BALANCE	LOADS	aerodynamic	coefficients		aero	angular deflections [deg]		
position	comp.	Zero	Re 5	Im 5	inertia [%]	Zero	Re 5	Im 5
main	CN Cm Cl	0.56090 0.02249 22050	0.04546 00235 02493	0.02569 00105 01372	6602.97 209.09 3718.29	049 105	0.000 001	0.000

ACCELERATIONS				vibration	mode			
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	[#] Y/b	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	1.300 0.622 0.495	60.521 61.673 -122.859	1	2.878	0.001	0.005
21 22	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.414	55.687 -100.594	2	28.034	0.002	0.005
31 32	-74.6 -10.6	-189.9 -189.9	0.233	11.694	3	45.540	0.007	0.001
33 41 42	141.4 29.4 89.4	-189.9 -304.9 -304.9	0.201 0.452 0.261	-28.380 5.736 9.532	4	73.118	0.015	0.006
12 13 12 22 33 33 44 43 55 55	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	0.502 0.449 0.347 0.339	-48.636 17.545 5.420 -21.789	5	89.904	0.022	0.001

DPN = 358

Unsteady Transonic Delta Program

PRESSURE	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
101 102 103 104	2.00 5.00 10.00 15.00	-1.319 -1.226 -1.189 -1.179	041 0.165 062 103	0.040 0.524 038 070
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 90.00 95.00 10.00 20.00 40.00 80.00	511 404 351 284 221 157 119 092 048 006 0.363 0.243 0.114 0.092 0.118	096 264 184 010 0.065 0.076 0.072 0.069 0.068 0.007 0.008 0.007 0.009	116422305062 0.038 0.055 0.049 0.045 0.039 0.007 0.002 0.003

PRESSURES section 2					
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5	
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 82.50 85.00 95.00 20.00 40.00 80.00	-1.019962887831823747653557439361278258229190131 0.362 0.248 0.135 0.094	053 0.069 0.131 094 019 433 330 116 0.105 0.223 0.235 0.237 0.224 0.007 0.009 0.012 0.012	0.062 0.412 0.527 0.019 0.012 465 156 0.249 0.215 0.215 0.215 0.215 0.149 0.000 0.003 0.000	

PRESSURES section 3			c = 194.13 mm y =-336.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 79.00 90.00	783 755 700 650 618 554 504 472 414 372 331 273	167 083 0.072 0.100 0.052 245 353 357 293 218 155 101	274 020 0.358 0.342 0.204 363 490 439 298 162 070 0.001

PRESSURE	S section	c = 144.42 mm y =-395.32 mm		
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 30.00 40.00 50.00 70.00 79.00 90.00	634 608 544 514 494 413 376 346 316 288 260	165 114 033 009 018 089 086 049 019 005 001	280 048 0.204 0.240 0.185 098 149 126 094 069 041

Unsteady Transonic Delta Program

DPN = 358

PRESSURES	section		82.70 mm 269.60 mm	
nr.	y/b [%]	Cp 0	ReCp 5	ImCp 5
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	136 134 148 215 304 443 574 580 449 443	006 006 008 006 0.002 004 031 033 001	001 004 004 007 0.008 018 019 0.009

PRESSURES	section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	674 772 580 505 419 382 464 -1.280 -1.369 -1.269	028 025 0.000 018 019 033 060 221 059 051	007 0.005 0.035 0.011 0.012 037 148 258 0.028 008 0.524

PRESSURES	section		17.90 mm .00.71 mm	
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.726 44.69 55.28 60.46 65.56 70.59 75.54 80.42 85.22 90.19 94.60	221 401 271 258 279 317 421 557 532 525 504 304 494 494	006 0.006 0.001 005 010 084 0.202 115 322 443 353 138 032 018	024 017 008 035 038 062 200 0.098 0.091 226 517 490 226 0.039 0.185

SECTION (	SECTION COEFFICIENTS						
section	comp.	Zero	Re 5	Im 5			
1	CN_u CN_l CN_t CM_u	0.481 0.155 0.636 008	0.042 0.012 0.055 0.005	0.064 0.007 0.071 011			
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u	012 020 0.545 0.158 0.703 058	004 0.000 0.005 0.017 0.022 0.037	002 014 084 0.009 075 0.038 004			
3	Cm_1 Cm_t CN_u Cm_u CN_u	011 069 0.462 066 0.380	0.030 0.177 053 0.042	0.033 0.148 044 0.043			
5	CM_u CN_u Cl_u	059 0.279 170	003 0.010 006	0.003			
6	CN_u Cl_u	0.710	0.013	050 0.051			
7	CN_u Cl_u	0.353	0.067 051	0.071 045			

Simple Strake configuration

alpha = 22.050 deg | Q = 17.344 kPa Mach = 0.600 Ptot = 87.810 kPa Re\*10^-6 = 8.061 Ttot = 300.862 K

BALANCE	LOADS	aerodynamic coefficients		aero	angular deflections [deg]			
position	comp.	Zero	Re 1	Im 1	inertia [%]	Zero	Re 1	Im 1
main	CN Cm Cl	1.09614 0.06577 37743	1.74730 0.20619 39155	0.58990 01744 23482	5071.63 365.33 1297.56	124 163	059 011	0.001 017

ACCEL	ERATIONS				vibration	тоде		
nr	[mm]	[mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	74,222 36.997 33.087	3.190 4.391 -179.215	1	2.878	1.932	8.058
11 112 112 122 122 123 123 123 133 142 143 155 153	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	22.774 26.942	16.119 -168.090	2	28.034	2.992	8.536
31 32	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	5.756 23.271	24.992 -171.665	3	45.540	3.553	5.963
41 42	29.4 89.4 152.4	-304.9 -304.9 -304.9	0.788 7.899 19.057	-17.774 -180.000 176.775	4	73.118	4.813	7.186
51 52	85.0 121.4 157.4	-374.9 -374.9 -374.9	12.719 23.224 25.204	-163.897 -162.816 -163.705	5	89.904	0.735	7.684

Unsteady Transonic Delta Program

DPN = 375

PRESSURES	section		00.65 mm 09.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
101 102 103 104	2.00 5.00 10.00 15.00	-1.449 -1.425 -1.442 -1.382	0.717 0.589 1.289 0.979	913 894 867 653
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 60.00	-1.271 -1.039 821 698 615 567 548 531 521 0.630 0.513 0.343 0.251	0.220 153 850 -1.289 -1.405 -1.570 -1.579 -1.675 -1.718 0.926 1.068 0.999 0.735	-1.084 867 689 667 625 604 587 578 525 0.107 0.140 0.162 0.161

PRESSURES	section		46.21 mm 73.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 79.00 82.50 90.00 95.00 90.00 95.00 90.00 95.00 80.00 80.00	935 924 916 882 887 848 816 770 698 654 587 557 557 559 0.611 0.345 0.237 0.119	0.736 0.707 0.730 0.726 124 0.448 0.332 0.296 0.099 458 563 634 805 -1.040 0.850 0.998 0.647 0.269	923 858 786 784 305 639 563 526 485 466 478 456 478 512 0.122 0.147 0.152 0.183

PRESSURES	PRESSURES section 3			.94.13 mm 36.06 mm
nr.	x/c [%]	Cp 0	ReCp 1	ImCp 1
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	613 628 635 637 631 628 615 617 571 548 535 512	0.661 0.615 0.635 0.592 0.572 0.409 0.243 0.080 067 224 379 559	632 647 662 605 588 566 568 549 412 407

PRESSURES	PRESSURES section 4			.44.42 mm 95.32 mm
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 90.00	475482469467467469457442428412397396	0.689 0.686 0.654 0.653 0.653 0.587 0.416 047 227 361 516	420 417 395 380 374 337 303 282 274 266 243 260

Unsteady Transonic Delta Program

$\begin{array}{c ccccccccccccccccccccccccccccccccccc$					
up         [%]        448         -1.480        177           501         6.62        448         -1.480        177           502         20.43        564         -2.311        131           503         34.05        804         -3.572        057           504         47.67         -1.179         -4.530         0.030           505         54.49         -1.336         -4.207        006           506         61.29         -1.338         -2.949        093           507         68.10         -1.212         -2.106        170           508         74.91         -1.028         -2.129        198           509         81.72        927         -2.054        196	PRESSURES section 5				
502         20.43        564         -2.311        131           503         34.05        804         -3.572        057           504         47.67         -1.179         -4.530         0.030           505         54.49         -1.336         -4.207        006           506         61.29         -1.338         -2.949        093           507         68.10         -1.212         -2.106        170           508         74.91         -1.028         -2.129        198           509         81.72        927         -2.054        166			Cp 0	ReCp 1	ImCp 1
	502 503 504 505 506 507 508 509	20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72	564 804 -1.179 -1.336 -1.338 -1.212 -1.028 927	-2.311 -3.572 -4.530 -4.207 -2.949 -2.106 -2.129 -2.054	131 057 0.030 006 093 170 198 196

PRESSURES section 6				233.73 mm -60.62 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.431 -1.436 -1.265 -1.141 -1.048 -1.636 -1.616 -2.256 -1.791 -1.635 -1.425	-1.497 -1.834 -2.262 -2.444 -2.845 -3.724 -2.226 3.113 1.548 1.582 0.589	898 -1.289 -1.004 723 720 813 967 805 820 968 894

PRESSURES section 7				117.90 mm 100.71 mm
nr.	y/b [%]	Cp 0	ReCp 1	ImCp 1
701 702 703 704 705 706 707 208 708 709 307 710 711 405	22.71 28.21 33.72 39.26 44.69 50.03 55.28 60.46 65.56 75.54 80.42 90.19 94.60	346 739 888 720 707 698 770 770 659 659 659 528 467	657 015 -1.273 -1.467 -1.289 502 0.144 0.296 0.155 0.169 0.243 0.381 0.526	126 591 -1.022 769 667 629 555 556 570 609 568 514 439 374

SECTION (	COEFFICI	ents		
section	comp.	Zero	Re 1	Im 1
1	CN_u CN_1 CN_t CM_u	0.915 0.324 1.239 123	0.537 0.744 1.281 416	0.738 0.141 0.879 149
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u Cm_1	027 149 0.729 0.307 1.036 136	108 524 026 0.665 0.639 131	036 185 0.553 0.124 0.677 113
3	Cm_t CN_u Cm_u	157 0.560 116	222 071 104	138 0.518 107
4	CN_u	0.418	141	0.307
5	Cm_u CN_u	0.853	2.679	0.123
6	Cl_u CN_u Cl_u	459 1.359 658	-1.337 1.115 333	068 0.886 439
7	CN_u Cl_u	0.582 302	0.308 025	0.484

Simple Strake configuration

test conditions

alpha = 22.050 deg | Q = 17.344 kPa
Mach = 0.600 | Ptot = 87.810 kPa
Re\*10^-6 = 8.061 | Ttot = 300.862 K

BALANCE	LOADS	aerodynamic coefficients		aero	angular defl	ections [deg]		
position	comp.	Zero	Re 2	Im 2	inertia [%]	Zero	Re 2	Im 2
main	CN Cm Cl	1.09614 0.06577 37743	75553 00748 0.28860	47267 01266 0.17838	****** 5377.40 *****	124 163	0.005 0.020	0.005 0.012

ACCEL	ERATIONS				vibration	mode		
nr	[mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/p	heave at p.a [mm]	pitch [deg]
11 12	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	4.888 3.397 4.548	-144.159 -150.675 34.434	1	2.878	0.091	0.163
21 22	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	1.531	-118.942 -162.554	2	28.034	0.238	0.045
11 12 13 21 22 23 31 32 33	-74.6 -10.6	-189.9 -189.9 -189.9	1.072	-154.278 -164.316	3	45.540	0.315	0.084
41	141.4 29.4 89.4	-304.9 -304.9	2.148 2.096 5.431	-156.840 149.598 177.430	4	73.118	0.044	0.325
43 51 52 53	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	1.028 0.279 0.983	129.427 -152.559 128.979	5	89.904	0.094	0.000

Unsteady Transonic Delta Program

PRESSURES section 1				300.65 mm 209.06 mm
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
101 102 103 104	2.00 5.00 10.00 15.00	-1.449 -1.425 -1.442 -1.382	1.377 1.402 1.624 1.425	0.275 0.335 0.514 0.365
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 80.00	-1.271 -1.039 -821 -698 -615 -567 -548 -531 -517 0.630 0.513 0.343 0.251	1.681 1.315 1.123 1.007 0.923 0.918 0.950 0.974 1.068 1.216 073 081 077 062 012	1.631 1.018 0.804 0.820 0.767 0.718 0.747 0.773 0.867 1.028 100 080 055 034

PRESSURES	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 50.00 60.00 79.00 82.50 90.00 95.00 90.00 95.00 90.00 95.00 80.00 80.00	- 935 - 924 - 916 - 882 - 887 - 848 - 816 - 770 - 654 - 587 - 557 - 539 - 516 0 611 0 .504 0 .237 0 .119	1.407 1.326 1.1257 1.184 0.107 0.867 0.723 0.629 0.572 0.467 0.461 0.509 0.628 097 102 077 044 0.042	0.701 0.624 0.564 0.362 154 0.320 0.286 0.208 0.166 0.077 0.074 0.086 0.136 0.263 113 093 059 023 0.043

PRESSURES section 3				194.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	613 628 635 637 631 628 615 617 571 548 535	0.646 0.647 0.637 0.558 0.540 0.448 0.395 0.324 0.221 0.243	0.353 0.353 0.402 0.382 0.374 0.283 0.196 0.142 080 079

PRESSURES section 4				144.42 mm 395.32 mm
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	475482469468467469457442428412397396	0.197 0.183 0.165 0.144 0.133 0.076 0.028 011 036 068 085 056	0.351 0.325 0.269 0.233 0.213 0.118 0.014 085 177 251 299 353

DPN = 375

Unsteady Transonic Delta Program

PRESSURES section 5				82.70 mm 269.60 mm
nr. up	y/b [%]	Cp 0	ReCp 2	ImCp 2
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	448 564 804 -1.179 -1.336 -1.338 -1.212 -1.028 927 956	0.203 0.240 0.355 0.550 0.571 0.444 0.302 0.329 0.353 0.340	0.073 0.084 0.222 0.478 0.491 0.356 0.179 0.200 0.224

PRESSURES section 6				233.73 mm -60.62 mm
nr. up	y/b [%]	Ср 0	ReCp 2	ImCp 2
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.431 -1.436 -1.265 -1.141 -1.048 -1.236 -1.616 -2.256 -1.791 -1.635 -1.425	1.922 2.112 1.775 1.297 1.100 1.529 1.644 1.204 1.238 1.557	1.235 0.795 0.813 0.638 1.174 1.986 1.452 0.370 0.584 0.335

PRESSURES	PRESSURES section 7			17.90 mm .00.71 mm
nr.	y/b [%]	Cp 0	ReCp 2	ImCp 2
701 702 703 704 705 109 706 707 208 708 709 307 710 711 405	22.71 28.21 33.72 44.69 50.03 55.28 60.46 65.59 75.54 80.42 90.19 94.60	346 739 888 720 707 698 770 770 659 615 579 528 467	186 0.873 2.203 1.498 1.153 1.007 0.845 0.744 0.629 0.509 0.418 0.299 0.198	- 184 0.559 1.747 1.193 0.905 0.820 0.576 0.339 0.208 0.192 0.192 0.251 0.251

SECTION COEFFICIENTS						
section	comp.	Zero	Re 2	Im 2		
1	CN J CN J CN J	0.915 0.324 1.239 123 027	-1.237 055 -1.292 0.253 0.006	843 042 885 0.230 001		
2	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	149 0.729 0.307 1.036 136 020	0.259 687 041 728 0.119 006	0.230 230 034 264 0.031 008		
3	Cm_t CN_u Cm_u	157 0.560 116	0.113 399 0.064	0.022 137 016		
4	CN_u	0.418	021 020	0.056 081		
5	CM_U	0.853	352 0.189	229 0.128		
6	Cl_u CN_u	1.359	-1.637	978		
7	Cl_u CN_u Cl_u	658 0.582 302	0.752 540 0.288	0.421 365 0.197		

Simple Strake configuration

alpha = 22.050 deg | Q = 17.344 kPa Mach = 0.600 | Ptot = 87.810 kPa Re\*10^-6 = 8.061 | Ttot = 300.862 K

BALANCE	BALANCE LOADS aerodynamic coefficients		aero	angular defl	ections [deg]			
position	comp.	Zero	Re 3	Im 3	inertia [%]	Zero	Re 3	Im 3
main	CN Cm Cl	1.09614 0.06577 37743	0.01355 00011 00486	07823 02563 0.00585	4049.73 839.22 400.87	124 163	001 0.000	0.005

ACCELERATIONS					vibration mode			
nr	x [mm]	[mw]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	3.763 2.218 1.769	-104.102 -109.575 107.849	1	2.878	0.054	0.044
21 22	-138.6 -46.6	-116.9 -116.9	1.340	-88.266	2	28.034	0.008	0.047
23 31 32	121.4 -74.6 -10.6	-116.9 -189.9 -189.9	1.151 0.734	100.270 -70.442	3	45.540	0.008	0.045
33 41	141.4 29.4	-189.9 -304.9	1.237 0.622 0.644	99.486 12.639 -98.562	4	73.118	0.056	0.047
42 43 51	89.4 152.4 85.0	-304.9 -304.9 -374.9	0.676 0.424	-113.601 108.692				
52 53	121.4 157.4	-374.9 -374.9	0.846 0.656	83.975 30.954	5	89.904	0.114	0.048

DPN = 375

Unsteady Transonic Delta Program

PRESSURES	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
101 102 103 104 105 106 107 108 110 111 111 112 113 114 115 151	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 79.00 82.50 90.00 95.00 10.00 20.00	-1.449 -1.425 -1.425 -1.382 -1.271 -1.039 821 698 615 567 548 531 517 521 0.630 0.513	164 278 103 229 0.050 026 122 072 022 0015 0.018 0.043 0.143 006	0.397 0.543 0.212 0.577 -0.004 0.083 0.078 0.071 0.030 0.001 -0.030 0.001 -0.095 -281 0.002 -0.013
153 154 155	40.00 60.00 80.00	0.343 0.251 0.170	0.008 0.015 0.034	020 020 028

PRESSURES	section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 253 254 255	2.00 5.00 10.00 18.00 30.00 60.00 70.00 79.00 82.50 85.00 95.00 10.00 40.00 60.00	935 924 916 882 887 709 654 587 539 516 0.611 0.504 0.345 0.237 0.119	0.061 0.033 001 0.018 0.021 009 0.058 0.104 0.073 0.047 0.028 0.014 031 031 006 0.001 0.010	0.323 0.328 0.325 0.267 0.007 0.030 027 089 124 041 0.005 0.032 0.076 0.085 025 027 023

PRESSURES	section		194.13 mm 336.06 mm	
nr. up	x/c [%]	Ср 0	ReCp 3	ImCp 3
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	613 628 635 637 631 628 615 617 571 548 535 512	079083080076074043002 0.029 0.068 0.098 0.099	0.009 010 084 167 204 276 246 229 198 134 060 0.049

PRESSURES	S section		144.42 mm 395.32 mm	
nr.	x/c [%]	Cp 0	ReCp 3	ImCp 3
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	475482469468467469457442428412397396	059 053 041 036 030 0.005 0.056 0.101 0.128 0.144	628 647 627 644 655 698 661 552 410 280 154 006

DPN = 375

Unsteady Transonic Delta Program

PRESSURES	section		82.70 mm 269.60 mm	
nr. up	y/b [%]	Cp 0	ReCp 3	ImCp 3
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	448 564 804 -1.179 -1.336 -1.338 -1.212 -1.028 927 956	0.026 005 030 010 0.014 0.009 020 022 0.007 0.017	0.043 0.133 0.208 0.213 0.204 0.198 0.252 0.255 0.161 0.133

PRESSURE	S section	n 6		233.73 mm -60.62 mm
nr.	y/b [%]	Cp 0	ReCp 3	ImCp 3
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.431 -1.436 -1.265 -1.141 -1.048 -1.236 -1.616 -2.256 -1.791 -1.635 -1.425	0.021 0.123 0.219 007 163 243 148 035 041 008 278	0.932 1.000 0.632 0.314 0.276 0.292 173 562 179 027 0.543

PRESSURE	S section	1 7		17.90 mm 100.71 mm
nr.	y/b [%]	Cp 0	ReCp 3	ImCp 3
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.72 50.03 55.28 60.46 65.56 70.59 75.54 85.22 90.19 94.60	346 739 888 720 707 698 770 697 659 615 579 579 528	081 002 0:147 103 113 072 009 0.036 0.104 0.042 002 024 024 036	019083308 0.136 0.194 0.089034087128246428594655

SECTION (	COEFFICI	ents		
section	comp.	Zero	Re 3	Im 3
1	CN_u CN_1 CN_t CM_u Cm_u	0.915 0.324 1.239 123 027	0.049 0.014 0.063 0.009 008	100 018 118 021 0.007
2	Cm_t CN_u CN_l CN_t Cm_u Cm_l Cm_t	149 0.729 0.307 1.036 136 020	0.001 035 0.015 021 0.008 008	014 033 027 059 013 0.008
3	CN_u Cm_u	0.560 116	014 0.021	0.135 019
4	CN_u Cm_u	0.418	050 0.034	0.436
5	CN_u Cl_u	0.853 459	0.001	167 0.092
6	CN_u Cl_u	1.359	0.052 051	535 0.176
7	CN_u Cl_u	0.582	0.025	0.136 113

Simple Strake configuration

BALANCE	NCE LOADS aerodynamic coefficients		derodynamic coccernos		angular deflections [deg]			
position	comp.	Zero	Re 4	Im 4	inertia [%]	Zero	Re 4	Im 4
main	CN Cm Cl	1.09614 0.06577 37743	02885 0.00683 0.02522	0.17815 00587 07916	****** 1281.30 *****	124 163	001 0.002	0.000

ACCEL	ERATIONS				vibration mode			
nr	[mm]	[mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	[#] A\p	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	1.558 0.853 2.508	-27.376 -20.121 -82.233	1	2.878	0.059	0.008
21 22	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.568	-7.553 143.791	2	28.034	0.006	0.010
11 12 13 21 22 23 31 32 33	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	0.677	85.780 130.385	3	45.540	0.048	0.018
	29.4 89.4 152.4	-304.9 -304.9 -304.9	1.358 1.732 2.554	86.058 100.106 127.386	4	73.118	0.069	0.040
41 42 43 51 52	85.0 121.4 157.4	-374.9 -374.9 -374.9	1.460 1.690 2.213	128.978 134.332 128.299	5	89.904	0.024	0.029

Unsteady Transonic Delta Program

PRESSURE	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
101 102 103 104	2.00 5.00 10.00 15.00	-1.449 -1.425 -1.442 -1.382	145 064 108 0.050	575 542 600 414
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 40.00 60.00 80.00	-1.271 -1.039 -821 -698 -615 567 548 531 521 0.630 0.513 0.343 0.251	0.136 0.078 0.079 0.126 0.128 0.138 0.165 0.189 0.256 0.282 001 001 004 013	387 419 286 190 177 204 209 217 236 237 0.007 0.014 0.021 0.027 0.029

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 15.00 40.00 50.00 60.00 79.00 82.50 90.00 90.00 91.00 20.00 40.00 40.00 80.00	935 924 916 882 887 848 816 770 698 557 557 557 559 0.611 0.504 0.237 0.119	153 154 099 100 031 0.022 0.007 0.013 0.042 020 025 0.036 0.128 0.002 0.002 013 014	336 346 385 350 006 311 206 154 126 147 144 157 181 0.009 0.017 0.022 0.023 0.011

PRESSURES section 3			c = 194.13 mm y =-336.06 mm		
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4	
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	613 628 635 637 631 628 615 617 571 548 535 512	0.129 0.146 0.205 0.242 0.263 0.288 0.214 0.145 0.044 061 126	376 385 382 365 291 214 154 084 035 027 063	

PRESSURES	section		144.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	475 482 469 468 467 469 457 442 428 412 397 396	0.462 0.458 0.422 0.418 0.386 0.289 0.158 0.010 098 177 242	289 284 264 259 258 172 081 0.000 0.056 0.083 0.067

DPN = 375

Unsteady Transonic Delta Program

PRESSURES	section		82.70 mm 269.60 mm	
nr. up	y/b	Cp 0	ReCp 4	ImCp 4
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	448 564 804 -1.179 -1.336 -1.338 -1.212 -1.028 927 956	045 044 015 029 074 069 006 0.038 041 098	012 019 026 020 007 009 025 038 017 003

PRESSURES	section		233.73 mm -60.62 mm	
nr. up	[#] Y/p	Cp 0	ReCp 4	ImCp 4
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.431 -1.436 -1.265 -1.141 -1.048 -1.236 -1.616 -2.256 -1.791 -1.635 -1.425	112 397 268 0.063 0.067 0.079 0.057 0.404 0.031 127 064	047 272 253 317 410 378 426 542 653 661 542

nr. up	[#] A\p	Cp 0	ReCp 4	ImCp 4
701 702 703 704 705 109 706 707 208 708 709 307 711 405	22.71 28.21 33.72 39.26 44.69 50.03 55.28 60.46 65.56 70.59 75.54 85.22 90.19 94.60	346 739 888 720 707 698 729 770 697 659 579 528 467	088 027 0.084 0.290 0.142 0.126 0.090 0.087 0.013 0.047 0.142 0.214 0.324 0.416	0.057 246 644 392 254 190 198 199 154 214 237 2261 258

SECTION C	COEFFICIE	INTS		
section	comp.	Zero	Re 4	Im 4
1	CN_u CN_l CN_t Cm_u Cm_l	0.915 0.324 1.239 123 027	104 014 118 0.051 0.007	0.325 0.022 0.346 048 008
2	Cm_t CN_u CN_l CN_t Cm_u Cm_l	149 0.729 0.307 1.036 136	0.058 0.004 007 003 0.013 0.004	056 0.197 0.016 0.213 035 004
3	Cm_t CN_u	157 0.560	0.017 084 019	039 0.183 009
4	Cm_u CN_u	116 0.418	127 043	0.091
5	Cm_u CN_u	087 0.853	0.042	0.017
6	Cl_u CN_u	459 1.359	023 0.051 010	008 0.288 205
7	Cl_u CN_u Cl_u	658 0.582 302	010 107 0.094	0.180 112

BALANCE	LOADS	aerodynamic coefficients		aero	angular deflections [deg]			
position	comp.	Zero	Re 5	Im 5	inertia [%]	Zero	Re 5	Im 5
main	CN Cm Cl	1.09614 0.06577 37743	04621 00713 0.01242	0.07858 0.00251 03073	8118.78 432.34 3050.37	124 163	0.001	001 002

ACCEL	ERATIONS				vibration	mode		
nr	x [mm]	[mm] A	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b	heave at p.a [mm]	pitch [deg]
11 12	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	2.288 1.463 1.140	101.046 102.881 -52.289	1	2.878	0.008	0.010
11 12 13 21 22 23 31 32 33	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.901	114.152 -62.376	2	28.034	0.005	0.009
31 32	-74.6 -10.6	-189.9 -189.9	0.781	134.019	3	45.540	0.013	0.009
	141.4 29.4 89.4 152.4	-189.9 -304.9 -304.9 -304.9	0.416 0.811 0.221 0.600	66.926 73.841 -31.988	4	73.118	0.031	0.016
41 42 43 51 52	85.0 121.4 157.4	-374.9 -374.9 -374.9	0.266 0.274 0.515	86.038 2.830 -89.581	5	89.904	0.040	0.019

DPN = 375

Unsteady Transonic Delta Program

PRESSURES	section		00.65 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
101 102 103 104	2.00 5.00 10.00 15.00	-1.449 -1.425 -1.442 -1.382	0.208 0.197 0.101 0.142	110 044 065 0.043
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 79.00 82.50 85.00 90.00 95.00 10.00 40.00 60.00	-1.271 -1.039 821 698 615 567 548 531 517 0.630 0.513 0.343 0.251	0.173 0.067 0.032 0.055 0.087 0.103 0.085 0.064 007 121 0.000 005 009 011 010	049 003 0.008 051 072 043 051 252 0.004 0.008 0.011 0.009

PRESSURES	section		46.21 mm 73.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 10.00 90.00 20.00 40.00 80.00	935 924 916 882 887 848 654 770 587 557 539 516 0.611 0.504 0.345 0.237	0.128 0.125 0.148 0.086 011 0.049 043 085 097 064 042 036 034 057 002 008 015 018 023	299259223012150152154133074073069068093 0.010 0.012 0.010 0.004012

PRESSURES section 3				.94.13 mm 36.06 mm
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	613 628 635 637 631 628 615 617 571 548 535	0.311 0.305 0.278 0.229 0.200 0.061 030 092 132 115 080 025	0.015 0.021 0.019 0.026 0.031 0.026 023 059 098 128 133 107

PRESSURES	PRESSURES section 4			.44.42 mm 395.32 mm
nr.	x/c [%]	Cp 0	ReCp 5	ImCp 5
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 90.00	475 482 469 468 467 457 442 428 412 397 396	058083112141158242299314287230147051	0.084 0.082 0.072 0.074 0.076 0.080 0.054 0.009 042 076 088 076

Unsteady Transonic Delta Program

PRESSURES section 5				82.70 mm 269.60 mm
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	448 564 804 -1.179 -1.336 -1.338 -1.212 -1.028 927 956	004 0.003 0.004 0.004 0.023 0.055 0.070 0.051 0.026	045 039 043 044 029 0.004 0.017 003 022 038

PRESSURES section 6				233.73 mm -60.62 mm
nr. up	y/b	Cp 0	ReCp 5	ImCp 5
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.431 -1.436 -1.265 -1.141 -1.048 -1.236 -1.616 -2.256 -1.791 -1.635 -1.425	0.277 0.306 0.277 0.252 0.276 0.130 0.162 0.133 0.061 0.141	020 371 561 327 193 204 0.053 0.178 131 078 044

PRESSURES	PRESSURES section 7			117.90 mm 100.71 mm
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.72 44.69 50.03 55.28 60.46 65.56 70.59 75.54 80.42 90.19 94.60	346 739 888 720 707 698 729 770 697 659 579 579 528 467	096 0.134 0.248 0.123 0.093 0.055 0.006077085037030046102158	0.143 0.031 209 144 051 055 115 154 089 023 0.029 0.071 0.076

SECTION COEFFICIENTS					
section	comp.	Zero	Re 5	Im 5	
1	CN_u CN_1 CN_t CM_u	0.915 0.324 1.239 123	082 008 089 0.003	0.054 0.007 0.061 025	
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u Cm_1	027 149 0.729 0.307 1.036 136 020	0.003 0.005 0.018 015 0.003 021 0.006	002 026 0.132 0.003 0.134 021 0.002	
3	Cm_t CN_u Cm_u	157 0.560 116	015 020 028	019 0.051 030	
4	CN_u Cm_u	0.418	0.189	002 019	
5	CN_u	0.853	017 0.012	0.030	
6	Cl_u CN_u Cl_u CN_u	459 1.359 658 0.582	230 0.101 0.019	0.098 052 0.001	
,	Cl_u	302	016	011	

Simple Strake configuration

test conditions

alpha = 10.368 deg | Q = 24.218 kPa
Mach = 0.901 | Ptot = 72.140 kPa
Re\*10^-6 = 8.089 | Ttot = 307.409 K

dalpha = 4.169 deg freq = 7.600 Hz k = 0.067 harm = 1

BALANCE	LOADS	aerodynamic coefficients		aero	angular deflections [deg]			
position	comp.	Zero	Re 1	Im 1	inertia [%]	Zero	Re 1	Im 1
main	CN Cm Cl	0.61701 0.00947 24576	2.66946 0.11016 79701	0.41525 03759 23785	5835.52 161.41 1856.67	046 165	033 028	0.004

ACCELE	RATIONS				vibration	mode		
nr	x [mm]	[mm] A	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	[#] Y/p	heave at p.a [mm]	pitch [deg]
11 12	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	60.358 35.114 27.145	1.298 1.244 -179.115	1	2.878	0.997	3.604
11 12 13 21 22 22 33 31 32 33 41 42 43 55 55 55	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	21.493	10.908	2	28.034	0.229	3.982
31 32	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	7.223 18.032	19.645	3	45.540	1.042	2.903
41 42	29.4 89.4 152.4	-304.9 -304.9 -304.9	1.475 11.335 13.641	112.733 174.010 -174.969	4	73.118	1.171	2.737
51 52	85.0 121.4 157.4	-374.9 -374.9 -374.9	14.836 21.898 29.040	-167.753 -166.783 -169.273	5	89.904	1.033	4.934

Unsteady Transonic Delta Program

PRESSURES section 1				300.65 mm 209.06 mm
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
101 102 103 104	2.00 5.00 10.00 15.00	-1.176 -1.159 926 820	-1.148 -1.202 -4.485 -5.172	157 270 154 134
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 70.00 79.00 82.50 85.00 90.00 95.00 20.00 40.00 60.00	726 672 601 594 433 244 220 197 127 124 0.359 0.076 0.056	-4.171 -2.541705 0.338 -1.668 -3.597 -3.856 -3.987 -3.854 1.6547 0.897	328 552 209 397 750 0.725 0.725 0.794 0.820 0.104 0.143 0.200 0.251

PRESSURES	S section		246.21 mm 273.97 mm	
nr. .up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 15.00 40.00 50.00 60.00 79.00 82.50 90.00 90.00 91.00 20.00 40.00 60.00 80.00	-1.109 -1.0249668357836512659611404299282282199 0.342 0.224 0.106 0.081 0.102	2.627 2.179 1.712 035 952 -1.669 -1.000 475 0.305 -4.283 -4.520 -4.493 -4.695 -4.858 1.913 1.770 0.744 105	-1.116 -1.156961742039739968 -1.170867 0.285 0.411 0.416 0.499 0.1584 0.130 0.159 0.251 0.287

PRESSURES section 3				194.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 79.00 90.00	985 953 910 865 831 712 664 532 313 313	4.524 3.913 3.273 3.236 3.062 0.649 0.484 0.586 511 -3.663 -3.983 -4.491	-1.730 -1.590 -1.527 -1.752 -1.753 -1.589 -1.501 941 642 0.740 0.976

PRESSURES section 4				144.42 mm 395.32 mm
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00	906878820813798766645525445413398360	5.446 4.970 4.359 4.316 4.141 3.469 1.018 883 -2.182 -2.367 -2.388 -3.182	-2.864 -2.707 -2.550 -2.466 -2.474 -2.259 -1.797 -1.639 -813 -618 -421 -086

DPN = 593

Unsteady Transonic Delta Program

PRESSURES section 5				82.70 mm 869.60 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	064 060 076 136 219 334 420 396 310	678 714 -1.039 -2.410 -3.767 -4.821 -3.499 838 -1.339 -1.805	006 008 0.013 0.134 0.259 0.372 0.240 090 015 0.045

PRESSURES section 6				233.73 mm -60.62 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	650 574 508 511 487 640 860 868 872 903 -1.159	-3.385 -1.933 -2.560 -2.417 -2.208 -4.530 -7.045 -6.606 -6.086 -5.470 -1.202	0.167 104 0.046 0.020 0.073 0.079 0.169 0.115 0.032 270

PRESSURE	PRESSURES section 7			17.90 mm .00.71 mm
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.72 44.69 50.03 55.28 60.46 65.56 70.59 75.54 80.42 85.22 90.19 94.60	653 629 611 552 583 594 632 649 616 630 664 797 798	0.231 0.270 - 166 0.179 0.527 0.338 - 299 - 375 - 475 - 066 0.068 0.484 1.442 3.376 4.141	0.019 377 225 324 521 397 835 -1.058 -1.170 895 917 -1.501 -1.841 -2.219 -2.474

	SECTION COEFFICIENTS					
le 1	Im 1					
2.774 1.048 3.823 651	0.069 0.209 0.279 0.068					
713 1.480 0.938 2.418 900	002 0.458 0.219 0.677 0.013					
907 0.477 961	059 0.618 0.139					
311	1.408					
1.817	080 072 0.044					
3.442	065					
-1.743 629 0.524	0.004 0.797 611					
	1.048 3.823 651 062 713 1.480 0.938 2.418 900 007 0.477 961 977 1.817 1.1060 3.442 1.743					

Simple Strake configuration

test conditions

alpha = 10.368 deg | Q = 24.218 kPa

Mach = 0.901 | Ptot = 72.140 kPa

Re\*10^-6 = 8.089 | Ttot = 307.409 K

dalpha = 4.169 deg freq = 7.600 Hz k = 0.067 harm = 2

BALANCE	CE LOADS aerodynamic coefficients			aero	angular defle	gular deflections [deg]		
position	comp.	Zero	Re 2	Im 2	inertia [%]	Zero	Re 2	Im 2
main	CN Cm Cl	0.61701 0.00947 24576	39376 0.00901 0.22970	34069 0.01509 0.20963	****** 1323.35 *****	046 165	0.000	001 0.012

ACCELI	ERATIONS				vibration	mode		
nr	[mm]	[mm] A	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	5.310 0.819 0.686	-146.017 -177.690 33.099	1	2.878	0.120	0.075
21 22	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.634	-154.708 -113.436	2	28.034	0.046	0.011
23 31 32 33 41 42	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	3.118 3.577	-136.342 -119.139	3	45.540	0.355	0.032
41 42 43	29.4 89.4 152.4	-304.9 -304.9 -304.9	3.465 3.317 2.060	-173.475 -144.257 -130.980	4	73.118	0.459	0.123
51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	1.077 0.886 1.202	-111.029 -105.889 -82.882	5	89.904	0.138	0.049

Unsteady Transonic Delta Program

PRESSURES section 1			c = 300.65 mm y =-209.06 mm		
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2	
101 102 103 104 105 106 107 108 109 110 111 112 113	2.00 5.00 10.00 15.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00	-1.176 -1.159926820726672601594433244220197	1.022 1.087 1.625 1.167 1.279 1.024 088 119 826 905 852	0.927 0.824 1.413 0.817 0.852 0.523 0.209 0.100 0.431 558 642 710	
114 115 151 152 153 154 155	90.00 95.00 10.00 20.00 40.00 60.00 80.00	157 124 0.359 0.230 0.076 0.056 0.115	975 -1.092 078 060 060 043 066	782 828 074 046 045 004 0.002	

PRESSURES	PRESSURES section 2			246.21 mm 273.97 mm
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 252 253 254 255	2.00 5.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	-1.109 -1.024 -966 -835 -783 -651 -652 -661 -404 -299 -282 -260 -232 -199 0.342 0.204 0.106 0.081 0.102	1.917 2.247 1.863 2.909 2.838 1.536 1.133 0.702 0.136 0.288 0.240 0.135 003 092 077 050 033 025	1.446 1.785 1.579 0.957 0.138 1.166 0.704 0.302 043 017 089 130 231 318 065 065 0.008

PRESSURES section 3				194.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	985 953 910 865 831 712 644 646 532 366 313 261	1.734 1.763 1.755 1.534 1.438 2.006 1.231 0.390 0.759 0.759 0.511 0.371	1.837 1.844 1.851 1.417 1.359 2.306 1.672 0.658 0.721 0.570 0.099 129

PRESSURE	PRESSURES section 4			144.42 mm 395.32 mm
nr.	x/c [%]	Cp 0	ReCp 2	ImCp 2
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	906878820813798766645525443398360	0.882 0.956 0.904 0.771 0.619 1.405 1.273 0.952 0.659	1.019 1.117 1.066 0.953 0.991 0.948 1.815 2.176 1.325 0.861 0.507

Unsteady Transonic Delta Program

PRESSURES section 5				82.70 mm 269.60 mm
nr. up	y/b [%]	Cp 0	ReCp 2	ImCp 2
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	064 060 076 136 219 334 420 396 310	025 046 146 383 360 0.101 0.795 0.334 092 0.038	037 063 176 436 416 0.059 0.771 0.352 074 0.042

PRESSURE	S section		233.73 mm -60.62 mm	
nr. up	y/b	Cp 0	ReCp 2	ImCp 2
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	650 574 508 511 487 640 860 868 872 903 -1.159	0.499 0.091 0.231 0.227 0.042 358 0.273 0.466 0.666 1.101 1.087	0.405 0.120 0.193 0.210 033 682 0.245 0.476 0.677 1.123 0.824

PRESSURE	S section		117.90 mm 100.71 mm	
nr. up	y/b [%]	Cp 0	ReCp 2	ImCp 2
701 702 703 704 705 109 706 707 208 708 709 307 711 405	22.71 28.21 33.726 44.69 50.03 55.28 60.46 65.56 70.59 75.54 85.22 94.60	653 629 619 552 583 994 626 630 616 630 728 797 798	0.023 -394 -363 0.124 0.260 -088 0.475 0.659 0.702 0.585 1.231 1.679 0.773	0.952 0.415 0.246 0.345 0.273 0.100 0.394 0.350 0.356 0.696 1.672 1.999 1.083 0.991

SECTION COEFFICIENTS						
section	comp.	Zero	Re 2	Im 2		
1	CN_u CN_1 CN_t CM_u	0.578 0.138 0.715 061	283 061 344 178	266 027 294 109		
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u Cm_1	008 069 0.580 0.143 0.723 072	0.014 164 -1.013 050 -1.063 0.010 0.005	001 110 427 020 447 051 009		
3	Cm_t CN_u Cm_u	081 0.571 069	0.015 944 0.080	060 992 0.039		
4	CN_u	0.563	935 0.227	-1.100 0.222		
5	Cm_u CN_u	0.184	0.010 0.018	0.029		
6	Cl_u CN_u Cl_u	118 0.665 343	469 0.267	369 0.210		
7	CN_u CN_u Cl_u	0.632	368 0.302	723 0.393		

Im 3

test conditions

Simple Strake configuration

Zero

angular deflections [deg]

Re 3

alpha = 10.368 deg | Q Mach = 0.901 | Ptot Re\*10^-6 = 8.089 | Ttot

Zero

aerodynamic coefficients

Re 3

= 4.169 deg = 7.600 Hz = 0.067 = 3 dalpha freq k harm

comp.

BALANCE LOADS

position

main	CN Cm Cl	0.6170 0.0094 2457	701478	0.09132 01251 07254	****** 806.18 7732.74	046 165	0.001 005	0.001 005
ACCELER	ATIONS				vibration	mode		
nr	[mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	2.615 1.121 0.880	-86.115 -76.993 132.434 -68.402	1	2.878	0.005	0.016
21 22 23	-138.6 -46.6 121.4	-116.9 -116.9 -116.9 -189.9	0.628 0.866 0.676	88.990 65.753	2	28.034	0.011	0.016
23 31 32 33 41	-74.6 -10.6 141.4 29.4	-189.9 -189.9 -189.9 -304.9	2.179 1.824	67.227 -34.685	3	45.540	0.058	0.019
42 43	89.4 152.4	-304.9 -304.9 -304.9	2.784 2.893 3.191	477 -31.881 62.730	4	73.118	0.088	0.025
51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	2.593 3.420	57.155 58.831	5	89.904	0.119	0.012

Im 3

aero

inertia [%]

Unsteady Transonic Delta Program

PRESSURES	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
101 102 103 104	2.00 5.00 10.00 15.00	-1.176 -1.159 926 820	0.348 0.348 0.301 082	330 041 127 1.068
105 106 107 108 109 110 111 112 113 114 115 151 153 154 155	30.00 40.00 50.00 60.00 79.00 82.50 85.00 90.00 95.00 10.00 40.00 80.00	726 672 601 594 433 244 220 197 157 124 0.359 0.230 0.076 0.056 0.115	399411224214430 0.485 0.449 0.439 0.427 0.378 0.011 0.018 0.021 0.036 0.053	1.508 0.806 - 436 - 772 - 497 0.156 0.135 0.268 0.244 0.162 0.006 0.003 0.009 0.022 0.064

PRESSURES	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 70.00 79.00 82.50 90.00 95.00 10.00 20.00 40.00 80.00	-1.109 -1.024966835783651652649611 -404299282282199 0.342 0.224 0.106 0.081 0.102	123 189 154 712 635 585 688 373 0.086 0.134 0.126 0.174 0.223 0.009 0.015 0.038 0.038	- 867 - 650 - 461 - 988 - 523 0 616 0 409 0 183 - 130 1 006 1 079 1 110 0 007 1 110 0 004 0 021 0 028 0 092

PRESSURES	section	c = 194.13 mm y =-336.06 mm		
nr.	x/c [%]	Cp 0	ReCp 3	ImCp 3
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	985 953 910 865 831 712 664 532 366 313 261	807 775 779 815 811 613 754 435 0.438 0.196 0.343 0.490	-1.463 -1.246 -1.188 -1.383 -1.358 -2.236 -1.176 -358 -1.600 0.605 0.893 1.051

PRESSURES	S section		144.42 mm 395.32 mm	
nr.	x/c [%]	Cp 0	ReCp 3	ImCp 3
401 402 403 404 405 406 407 408 409 410 411	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	906 878 820 813 798 766 645 525 443 398 360	-1.345 -1.294 -1.261 -1.205 -1.234 -1.054 631 292 -366 401 305 177	-1.938 -1.898 -1.656 -1.590 -1.586 -1.522 -2.286 -1.681 -375 0.098 0.323 0.368

DPN = 593

Unsteady Transonic Delta Program

PRESSURES	section		82.70 mm 269.60 mm	
nr. up	y/b [%]	СрО	ReCp 3	ImCp 3
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	064 060 076 136 219 334 420 396 310 318	0.005 0.006 0.009 0.001 025 048 027 0.029 004 0.000	021 024 036 0.039 0.201 0.286 0.011 370 0.088 0.001

PRESSURES section 6 b = 233.73 mm $x = -60.62$ mm					
nr. up	y/b [%]	Cp 0	ReCp 3	ImCp 3	
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	650 574 508 511 487 640 860 868 872 903 -1.159	0.054 008 012 024 0.020 0.066 310 230 155 0.029 0.348	326 045 0.071 0.055 013 456 0.847 0.672 0.540 0.158 041	

PRESSURES	section		17.90 mm 100.71 mm	
nr. up	y/b [%]	Ср О	ReCp 3	ImCp 3
701 702 703 704 705 109 706 707 208 708 708 708 709 307 710 711 405	22.71 28.21 33.72 39.26 44.69 50.03 55.28 60.46 65.56 70.59 75.54 86.42 85.22 90.19 94.60	653 629 611 552 583 626 632 649 616 630 664 728 797	678728474306268214531633688571754753 -1.087	713 976 697 411 450 772 058 0.125 0.183 0.021 142 -1.176 -2.195 -1.420 -1.586

SECTION (	SECTION COEFFICIENTS							
section	comp.	Zero	Re 3	Im 3				
1	CN_u CN_l CN_t CM_u	0.578 0.138 0.715 061	0.033 0.032 0.064 0.022	237 0.027 210 015				
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u Cm_1	008 069 0.580 0.143 0.723 072 010	013 0.009 0.308 0.035 0.342 014	014 028 275 0.039 236 0.225 019				
3	Cm_t CN_u Cm_u	081 0.571 069	028 0.205 0.095	0.205 0.423 0.184				
4	CN_u Cm_u	0.563	0.644	0.912 0.021				
5	CN_u	0.184	0.002	008 0.005				
6	Cl_u CN_u	118 0.665	002 042	0.066				
7	Cl_u CN_u Cl_u	343 0.632 315	0.028 0.651 357	0.025 0.706 401				

BALANCE	LOADS	aerodynamic coefficients		icients aero		angular deflections [deg]		
position	comp.	Zero	Re 4	Im 4	inertia [%]	Zero	Re 4	Im 4
main	CN Cm Cl	0.61701 0.00947 24576	04314 0.00649 0.02598	0.10158 00991 05591	****** 844.84 7077.15	046 165	001 0.002	0.001 003

ACCEL	ERATIONS				vibration	mode		
nr	x (mm)	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	2.389 0.941 0.353	-82.023 -67.016 122.100	1	2.878	0.004	0.008
21	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	0.508	-63.185 106.020	2	28.034	0.002	0.006
23 31 32	-74.6 -10.6	-189.9 -189.9	0.403	7.018	3	45.540	0.014	0.014
33 41 42	141.4 29.4 89.4	-189.9 -304.9 -304.9 -304.9	1.710 1.971 3.145 2.715	58.402 99.268 88.700	4	73.118	0.058	0.019
43 51 52 53	152.4 85.0 121.4 157.4	-374.9 -374.9 -374.9	1.945 2.128 1.681	117.238 124.596 119.318	5	89.904	0.067	0.006

Unsteady Transonic Delta Program

DPN = 593

PRESSURES section 1 $c = 300.65 \text{ mm}$ y = -209.06  mm					
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4	
101 102 103 104	2.00 5.00 10.00 15.00	-1.176 -1.159 926 820	183 295 571 0.153	0.170 0.270 0.424 0.120	
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 79.00 82.50 85.00 90.00 95.00 10.00 40.00 60.00	726 672 601 594 433 244 220 197 124 0.359 0.076 0.056 0.115	211 036 0.177 0.168 0.448 222 172 249 207 174 0.001 0.003 0.007 0.006	185 541 201 063 0.059 0.490 0.499 0.494 002 001 0.002 0.010	

PRESSURES	PRESSURES section 2 $c = 246.21 \text{ mm}$ y = -273.97  mm						
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4			
201 202 203 204 205 206 207 208 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 18.00 30.00 60.00 70.00 79.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	-1.109 -1.024 -966 -835 -783 -651 -652 -649 -611 -404 -299 -282 -260 -232 -199 0.342 0.224 0.106 0.081 0.102	084 0.081 0.009 523 329 0.408 0.256 0.045 214 0.096 0.070 0.053 056 0.001 056 0.000 0.000	616 668 549 0.433 0.157 755 692 437 054 356 332 292 201 068 005 002 004 0.009 0.010			

PRESSURE	section	c = 194.13 mm y =-336.06 mm		
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	985 953 910 865 831 712 664 646 332 366 313 261	0.411 0.407 0.291 0.201 0.159 167 0.718 0.427 176 0.466 0.219	671 694 603 543 435 0.078 030 0.016 163 576 570

PRESSURES	S section		.44.42 mm 95.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	906 878 820 813 798 766 645 525 443 398 360	0.157 0.289 0.314 0.248 0.247 0.135 595 0.170 0.653 0.721 0.543	070 109 094 0.028 0.042 0.214 0.510 0.198 495 495 482

Unsteady Transonic Delta Program

PRESSURES section 5 $b = 82.70 \text{ mm}$ x = -269.60  mm					
nr. up	y/b [%]	СрО	ReCp 4	ImCp 4	
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	064 060 076 136 219 334 420 396 310	003 003 010 041 023 0.056 0.093 058 0.031	0.003 0.003 0.008 0.025 0.014 029 048 0.046 022 0.004	

PRESSURE	S section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 4	ImCp 4
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	650 574 508 511 487 640 860 868 872 903 -1:159	0.312 0.022 0.014 0.025 031 003 0.151 0.279 0.308 061	146 0.000 014 014 0.009 059 035 049 037 0.144 0.270

PRESSURI	ES section		17.90 mm .00.71 mm	
nr.	y/b [%]	Cp 0	ReCp 4	ImCp 4
701 702 703 704 705 109 706 707 208 708 708 709 307 710 711 405	22.71 28.21 33.26 44.69 50.03 55.28 60.46 70.59 75.54 85.22 85.22 94.60	653 629 611 552 583 594 626 630 616 630 728 797 798	0.560 0.258 0.209 0.170 0.117 0.168 0.047 0.092 0.045 0.119 0.408 0.718 231 0.209	727 305 094 186 214 063 301 339 437 265 222 030 0.249 029 0.042

SECTION COEFFICIENTS							
section	comp.	Zero	Re 4	Im 4			
1	CN_u CN_l CN_t CN_t	0.578 0.138 0.715 061 008	0.036 0.004 0.040 0.003 001	079 0.009 070 0.058 005			
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u Cm_1 Cm_t	069 0.580 0.143 0.723 072 010	0.002 010 0.007 003 0.004 002	0.053 0.326 0.003 0.330 059 002			
3	CN_u Cm_u	0.571	211 0.037	0.332			
4	CN_u Cm_u	0.563 079	291 0.118	0.142 104			
5	CN_u Cl_u	0.184	0.000	002 0.001			
6	CN_u Cl_u	0.665 343	119 0.012	0.027 0.020			
7	CN_u Cl_u	0.632 315	280 0.107	0.299			

Im 5

test conditions

Simple Strake configuration

angular deflections [deg]

Re 5

= 24.218 kPa = 72.140 kPa = 307.409 K alpha = 10.368 deg Mach = 0.901 Re\*10^-6 = 8.089 Q Ptot Ttot

Zero

aerodynamic coefficients

Re 5

= 4.169 deg = 7.600 Hz = 0.067 = 5 dalpha freq k harm

comp.

BALANCE LOADS

position

main	CN Off C1	0.6170 0.0094 2457	700190	04855 0.00885 0.03079	2326.70 270.99 1585.64	046 165	0.000 001	001 0.002
ACCELER	ATIONS				vibration	n mode		
nr	[mm]	y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	[%] Y/p	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	4.694 2.359 1.619	27.874 36.402 -146.820	1	2.878	0.004	0.011
21	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	1.412	39.703 -133.801	2	28.034	0.001	0.010
23 31 32 33	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	0.797 1.600	28.502 -110.877	3	45.540	0.007	0.011
41 42 43	29.4 89.4 152.4	-304.9 -304.9 -304.9	1.056 2.289 3.084	-132.425 -111.552 -109.027 -74.986	4	73.118	0.014	0.018
51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	2.743 3.037 2.747	-74.986 -76.258 -98.435	5	89.904	0.064	0.016

Im 5

aero

inertia [%]

DPN = 593

Unsteady Transonic Delta Program

PRESSURES	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
101 102 103 104	2.00 5.00 10.00 15.00	-1.176 -1.159 926 820	202 220 176 125	192 162 067 057
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 80.00	726 672 601 594 433 244 220 197 157 124 0.359 0.230 0.076 0.076	0.385 0.555 240 416 498 239 172 199 209 0.003 0.003 0.001 0.001	083 012 0.088 0.159 0.364 339 247 187 102 0.000 0.001 001

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 5	ImCp 5
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	-1.109 -1.024 -966 -835 -783651652649611404299282260232199 0.342 0.224 0.106 0.081 0.102	215 068 025 0.304 015 0.427 0.409 0.339 0.199 0.345 0.345 0.354 0.354 0.003 0.003 0.010	394185218220021 0.178 0.339 0.328 0.114114129117124134 0.004 0.003005019

PRESSURES	section		194.13 mm 336.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 79.00 90.00	985 953 910 865 831 712 664 532 366 313 261	305 160 051 274 240 412 850 202 0.115 109 0.120 0.294	0.429 0.417 0.436 0.416 0.428 0.335 028 018 099 268 259 264

PRESSURES	section		144.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	906878820813798766645525443398360	- 494 584 468 403 359 313 0 .129 681 500 206 158	0.914 0.890 0.869 0.838 0.689 0.723 211 009 0.120 0.186 0.188

Unsteady Transonic Delta Program

DPN = 593

$\begin{array}{ c c c c c c c c }\hline \text{PRESSURES section 5} & b = 82.70 \text{ mm} \\ x = -269.60 \text{ rm} \\ \hline \\ nr. & y/b & Cp 0 & ReCp 5 & ImCp 5 \\ \hline \\ 501 & 6.62 &064 &001 &003 \\ 502 & 20.43 &060 &001 &003 \\ 503 & 34.05 &076 &004 &005 \\ 504 & 47.67 &136 &002 &003 \\ 505 & 54.49 &219 & 0.008 & 0.002 \\ 506 & 61.29 &334 &009 & 0.002 \\ 507 & 68.10 &420 &025 &008 \\ 508 & 74.91 &396 &020 &009 \\ 509 & 81.72 &310 &015 &006 \\ 510 & 88.53 &318 &003 &004 \\ \hline \end{array}$					
up         [%]           501         6.62        064        001        003           502         20.43        060        001        003           503         34.05        076        004        005           504         47.67        136        002        003           505         54.49        219         0.008         0.002           506         61.29        334        009         0.000           507         68.10        420        025        008           508         74.91        396        020        009           509         81.72        310        015        006	PRESSURES	S section			
502         20.43        060        001        003           503         34.05        076        004        005           504         47.67        136        002        003           505         54.49        219         0.008         0.002           506         61.29        334        009         0.00           507         68.10        420        025        008           508         74.91        396        020        009           509         81.72        310        015        006			Cp 0	ReCp 5	ImCp 5
	502 503 504 505 506 507 508 509	20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72	060 076 136 219 334 420 396 310	001 004 002 0.008 009 025 020	003 005 003 0.002 0.000 008 009

PRESSURE	S section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	650 574 508 511 487 640 860 868 872 903 -1.159	147 0.064 0.015 0.007 0.000 274 0.093 043 123 123 143	051 0.033 0.007 0.011 139 0.116 0.006 073 056 162

PRESSURE	S section		17.90 mm .00.71 mm	
nr.	y/b [%]	Cp 0	ReCp 5	ImCp 5
701 702 703 704 705 109 706 707 208 708 709 307 711 405	22.71 28.21 33.726 44.69 50.09 55.28 60.46 65.56 70.59 75.54 80.42 85.29 90.19 94.60	653 629 611 552 583 594 626 630 664 728 727 797	0.495 -133 -207 -087 -092 -416 0.089 0.244 0.339 0.212 -030 -213 -276 -359	0.528 0.551 0.433 0.300 0.215 0.159 0.291 0.370 0.328 0.240 0.168 028 0.539 0.731 0.838

SECTION COEFFICIENTS						
section	comp.	Zero	Re 5	Im 5		
2	CN_u CN_1 CN_t CN_u CM_1 CM_1 CN_u CN_1	0.578 0.138 0.715 061 008 069 0.580 0.143	0.100 0.003 0.104 060 001 061 254 0.008	0.030 005 0.025 007 0.003 003 023 007		
3 4	CN_1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	0.723 072 010 081 0.571 069 0.563 079	246 0.092 003 0.089 0.140 0.023 0.356 073	030 0.005 0.004 0.010 032 070 391 0.015		
5	CN_u	0.184	0.006	0.004		
6	Cl_u CN_u	118 0.665	004 0.112	0.048		
7	Cl_u CN_u Cl_u	343 0.632 315	057 027 057	029 419 0.209		

Simple Strake configuration

alpha = 22.459 deg | Q = 24.255 kPa Mach = 0.900 | Ptot = 72.335 kPa Re\*10^-6 = 8.080 | Ttot = 308.220 K

dalpha = 8.272 deg freq = 7.600 Hz k = 0.067 harm = 1

BALANCE	LOADS	aerodynamic coefficients			aero	angular deflections [deg]		
position	comp.	Zero	Re 1	Im 1	inertia [%]	Zero	Re 1	Im 1
main	CN Cm Cl	1.05469 0.03674 36892	1.69158 0.13315 46789	0.37099 03980 15778	3746.46 193.01 1103.92	115 226	066 027	0.009

ACCEL	ERATIONS				vibration mode			
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11	-425.6 -215.6	-12.0 -12.0	122.232 61.817	2.986 4.082	1	2.878	1.585	7.237
11 12 13 21 22 23 31 32 33	167.4 -138.6 -46.6	-12.0 -116.9 -116.9	48.858 40.115	175.730 10.969	2	28.034	0.743	7.520
23 31	121.4 -74.6	-116.9 -189.9	37.724 18.145	-171.936 11.651	3	45.540	0.658	6.316
32 33 41	-10.6 141.4 29.4	-189.9 -189.9 -304.9	36.228 2.573	-174.749 -129.807	3	45.540	0.658	
12 13	89.4 152.4	-304.9 -304.9	15.932 23.085	-161.811 -177.825	4	73.118	2.088	4.401
42 43 51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	21.031 39.818 34.522	-167.759 -167.393 -166.339	5	89.904	5.880	4.657

Unsteady Transonic Delta Program

PRESSURES	section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 1	ImCp 1
101 102 103 104 105 106 107 108 109 110 111 112 113 114 115 152 153 154 155	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 79.00 82.50 90.00 90.00 10.00 20.00 40.00 60.00	-1.102 -1.088 -1.048 -1.002 -910 -844 -742 -699 -649 -575 -597 -575 -548 -533 0.662 0.381 0.286 0.286	0.097 036 216 287 137 183 223 176 130 151 151 158 214 340 1.146 1.308 1.356 1.173 0.839	017022019060270410439417404360333314265208 0.093 0.116 0.143 0.160 0.161

PRESSURES	section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	0 qD	ReCp 1	ImCp 1
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 18.00 30.00 40.00 60.00 70.00 79.00 82.50 85.00 95.00 10.00 40.00 80.00	836 796 778 762 737 762 735 735 659 659 659 588 0.638 0.531 0.388 0.290 0.207	0.045 100 058 171 260 277 251 186 099 106 078 059 1.059 1.259 1.089 0.788	464 457 462 367 433 438 471 394 394 349 328 0.105 0.122 0.136 0.155 0.150

DPN = 605

PRESSURES	S section		194.13 mm 336.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 1	ImCp 1
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 90.00	654 674 663 642 629 615 631 606 609 593	108 178 077 141 147 157 154 148 119 085 063 052	706 757 592 468 408 409 428 419 415 408

PRESSURES	section		144.42 mm 395.32 mm	
nr.	x/c [%]	Cp 0	ReCp 1	ImCp 1
401 402 403 404 405 406 407 408 409 410 411	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	545 547 530 523 518 530 541 539 542 532 526	0.084 0.080 0.070 0.078 0.083 0.112 0.123 0.140 0.145 0.155 0.137	373 366 353 345 345 361 377 392 399 400 385 373

Unsteady Transonic Delta Program

PRESSURE	PRESSURES section 5			82.70 mm 269.60 mm
nr. up	[#] A\p	Cp 0	ReCp 1	ImCp 1
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	305 384 561 809 932 866 746 661 625 637	-1.353 -1.811 -2.500 -2.734 -2.538 -1.428 -1.246 -1.603 -1.532 -1.371	0.027 0.070 0.146 0.212 0.232 0.109 0.050 0.081 0.077

PRESSURE	S section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 1	ImCp 1
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.065 999 919 900 889 -1.127 -1.241 -1.201 -1.154 -1.098 -1.088	127 -1.012 -1.073 -1.103 -1.237 0.334 0.710 0.241 0.129 018 036	342 222 183 190 138 006 0.267 0.350 0.238 0.122 022

PRESSURES	S section		117.90 mm 100.71 mm	
nr.	y/b [%]	Cp 0	ReCp 1	ImCp 1
701 702 703 704 705 109 706 707 208 708 708 709 710 711 405	22.71 28.21 33.726 44.69 50.08 60.466 67.59 75.542 85.22 90.19 94.60	637 696 705 617 670 699 699 709 665 645 600 575 518	1.414 0.488 0.173 269 420 223 130 187 251 205 182 154 032 0.083	- 337 - 318 - 223 - 255 - 373 - 417 - 449 - 444 - 471 - 425 - 422 - 409 - 394 - 345

SECTION (	COEFFICI	ents		
section	comp.	Zero	Re 1	Im 1
1	CN_u CN_l CN_t CM_u	0.770 0.361 1.132 134	0.202 1.126 1.327 056	0.280 0.140 0.421 094
2	Cm_1 Cm_t CN_u CN_1 CN_t CM_u Cm_u	037 171 0.675 0.352 1.028 141 035	237 292 0.168 1.059 1.227 034 220	042 136 0.416 0.137 0.554 039 136
3	Cm_t CN_u Cm_u	175 0.588 127	255 0.113 019	0.447
4	CN_u Cm_u	0.506 114	123 0.038	0.376
5	CN_u Cl_u	0.572	1.824	100 0.051
6	CN_u Cl_u	1.001	0.241	0.139
7	CN_u Cl_u	0.631	291 003	0.367

Simple Strake configuration

alpha = 22.459 deg | Q = 24.255 kPa Mach = 0.900 | Ptot = 72.335 kPa Re\*10^-6 = 8.080 | Ttot = 308.220 K

dalpha = 8.272 deg freq = 7.600 Hz k = 0.067 harm = 2

BALANCE LOADS		aerodynamic coefficients			aero	angular defl	ections [deg]	
position	comp.	Zero	Re 2	Im 2	inertia [%]	Zero	Re 2	Im 2
main	CN Cm C1	1.05469 0.03674 36892	42244 0.01045 0.16732	25033 00396 0.08876	****** 1399.47	115 226	0.000 0.016	0.003

ACCEL	ERATIONS				vibration	mode		
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	12.114 6.001 7.985	-145.798 -152.525 33.708	1	2.878	0.177	0.209
21 22	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	2.580	-107.085 83.505	2	28.034	0.038	0.128
11 12 13 21 22 23 31 32 33	-74.6 -10.6 141.4	-189.9 -189.9 -189.9	1.207	-115.326 161.230	3	45.540	0.127	0.070
41	29.4 89.4 152.4	-304.9 -304.9 -304.9	1.937 1.472 6.163	65.615 140.436 118.983	4	73.118	0.200	0.265
43 51 52 53	85.0 121.4 157.4	-374.9 -374.9 -374.9	2.434 1.268 3.412	96.225 139.549 105.745	5	89.904	0.029	0.095

DPN = 605

Unsteady Transonic Delta Program

PRESSURE	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
101 102 103 104 105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	2.00 5.00 10.00 15.00 30.00 40.00 60.00 79.00 82.50 90.00 90.00 95.00 10.00 20.00 40.00 60.00	-1.102 -1.088 -1.048 -1.002 -910 844 742 699 605 597 575 548 533 0.662 0.381 0.286	004 0.061 0.117 0.174 0.464 0.621 0.594 0.527 0.382 0.303 0.253 0.149 0.078 034 022 006 0.012	073 0.009 0.102 0.164 0.314 0.234 0.246 0.220 0.179 0.127 0.057057057039015

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 2	ImCp 2
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	- 836 - 796 - 7778 - 7737 - 762 - 735 - 7059 - 6559 - 6559 - 607 - 609 - 558 - 577 - 558 0 638 0 .531 0 .290 0 .207	0.716 0.713 0.702 0.692 0.579 0.675 0.664 0.679 0.469 0.447 0.350 0.281 035 026 001 0.009	0.289 0.249 0.262 0.000 146 0.244 0.207 0.180 0.169 0.173 0.164 0.154 0.135 068 068 068

PRESSURES section 3				194.13 mm 336.06 mm
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
301 302 303 304 305 306 307 308 309 310 311	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	654 674 663 623 629 615 631 606 609 593 579	0.859 0.920 0.756 0.649 0.623 0.552 0.537 0.541 0.467 0.467	0.087 0.066 0.132 0.179 0.172 0.142 0.126 0.117 0.125 0.136 0.148 0.151

PRESSURES	S section		144.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 2	ImCp 2
401 402 403 404 405 406 407 408 409 410 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	545 547 530 523 518 530 541 539 542 532 526	0.524 0.516 0.487 0.471 0.468 0.475 0.478 0.470 0.447 0.407 0.348 0.278	0.236 0.245 0.235 0.235 0.229 0.225 0.217 0.210 0.195 0.147 0.106

Unsteady Transonic Delta Program

DPN = 605

up         [%]           501         6.62        305        039        502        384         0.094	
up     [%]       501     6.62    305    039       502     20.43    384     0.094	
502   20.43  384   0.094   0	nCp 2
504 47.67809 0.421 0.505 54.49932 0.315 0.506 61.29866 0.029 0.507 68.10746068 0.508 74.91661 0.001 0.509 81.72625027	0.051 0.084 0.282 0.459 0.359 0.066 051 0.025 001

PRESSURES	section		233.73 mm -60.62 mm	
nr. up	[#]	Cp 0	ReCp 2	ImCp 2
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.065 999 919 900 889 -1.127 -1.241 -1.201 -1.154 -1.098 -1.088	1.023 1.100 0.864 0.836 0.642 0.147 162 386 198 021	0.563 0.772 0.585 0.502 0.390 0.344 0.013 055 020 0.009

PRESSURES	section		17.90 mm .00.71 mm	
nr.	k] [#]	Ср 0	ReCp 2	ImCp 2
701 702 703 704 705 109 706 707 208 708 709 307 711 405	22. 71 28. 21 33. 726 44. 69 55. 28 60. 46 65. 56 70. 59 75. 24 80. 42 85. 22 94. 60	637 696 705 617 670 699 692 709 665 643 615 500 575 518	137 0.059 0.192 0.280 0.523 0.594 0.620 0.643 0.578 0.553 0.553 0.505 0.468	0.351 0.400 0.262 0.168 0.246 0.194 0.190 0.136 0.117 0.126 0.158 0.198

SECTION (	SECTION COEFFICIENTS						
section	comp.	Zero	Re 2	Im 2			
1	CN_u CN_1 CN_t	0.770 0.361 1.132	375 0.001 373	178 027 204			
2	Cm_u Cm_t Cm_t CN_u CN_1 CN_t CM_t Cm_1	134 037 171 0.675 0.352 1.028 141 035	0.101 007 0.094 576 0.001 574 0.111 007	0.041 003 0.038 161 030 191 0.041 003			
3	Cm_t CN_u	175 0.588 -,127	0.104 554 0.106	0.038 136 0.036			
4	Cm_u CN_u Cm u	0.506	426 0.086	191 0.036			
5	CN_u Cl_u	0.572	098 0.038	115 0.051			
6	CN_u Cl_u	1.001	583 0.165	366 0.119			
7	CN_u Cl_u	0.631	321 0.234	243 0.102			

est con	aıt	ions				Simple Strake configuration
alpha Mach Re*10^-6	=	22.459 0.900 8.080	deg	Q Ptot Ttot	= 24.255 kPa = 72.335 kPa = 308.220 K	
dalpha freq k harm	11 11 11 11	8.272 7.600 0.067				

BALANCE	LOADS	aerodynamic coefficients			aero	angular defl	ections [deg]	
position	comp.	Zero	Re 3	Im 3	inertia [%]	Zero	Re 3	Im 3
main	CN Cm Cl	1.05469 0.03674 36892	06585 00499 0.02737	06785 01212 0.02870	3233.63 287.74 1401.68	115 226	0.001	0.004

ACCELERATIONS					vibration	mode		
nr	[mm]	[mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	7.404 3.655 1.990	-120.501 -126.180 13.850	1	2.878	0.031	0.043
21 22	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	2.053 1.673	-95.724 97.007	2	28.034	0.011	0.040
31 32	-74.6 -10.6	-189.9 -189.9	1.560	-99.039	3	45.540	0.040	0.029
33 41 42	141.4 29.4 89.4	-189.9 -304.9 -304.9	0.866 1.583 2.880	124.337 138.482 -178.237	4	73.118	0.105	0.101
11 12 13 21 22 22 23 31 32 33 41 42 43 551 552	152.4 85.0 121.4	-304.9 -374.9 -374.9	4.778 1.208 0.900 3.253	-153.408 -153.272 159.631 -132.429	5	89.904	0.144	0.084

Unsteady Transonic Delta Program

PRESSURES	section		00.65 mm 09.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
101 102 103 104 105 106 107 108 109 110 111 112 113 114 115 152 153 154 155	2.00 5.00 10.00 15.00 30.00 40.00 50.00 70.00 79.00 85.00 90.00 10.00 40.00 40.00 60.00	-1.102 -1.048 -1.048 -1.002 -910 -844 -742 -699 -699 -597 -575 -5748 -533 0.662 0.381 0.286 0.286	028 012 0.006 0.037 0.088 0.177 0.139 0.050 0.050 0.031 0.029 0.046 003 004 008	0.066 0.050 0.077 0.147 0.274 0.275 0.183 0.036 109 146 170 152 0.001 004 011

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 3	ImCp 3
201 202 203 204 205 206 207 208 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 95.00 10.00 20.00 40.00 80.00	836 796 778 737 762 738 735 659 659 659 659 577 558 577 558 0.531 0.388 0.290	0.379 0.301 0.280 0.337 0.285 0.223 0.182 0.145 0.047 0.043 0.037 0.016 006 008 013	0.417 0.422 0.375 0.131 0.029 0.363 0.304 0.269 0.171 018 059 088 151 208 0.000 007 007

PRESSURE	S section		194.13 mm 336.06 mm	
nr. up	x/c [%]	Cp 0	ReCp 3	ImCp 3
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 79.00 90.00	654 663 642 629 615 631 609 593 579	0.249 0.278 0.217 0.156 0.151 0.128 0.116 0.108 0.084 0.030 0.005	0.519 0.583 0.413 0.256 0.235 0.190 0.168 0.152 0.117 0.068 0.020 054

PRESSURES	section		144.42 mm 395.32 mm	
nr. up	x/c [%]	Cp 0	ReCp 3	ImCp 3
401 402 403 404 405 406 407 408 409 411 412	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 90.00	545 547 530 523 518 530 541 539 542 532 526	0.025 0.016 0.006 001 014 023 035 044 058 070	0.055 0.044 0.033 0.021 0.017 0.010 012 056 104 160 201

Unsteady Transonic Delta Program

PRESSURES	section		82.70 mm 269.60 mm	
nr. up	y/b [%]	Cp 0	ReCp 3	ImCp 3
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	305 384 561 809 932 866 746 661 625 637	0.025 0.005 0.004 0.023 0.034 0.015 003 0.008 0.012	070 0.071 0.090 0.021 0.082 0.104 0.159 0.062 007 039

PRESSURI	PRESSURES section 6			233.73 mm -60.62 mm
nr.	[#] A\p	Cp 0	ReCp 3	ImCp 3
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.065 999 919 900 889 -1.127 -1.241 -1.201 -1.154 -1.098 -1.088	0.352 0.256 0.215 0.225 0.141 0.173 0.036 125 080 041 012	0.018 0.161 0.257 0.297 0.282 -436 -414 -244 -089 0.027 0.050

PRESSURES section 7				117.90 mm 100.71 mm
nr. up	[#] À\p	Cp 0	ReCp 3	ImCp 3
701 702 703 704 705 109 706 707 708 708 709 709 711 405	22.71 28.21 33.72 39.26 44.69 50.03 50.46 65.56 70.59 75.54 85.22 90.19 94.60	637 696 705 617 670 699 692 709 665 645 600 575 518	278020 0.006 0.029 0.074 0.139 0.171 0.184 0.182 0.172 0.151 0.073 0.031004	825 429 157 0.142 0.202 0.183 0.191 0.224 0.269 0.220 0.203 0.108 0.126 0.017

SECTION (	SECTION COEFFICIENTS						
section	comp.	Zero	Re 3	Im 3			
1	CN_u CN_l CN_t CM_u	0.770 0.361 1.132 134	085 006 090 0.024 0.001	093 015 108 007			
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_1 Cm_1	037 171 0.675 0.352 1.028 141 035	0.026 169 009 178 0.012 0.002	001 149 016 166 008 0.007			
3	Cm_t CN_u Cm_u	175 0.588 127	0.015 102 0.005	0.000 155 005			
4	CN_u	0.506 114	0.037 019	0.080			
5	Cm_u CN_u Cl_u	0.572	014 0.007	036 0.019			
6	CN_u	1.001	~.187 0.050	014 003			
7	Cl_u CN_u Cl_u	479 0.631 302	0.035 0.034	0.139 0.029			

test conditions

alpha = 22.459 deg | Q = 24.255 kPa | Mach = 0.900 | Ptot = 72.335 kPa | Re\*10^-6 = 8.080 | Ttot = 308.220 K | dalpha = 8.272 deg | freq = 7.600 Hz | k = 0.067 | harm = 4

BALANCE	LOADS	aerodynamic coefficients			aero	angular defl	ections [deg]	ons [deg]	
position	comp.	Zero	Re 4	Im 4	inertia [%]	Zero	Re 4	Im 4	
main	CN Ch Cl	1.05469 0.03674 36892	01766 0.00233 0.01035	0.07405 01090 03959	8887.86 835.79 4936.96	115 226	001 0.001	0.003 004	

ACCELI	ERATIONS				vibration	mode		
nr	[mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12 13	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	3.242 1.941 1.495	-79.491 -93.905 -116.243	1	2.878	0.040	0.007
21 22	-138.6 -46.6	-116.9 -116.9	0.896	-45.947 -159.205	2	28.034	0.011	0.006
23 31 32	121.4 -74.6 -10.6	-116.9 -189.9 -189.9	0.251 0.036	-147.871 110.769	3	45.540	0.016	0.013
33 41 42	141.4 29.4 89.4	-189.9 -304.9 -304.9	1.741 1.651 2.477	123.225 112.313 -99.362	4	73.118	0.096	0.030
12 13 21 22 23 31 32 33 41 42 43 551 552	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	0.907 1.103 2.710 0.674	55.075 111.890 135.777	5	89.904	0.080	0.026

Unsteady Transonic Delta Program

PRESSURE	S section		300.65 mm 209.06 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
101 102 103 104	2.00 5.00 10.00 15.00	-1.102 -1.088 -1.048 -1.002	043 0.005 0.086 0.139	024 057 142 206
105 106 107 108 109 110 111 112 113 114 115 151 152 153 154 155	30.00 40.00 50.00 60.00 70.00 79.00 82.50 85.00 90.00 95.00 10.00 20.00 40.00 80.00	910 844 742 699 649 597 575 548 533 0 .662 0 .381 0 .286	0.123 0.002 010 0.012 0.002 009 019 053 085 0.004 0.005 0.004	061 050 052 069 120 069 047 0.027 0.092 005 006 006 006

PRESSURE	S section		246.21 mm 273.97 mm	
nr. up low	x/c [%]	Cp 0	ReCp 4	ImCp 4
201 202 203 204 205 206 207 208 207 210 211 212 213 214 215 252 253 254 255	2.00 5.00 10.00 15.00 15.00 16.00 50.00 60.00 79.00 82.50 90.00 95.00 90.00 95.00 40.00 40.00 80.00	836 796 778 737 762 738 709 655 607 609 588 577 588 0.531 0.388 0.290 0.207	0.193 0.153 0.153 0.062 0.027 0.049 0.031 0.012 0.014 0.050 0.050 0.049 0.030 0.002 0.003 0.003	155 074 053 086 064 053 048 093 132 207 204 191 170 127 004 006 011 009 010

PRESSURES section 3			c = 194.13 mm y =-336.06 mm		
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4	
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 30.00 40.00 50.00 70.00 79.00 90.00	654 674 663 642 629 615 631 606 609 593	097 145 007 0.030 0.032 0.023 0.020 0.009 0.009 0.019 0.035	134 156 156 132 130 131 138 151 164 179 191 189	

PRESSURE	PRESSURES section 4			144.42 mm 395.32 mm
nr. up	x/c [%]	Cp 0	ReCp 4	ImCp 4
401 402 403 404 405 406 407 408 409 410 411	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	545 547 530 523 518 530 541 539 542 532 526 526	0.049 0.057 0.053 0.057 0.058 0.059 0.057 0.055 0.057 0.057	199207202209234258276264242214

Unsteady Transonic Delta Program

PRESSURES	section		82.70 mm 269.60 mm	
nr. up	/g (%)	Cp 0	ReCp 4	ImCp 4
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	305 384 561 809 932 866 746 661 625 637	002 0.007 0.043 0.040 016 063 0.025 0.003 020 043	0.009 0.001 020 024 0.014 0.054 011 0.003 0.015 0.031

PRESSURE	S section		233.73 mm -60.62 mm	
nr. up	y/b [%]	Cp 0	ReCp 4	ImCp 4
601 602 603 604 605 606 607 608 609 610 102	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.065 999 919 900 889 -1.127 -1.241 -1.201 -1.154 -1.098 -1.088	047 0.175 0.143 0.132 0.143 173 251 127 031 0.006 0.005	0.148 0.021 0.038 0.046 0.186 0.053 061 088 109 057

PRESSURES	PRESSURES section 7			17.90 mm 00.71 mm
nr.	y/b [%]	Cp 0	ReCp 4	ImCp 4
701 702 703 704 705 109 706 707 208 708 709 307 711 405	22.71 28.21 33.726 44.69 55.28 60.46 65.56 70.59 75.54 80.42 85.22 90.19 94.60	637 696 705 617 670 699 699 709 665 643 615 575 518	0.206 0.038 027 084 039 0.012 0.011 0.012 0.014 0.020 0.028 0.044 0.058	154 142 175 067 059 097 099 093 102 138 122 211 209

SECTION (	COEFFICIE	ents		
section	comp.	Zero	Re 4	Im 4
1	CN_u CN_1 CN_t CM_u	0.770 0.361 1.132 134	023 0.003 019 008 001	0.069 006 0.063 006 0.001
2	Cm_1 Cm_t CN_u CN_1 CN_t Cm_u Cm_1	171 0.675 0.352 1.028 141 035	008 049 0.003 045 0.005 001	005 0.113 009 0.104 039 0.003
3	Cm_t CN_u Cm_u	175 0.588 127	0.004 008 0.008	037 0.159 046
4	CN_u	0.506	056	0.238
5	Cm_u CN_u Cl_u	114 0.572 305	0.014 0.001 005	062 006 0.006
6	CN_u	1.001	0.014	059 0.007
7	Cl_u CN_u Cl_u	479 0.631 302	059 0.015	0.133

test conditions

alpha = 22.459 deg | Q = 24.255 kPa | Mach = 0.900 | Ptot = 72.335 kPa | Re\*10^-6 = 8.080 | Ttot = 308.220 K |

dalpha = 8.272 deg | freq = 7.600 Hz | k = 0.067 | harm = 5

BALANCE	LOADS	aerodynamic coefficients		aero	angular defl	ections [deg]		
position	comp.	Zero	Re 5	Im 5	inertia [%]	Zero	Re 5	Im 5
main	CN Cm Cl	1.05469 0.03674 36892	0.02340 0.00165 00974	0.04377 00179 02117	3721.21 117.22 1805.53	115 226	001 001	0.000

ACCEL	ERATIONS				vibration	n mode		
nr	x [mm]	Y [mm]	Amplitude [m/s^2]	Phase angle rel. to LVDT [deg]	section	y/b [%]	heave at p.a [mm]	pitch [deg]
11 12	-425.6 -215.6 167.4	-12.0 -12.0 -12.0	3.257 1.212 1.876	13.865 -5.472 -163.577	1	2.878	0.013	0.009
11 12 13 21 22 23 31 32 33 41 42	-138.6 -46.6 121.4	-116.9 -116.9 -116.9	1.300 0.683	31.600 -113.879	2	28.034	0.006	0.007
23 31 32	-74.6 -10.6	-189.9 -189.9	1.006	28.556 155.372	3	45.540	0.010	0.007
33 41 42	141.4 29.4 89.4	-189.9 -304.9 -304.9	0.584 1.088 1.406	84.414 100.927	4	73.118	0.033	0.010
43 51 52 53	152.4 85.0 121.4 157.4	-304.9 -374.9 -374.9 -374.9	0.625 1.504 1.532 1.989	174.219 49.481 105.067 117.477	5	89.904	0.052	0.028

DPN = 605

Unsteady Transonic Delta Program

PRESSURES section 1			c = 300.65 mm y =-209.06 mm	
nr. up low	x/c [%]	0 qD	ReCp 5	ImCp 5
101 102 103 104 105 106 107 108 109 110 111 112 113 114 115 151 153 154 155	2.00 5.00 10.00 15.00 30.00 40.00 50.00 79.00 79.00 82.50 90.00 95.00 90.00 95.00 96.00 96.00 96.00 96.00	-1.102 -1.088 -1.048 -1.002 910 844 742 605 597 575 548 533 0.662 0.381 0.286	122 129 156 194 087 015 0.025 0.023 0.014 0.005 0.005 0.003 0.003 0.003	0.197 0.168 0.101 0.062 041 072 097 082 060 0.009 0.045 0.057 0.058 0.051 001 0.003 0.003

PRESSURES section 2				246.21 mm 273.97 mm
nr. up low	x/c [%]	Ср 0	ReCp 5	ImCp 5
201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 251 252 253 254 255	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 70.00 79.00 82.50 85.00 10.00 20.00 40.00 80.00	836 796 778 737 762 738 735 659 659 659 659 588 588 558 0.638 0.290 0.207	172130132138090065052016 0.008 0.024 0.007006040071 0.005 0.005 0.008 0.004	202166157008 0.011141145140120047040028008 0.013001 0.000 0.002 0.004 0.005

PRESSURES section 3				194.13 mm 136.06 mm
nr. up	x/c [%]	Cp 0	ReCp 5	ImCp 5
301 302 303 304 305 306 307 308 309 310 311 312	2.00 5.00 10.00 15.00 30.00 40.00 50.00 60.00 70.00 79.00	654 674 663 642 629 615 631 606 609 593 579	0.020 0.038 014 015 016 0.004 0.008 0.013 0.028 0.041 0.041	- 281 - 301 - 198 - 123 - 121 - 121 - 116 - 115 - 098 - 071 - 041 - 008

PRESSURES	PRESSURES section 4			144.42 mm 395.32 mm
nr. up	x/c [%]	Ср 0	ReCp 5	ImCp 5
401 402 403 404 405 406 407 408 409 410 411	2.00 5.00 10.00 15.00 18.00 30.00 40.00 50.00 60.00 70.00 79.00 90.00	545 547 530 523 518 530 541 539 542 532 526	0.065 0.068 0.060 0.057 0.056 0.061 0.055 0.036 0.015 005	091 085 075 068 076 079 071 050 027 003

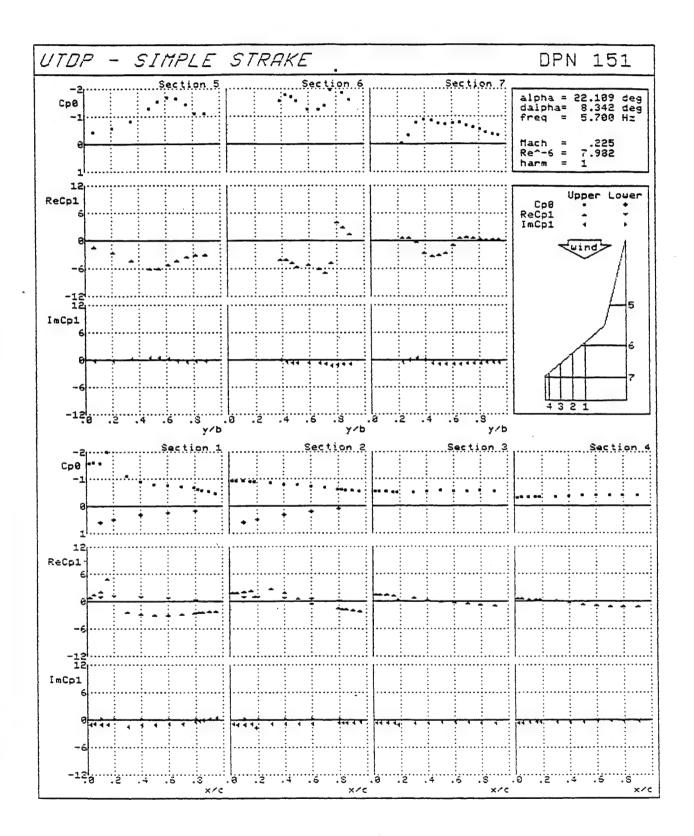
Unsteady Transonic Delta Program

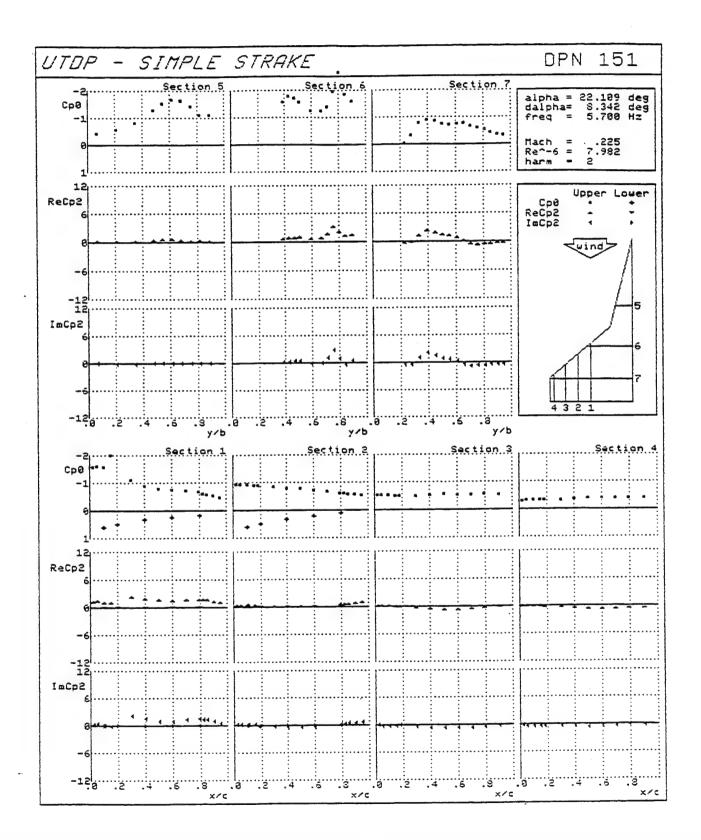
PRESSURES section 5				82.70 mm 269.60 mm
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
501 502 503 504 505 506 507 508 509 510	6.62 20.43 34.05 47.67 54.49 61.29 68.10 74.91 81.72 88.53	305 384 561 809 932 746 661 625 637	023 0.003 014 011 0.047 0.051 0.025 002 009 006	002 0.009 001 001 0.008 0.025 0.010 0.005 0.005

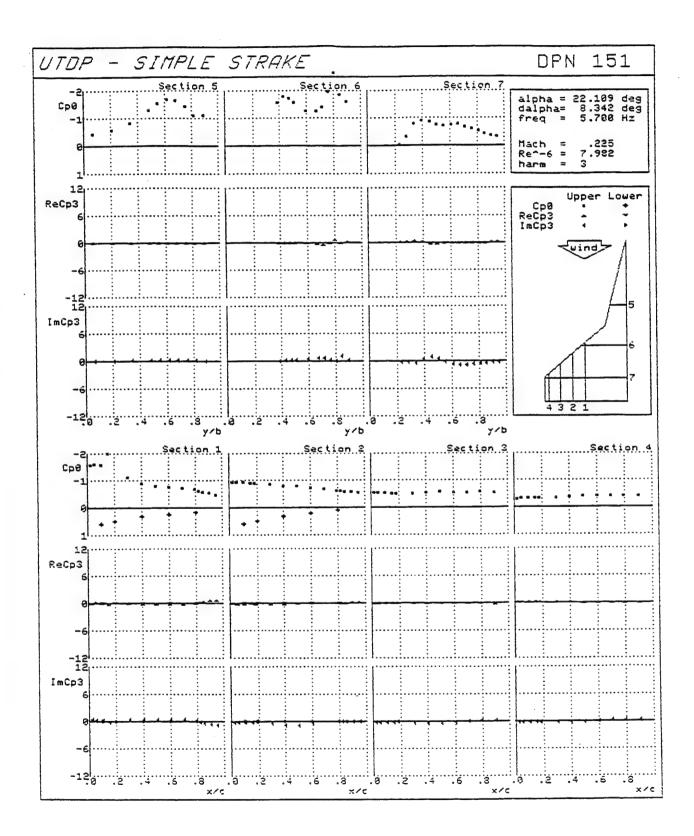
PRESSURE	PRESSURES section 6			233.73 mm -60.62 mm
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
601 602 603 604 605 606 607 608 609 610	38.90 42.93 46.93 50.99 59.03 67.07 71.11 75.56 80.00 84.44 89.45	-1.065 999 919 900 889 -1.127 -1.241 -1.201 -1.154 -1.098	0.021 064 012 0.000 0.046 0.041 189 157 151 182 129	034 087 055 047 0.031 0.093 0.141 0.196 0.201 0.203 0.168

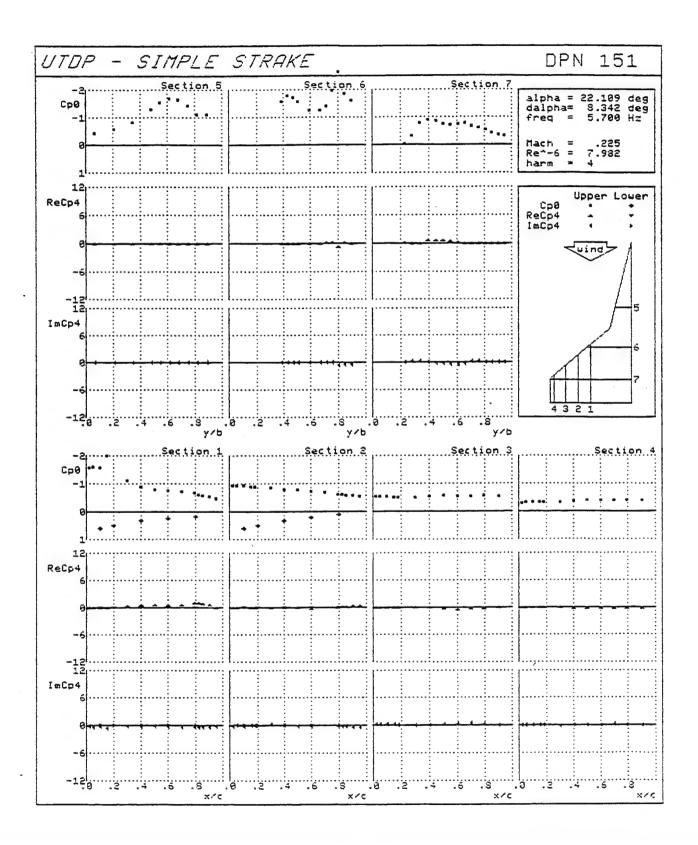
PRESSURES section 7				17.90 mm 100.71 mm
nr. up	y/b [%]	Cp 0	ReCp 5	ImCp 5
701 702 703 704 705 109 706 707 208 708 708 709 307 711 405	22.71 28.21 33.726 44.69 50.03 65.528 60.46 65.56 70.59 75.542 85.22 90.19	637 696 7617 617 699 699 709 665 643 600 575 518	0.004 079 087 0.044 0.066 0.025 0.000 011 016 023 014 0.031 0.048 0.056	0.180 0.160 0.130 0.037 041 082 110 125 116 116 102 068

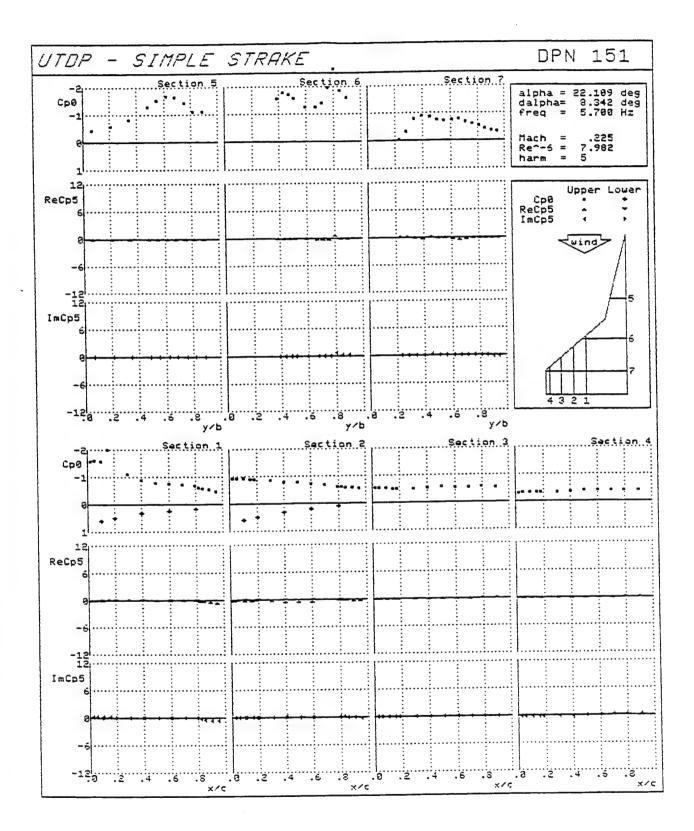
SECTION (	COEFFICI	ents		
section	comp.	Zero	Re 5	Im 5
1	CN_u CN_1 CN_t CM_u	0.770 0.361 1.132 134	0.054 0.004 0.058 0.008	0.000 0.002 0.002 006
2	Cm_1 Cm_t CN_u CN_1 CN_t CM_u Cm_u Cm_1	037 171 0.675 0.352 1.028 141 035	001 0.006 0.047 0.006 0.053 0.000 001	001 008 0.089 0.003 0.092 012
3	Cm_t CN_u Cm_u	175 0.588 127	002 015 0.007	014 0.104 007
4	CN_u	0.506	033	0.045
5	Cm_u CN_u	0.572	001 001	001 005
6	Cl_u CN_u	305 1.001	0.002 0.036	0.003 041
7	Cl_u CN_u Cl_u	479 0.631 302	035 004 0.006	0.045 004 ~.031

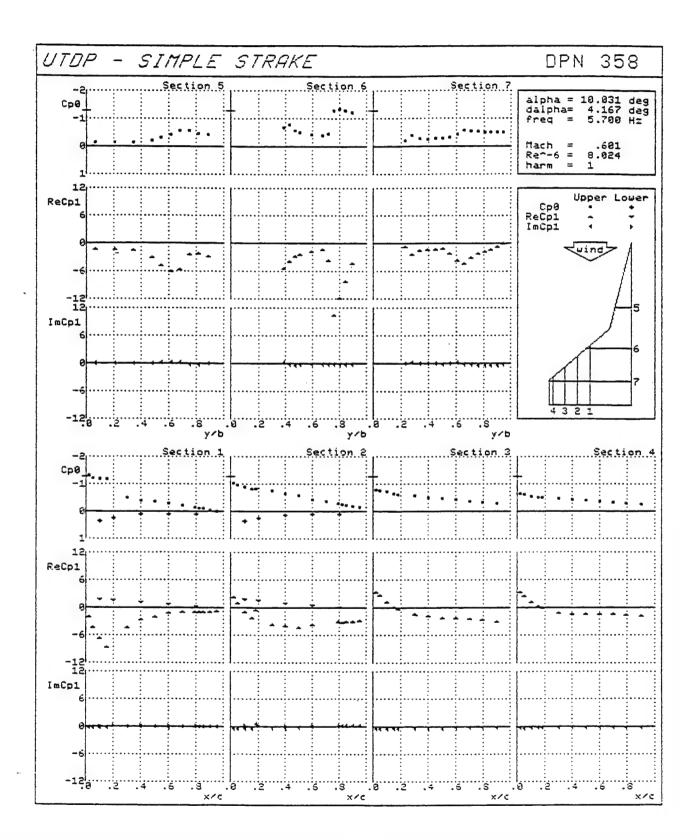


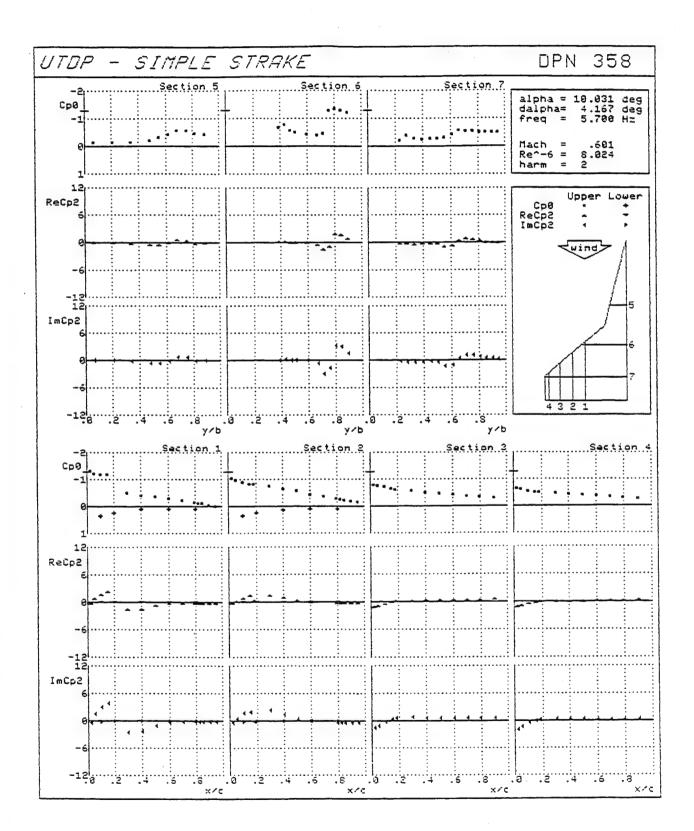


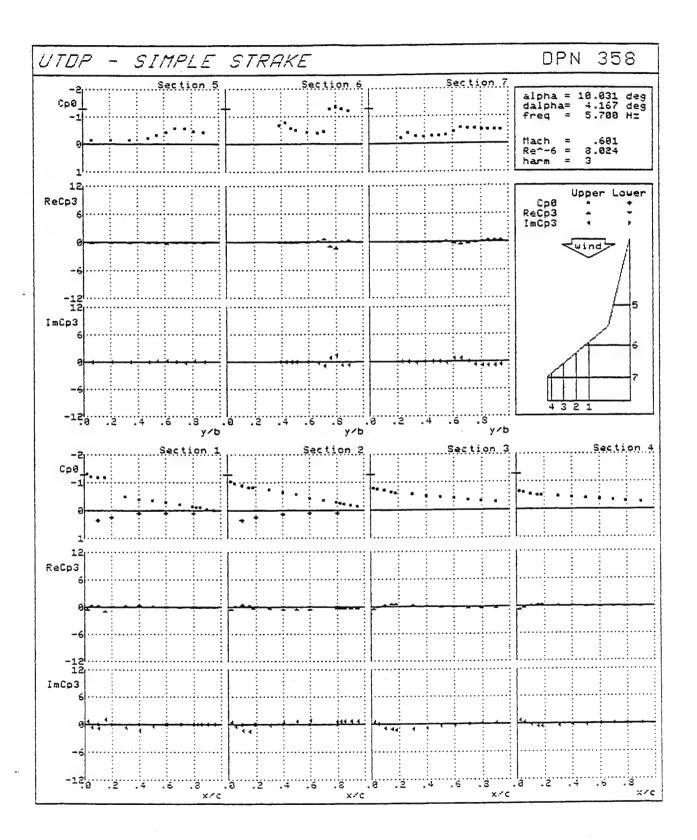


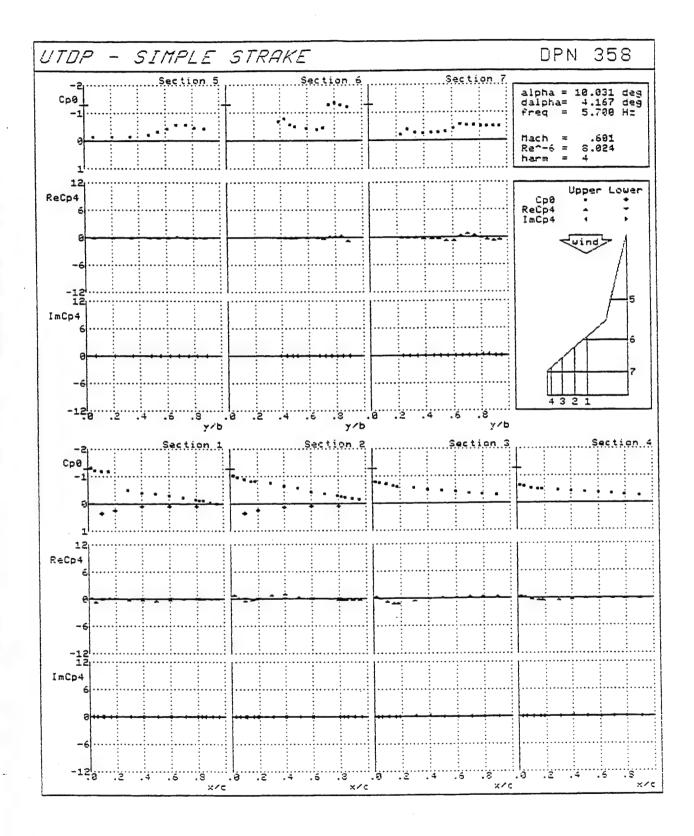


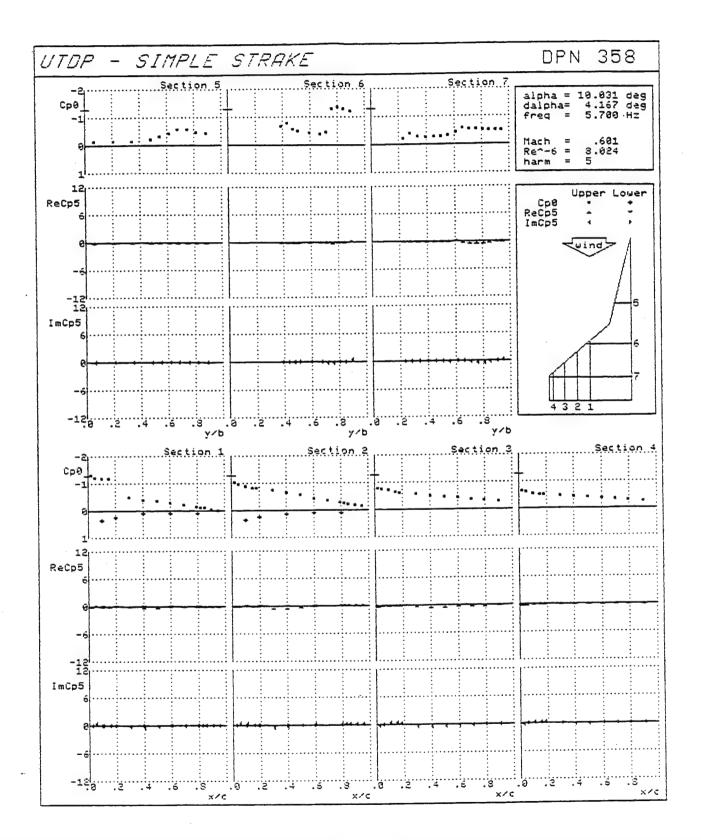


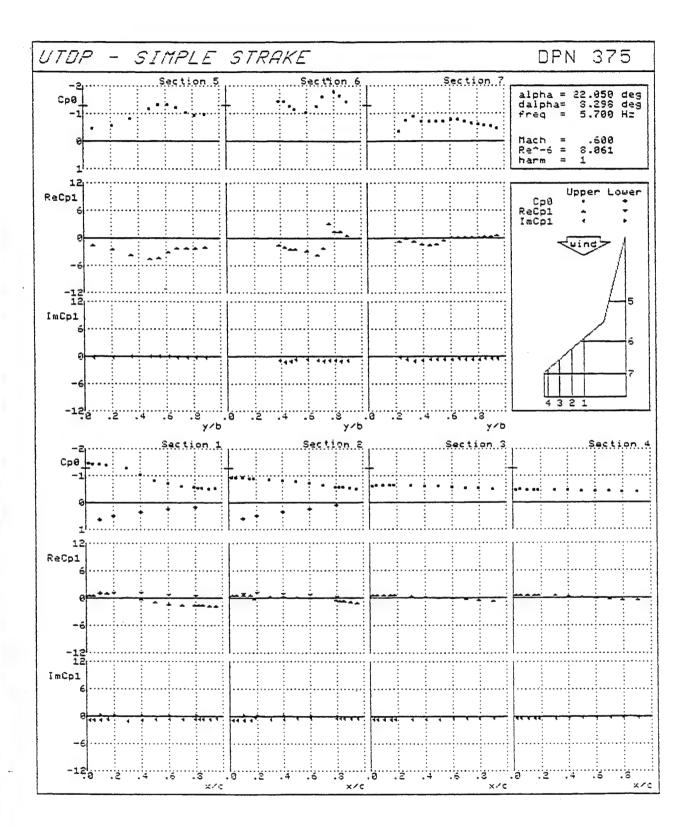


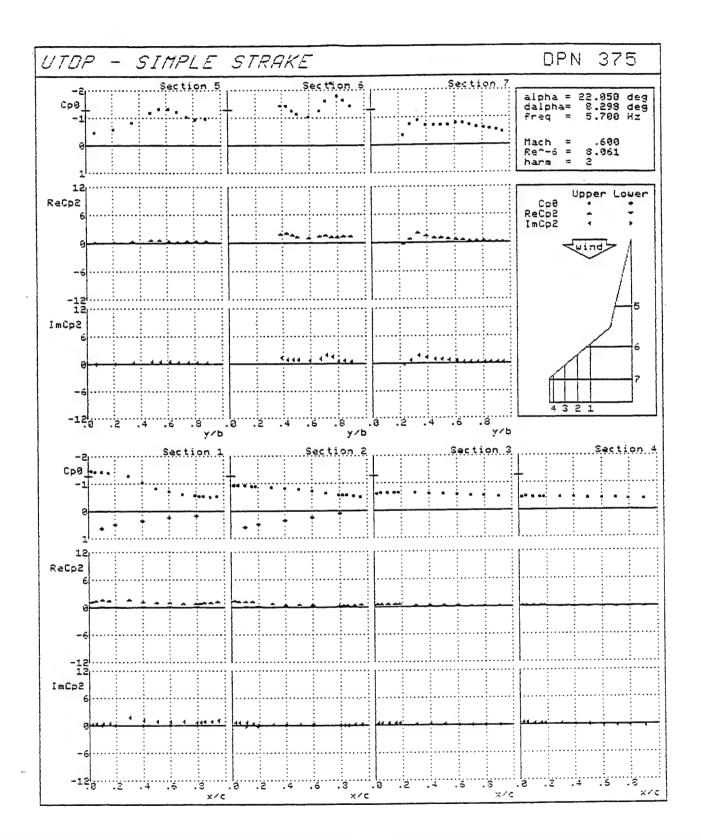


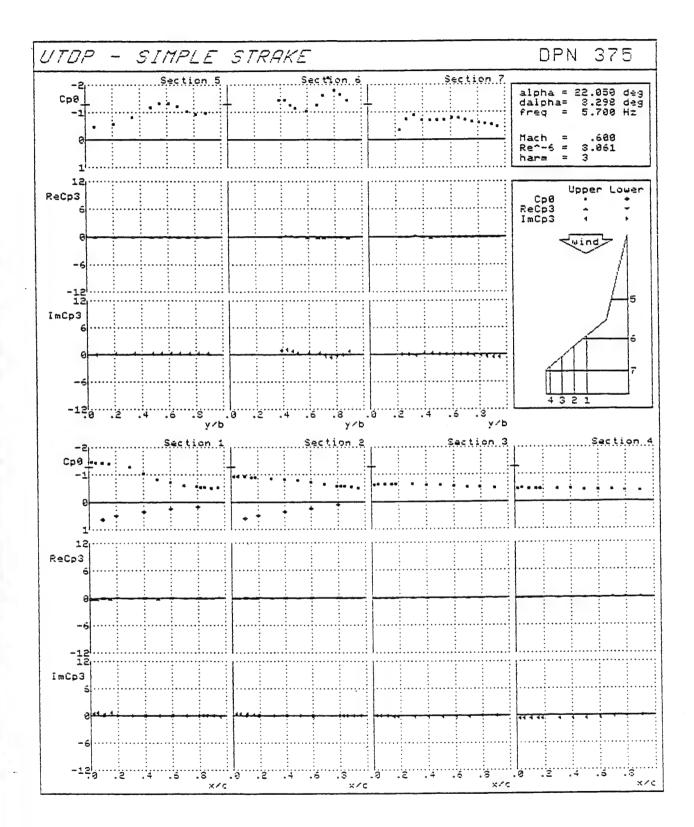


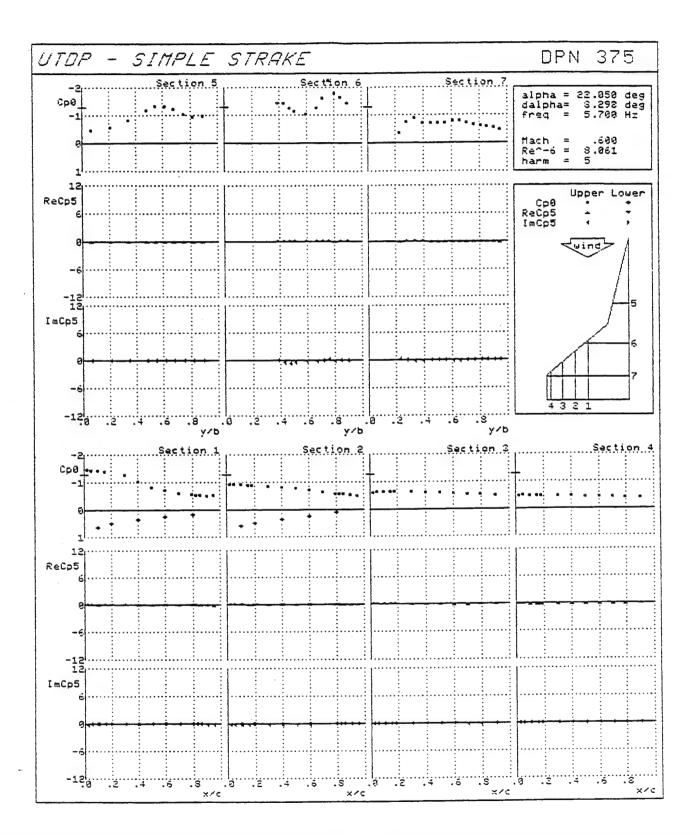


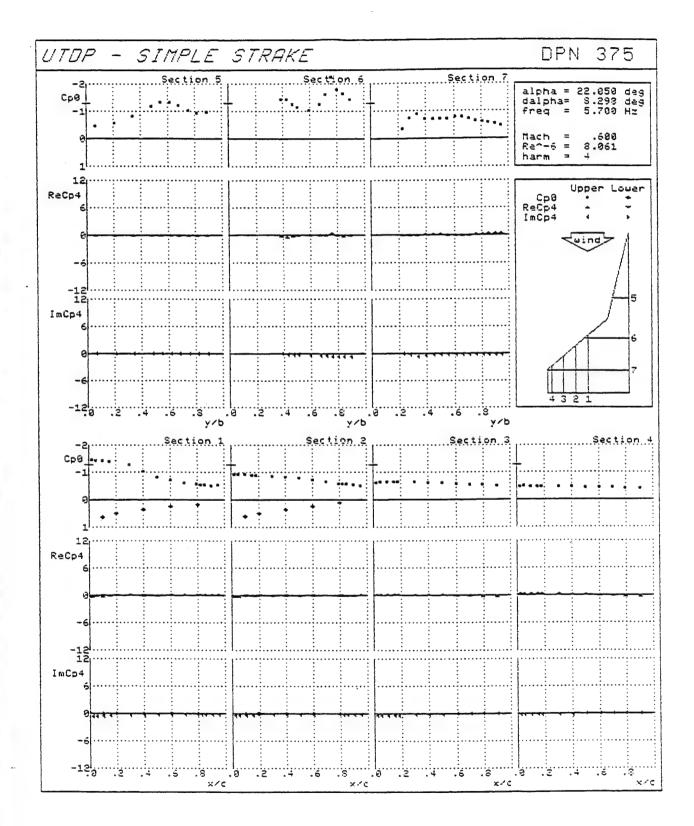


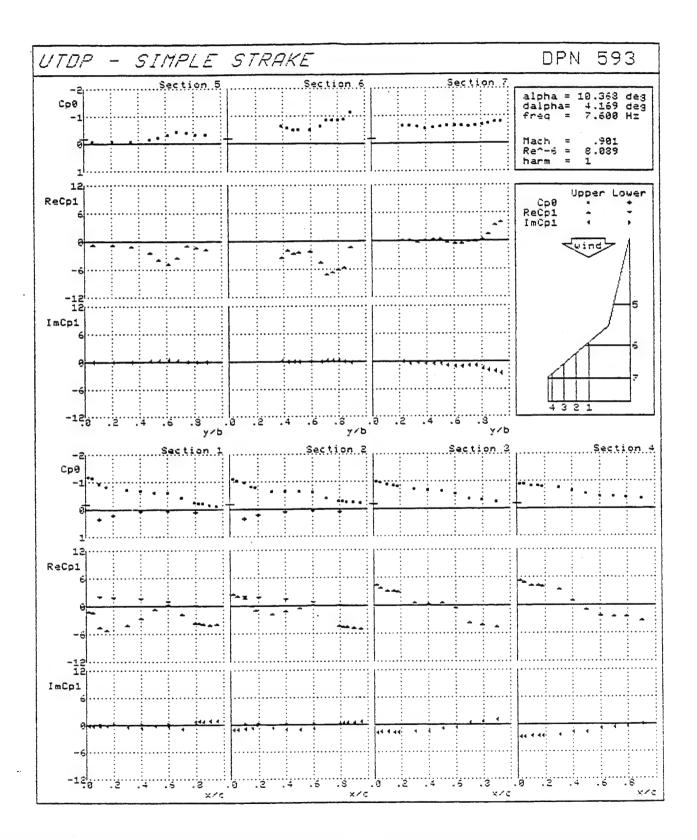


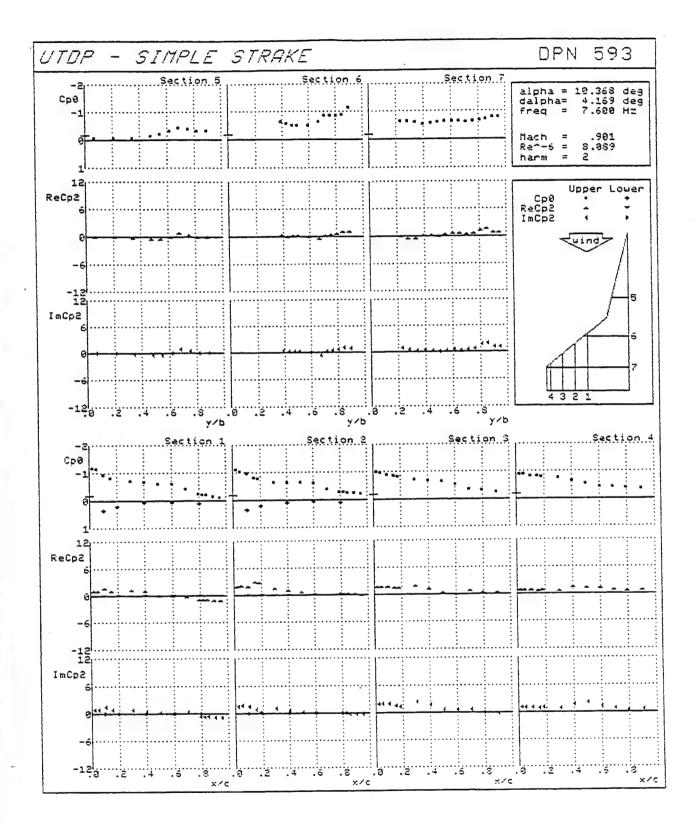


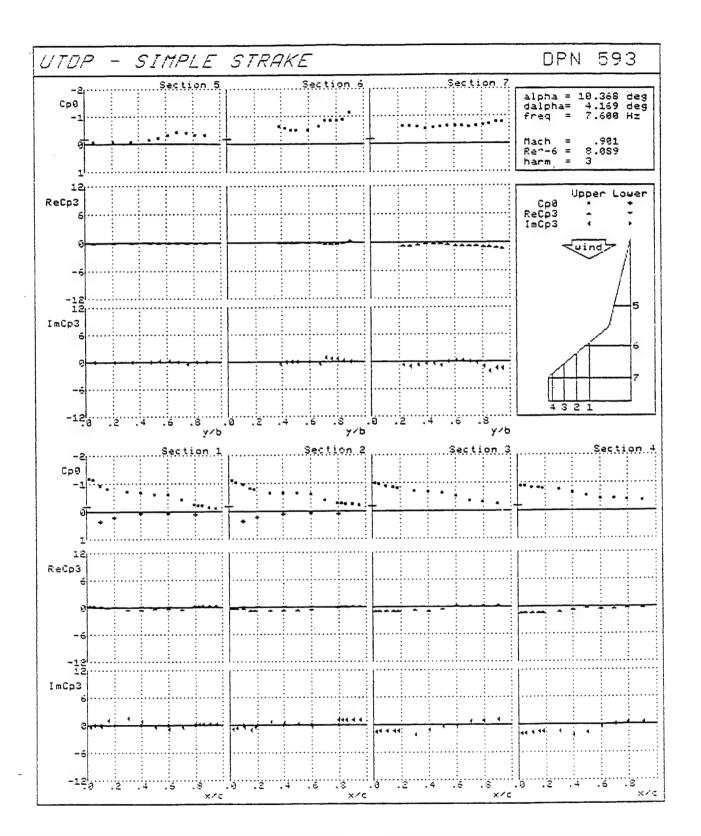


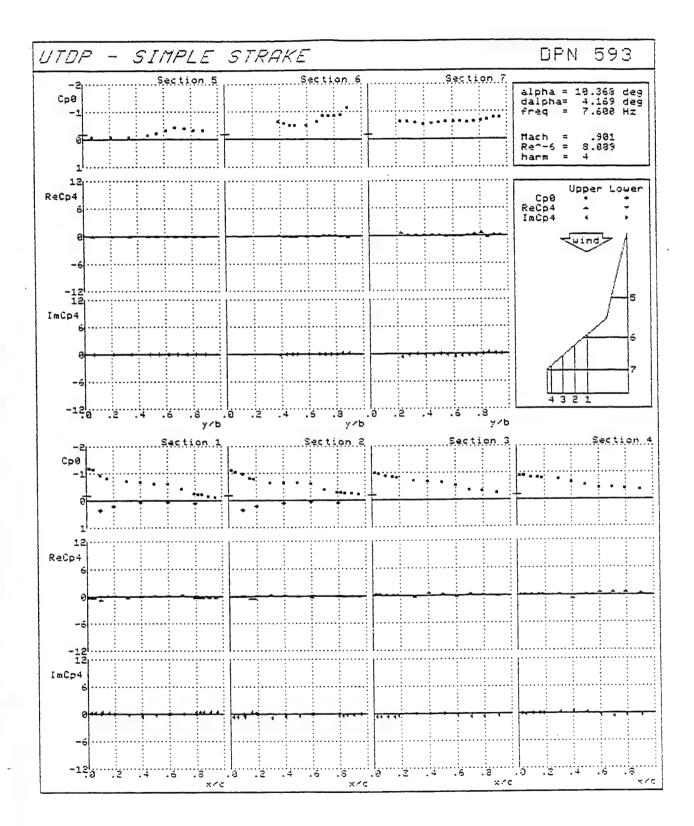


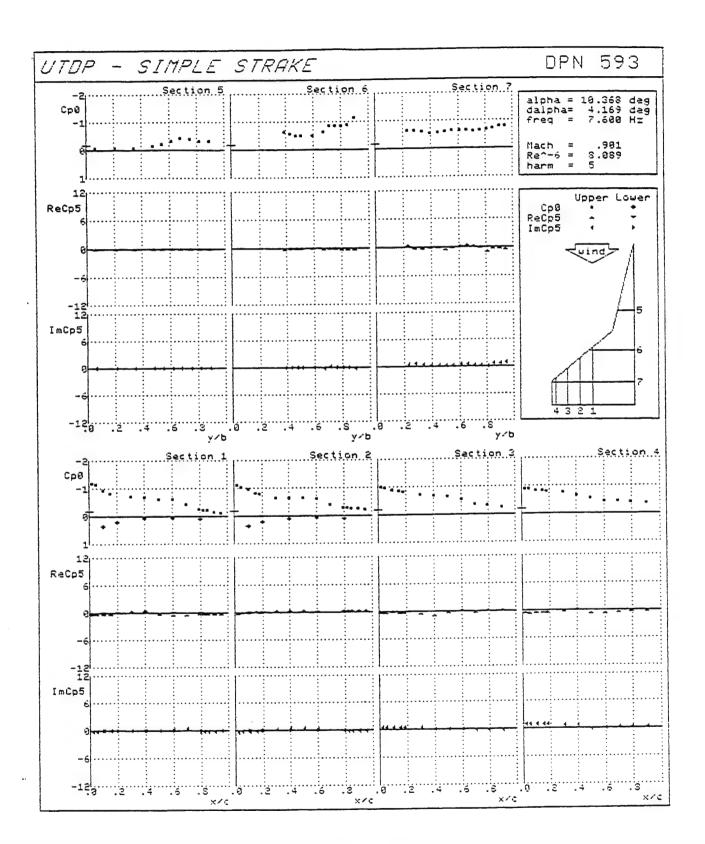


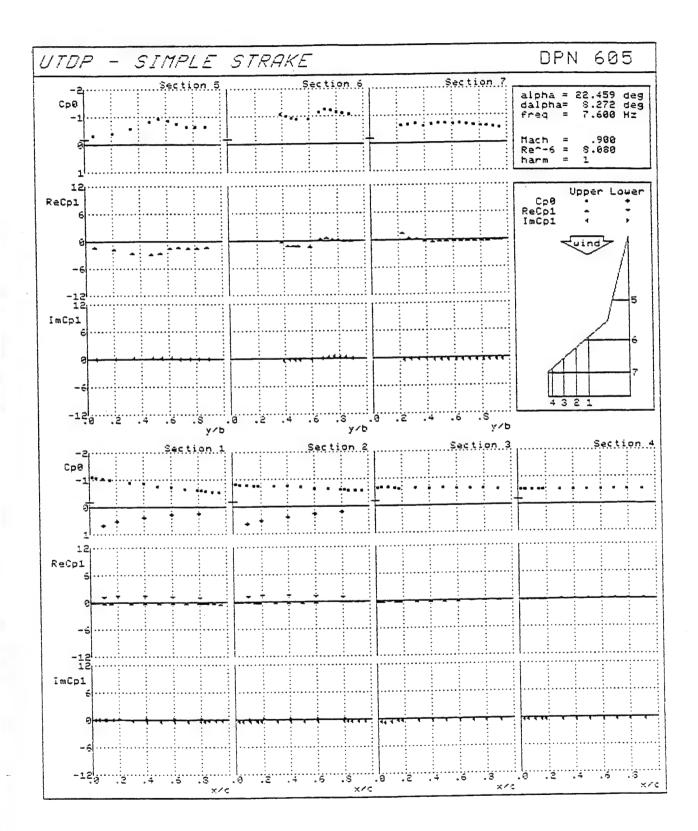


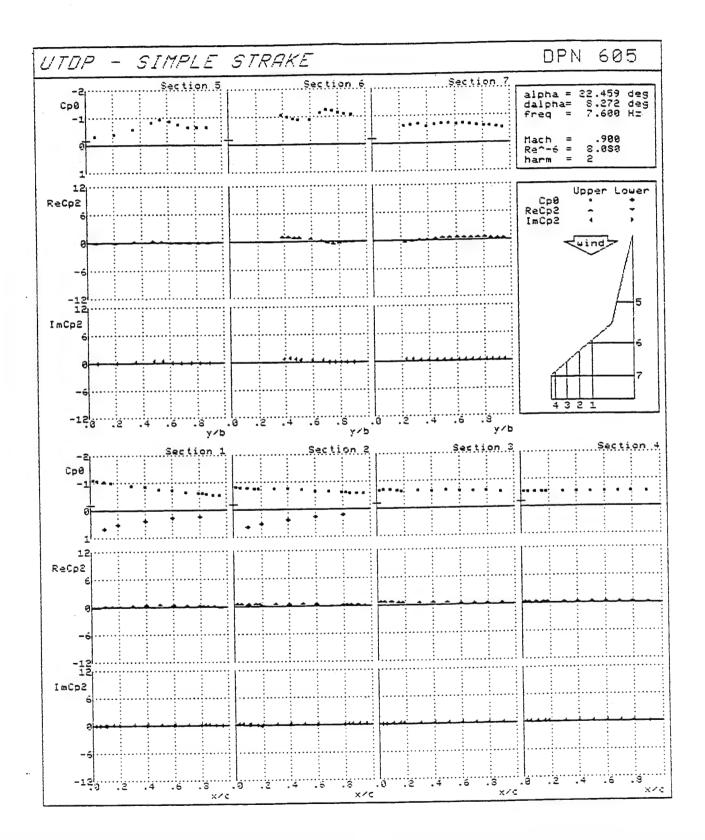


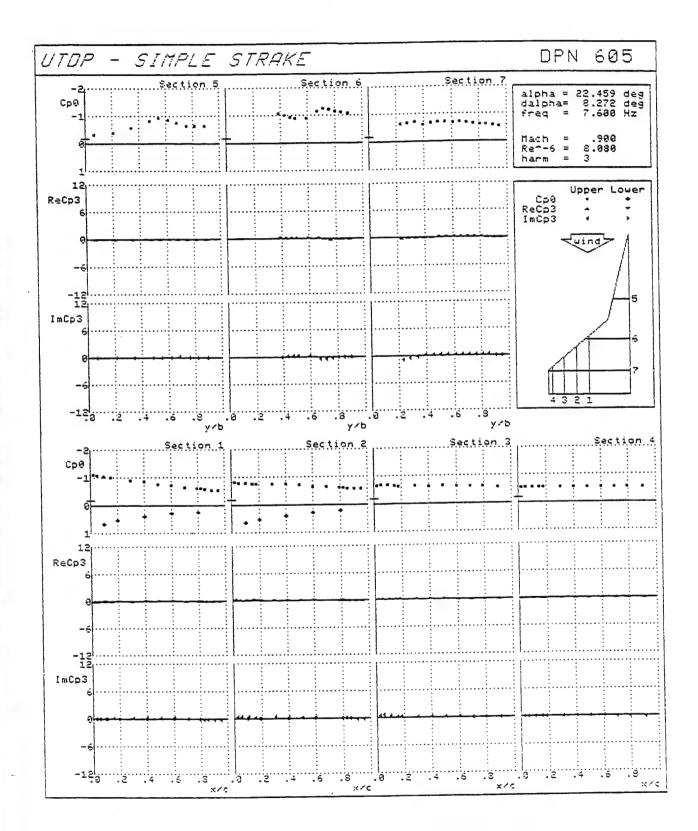


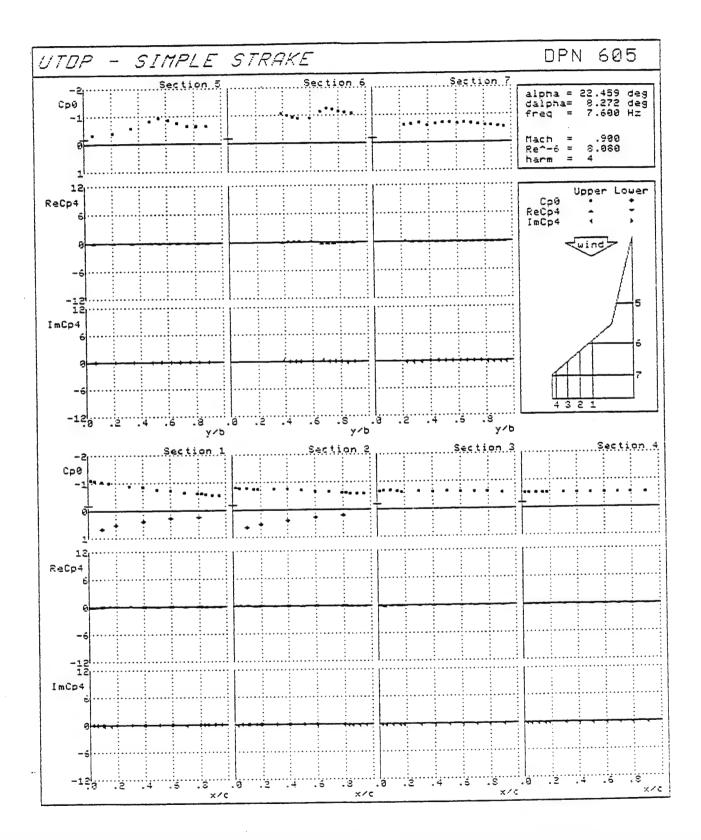


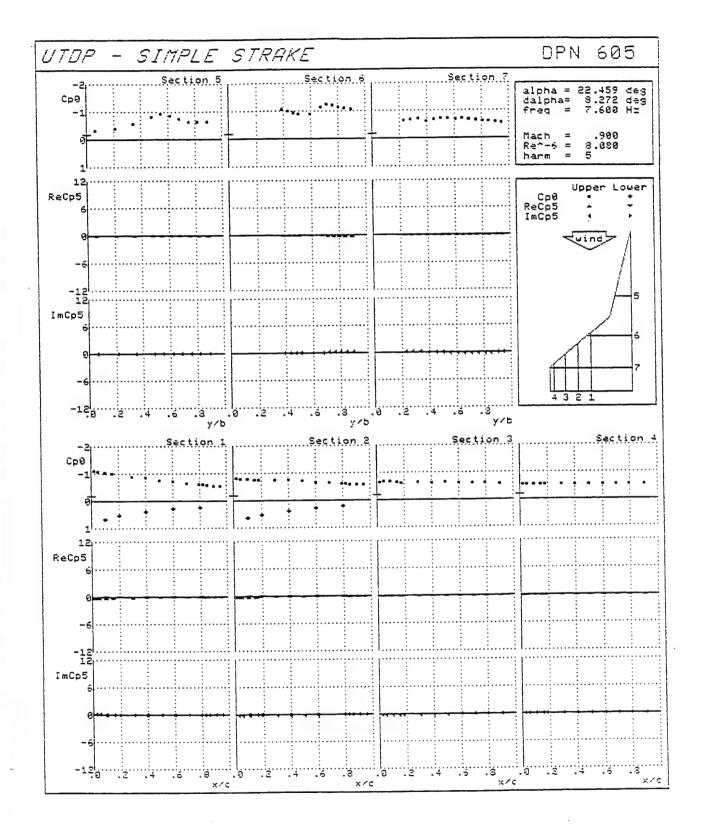












## APPENDIX B PLOTS OF NORMAL FORCE AND MOMENT COEFFICIENTS VERSUS INCIDENCE

## LIST OF FIGURES

		page
Figure B.1	CN mean versus incidence with (■) and without (♦) filler plate	105
Figure B.2	Cm mean versus incidence with (■) and without (♦) filler plate	105
_	CN u 5 mean versus incidence with (■) and	106
•	without (♦) filler plate	
Figure B.4	CN_u_6 mean versus incidence with (■) and	106
-	without (♦) filler plate	
Figure B.5	CN_u_7 mean versus incidence with (■) and	107
	without (♦) filler plate	
Figure B.6	CN mean versus incidence at	108
	Reynolds number of 3.8 ( $\blacksquare$ ) and 8.0 ( $\spadesuit$ ) $10^6$	
Figure B.7	Cm mean versus incidence at	108
	Reynolds number of 3.8 ( $\blacksquare$ ) and 8.0 ( $\spadesuit$ ) $10^6$	
Figure B.8	CN mean versus incidence at	109
	Mach number of 0.225 ( $\spadesuit$ ), 0.600 (+) and 0.900 (x)	
Figure B.9	Cm mean versus incidence at	109
	Mach number of 0.225 ( $\spadesuit$ ), 0.600 (+) and 0.900 (x)	
Figure B.10	Cm real versus incidence at a	110
	frequency of 7.6 ( $\blacksquare$ ), 11.4 ( $\spadesuit$ ) and 15.2 (+) Hz	
Figure B.11	Cm imaginary versus incidence at a	110
	frequency of 7.6 ( $\blacksquare$ ), 11.4 ( $\diamondsuit$ ) and 15.2 (+) Hz	
Figure B.12	Cm real versus incidence at an	111
	amplitude of 2.1 ( $\blacksquare$ ), 4.2 ( $\spadesuit$ ) and 8.4 (+) deg	
Figure B.13	Cm imaginary versus incidence at an	111
	amplitude of 2.1 ( $\blacksquare$ ), 4.2 ( $\spadesuit$ ) and 8.4 (+) deg	
Figure B.14	CN_u_6 mean versus incidence at a	112
	frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz	
Figure B.1:	CN_u_6 real versus incidence at a	113
	frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz	

## LIST OF FIGURES (Continued)

	· ·	age
Figure B.16	CN_u_6 imaginary versus incidence at a	113
	frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz	
Figure B.17	CN_u_6 mean versus incidence at an	114
	amplitude of 0.55 (■), 4.2 (♦) and 8.4 (+) deg	
Figure B.18	CN_u_6 real versus incidence at an	115
0	amplitude of 0.55 (■), 4.2 (♦) and 8.4 (+) deg	
Figure B.19	CN_u_6 imaginary versus incidence at an	115
	amplitude of 0.55 (■), 4.2 (♦) and 8.4 (+) deg	
Figure B.20	CN u 6 mean versus incidence at a	116
	Mach number of 0.225 (■; frequency = 5.7 Hz),	
	0.600 (♦; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz)	
	(this plot includes the three selected (see part I)	
	data points 151, 375 and 605)	
Figure B.21	CN_u_6 real versus incidence at a	117
	Mach number of 0.225 (■; frequency = 5.7 Hz),	
	0.600 (♦; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz)	
	(this plot includes the three selected (see part I)	
	data points 151, 375 and 605)	
Figure B.22	CN_u_6 imaginary versus incidence at a	117
	Mach number of 0.225 (■; frequency = 5.7 Hz),	
	0.600 ( $\blacklozenge$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz)	
	(this plot includes the three selected (see part I)	
	data points 151, 375 and 605)	
Figure B.23	CN_u_4 mean versus incidence at a	118
	Mach number of 0.225 (■; frequency = 5.7 Hz),	
	0.600 ( $\spadesuit$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz)	
	(this plot includes the two selected (see part I)	
	data points 358 and 593)	

## LIST OF FIGURES (Continued)

			Page
Figure	B.24	CN_u_4 real versus incidence at a	119
		Mach number of 0.225 (■; frequency = 5.7 Hz),	
		0.600 ( $\spadesuit$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz)	
		(this plot includes the two selected (see part I)	
		data points 358 and 593)	
Figure	B.25	CN_u_4 imaginary versus incidence at a	119
		Mach number of 0.225 (■; frequency = 5.7 Hz),	
		0.600 ( $\spadesuit$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz)	
		(this plot includes the two selected (see part I)	
		data points 358 and 593)	
Figure B.2	B.26	Cm mean versus incidence at	120
		Reynolds number of 8.0 (+) and 14.0 (x) 10 $^6$	

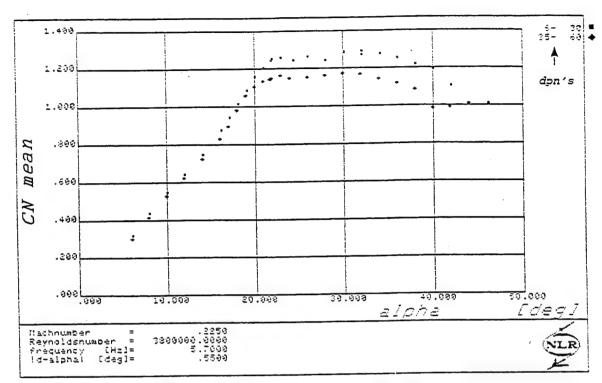


Figure B.1 CN mean versus incidence with (■) and without (♦) filler plate

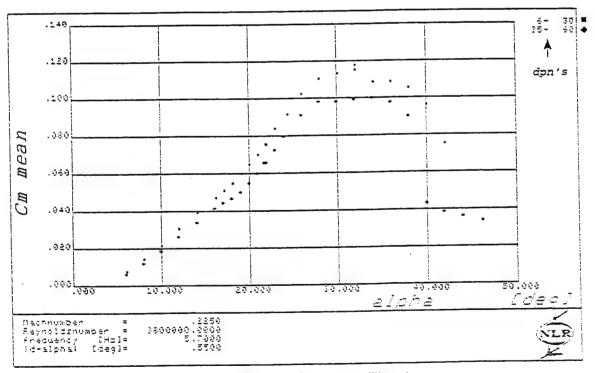


Figure B.2 Cm mean versus incidence with (■) and without (♦) filler plate

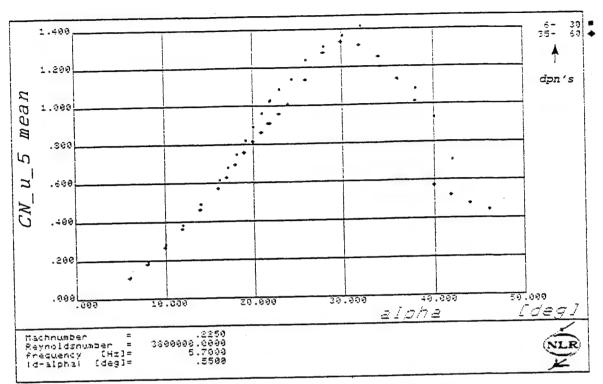


Figure B.3 CN\_u\_5 mean versus incidence with (■) and without (♦) filler plate

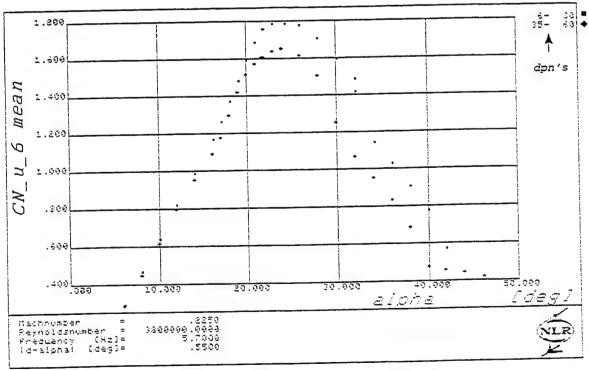


Figure B.4 CN\_u\_6 mean versus incidence with (■) and without (♦) filler plate

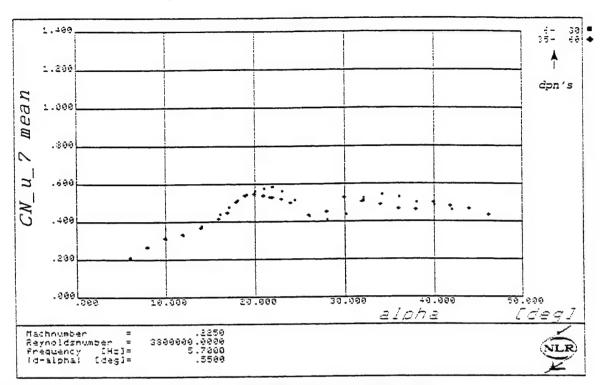


Figure B.5 CN\_u\_7 mean versus incidence with (■) without (♦) filler plate

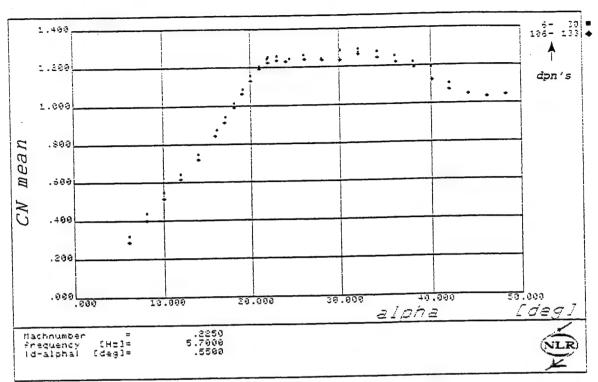


Figure B.6 CN mean versus incidence at Reynolds number of 3.8 (■) and 8.0 (♦) 10°

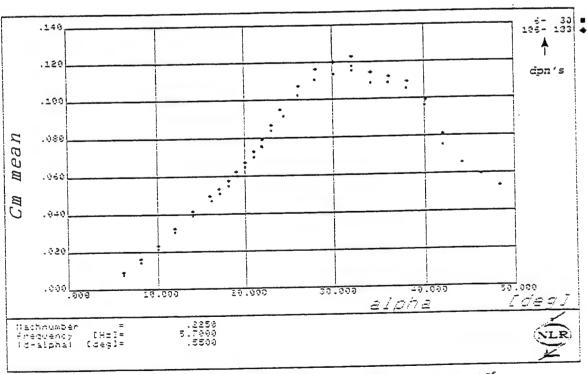


Figure B.7 Cm mean versus incidence at Reynolds number of 3.8 ( $\blacksquare$ ) and 8.0 ( $\spadesuit$ ) 10°

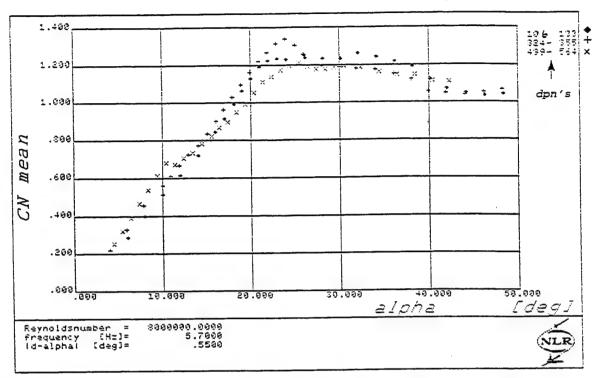


Figure B.8 CN mean versus incidence at Mach number of 0.225 (♦), 0.600 (+) and 0.900 (x)

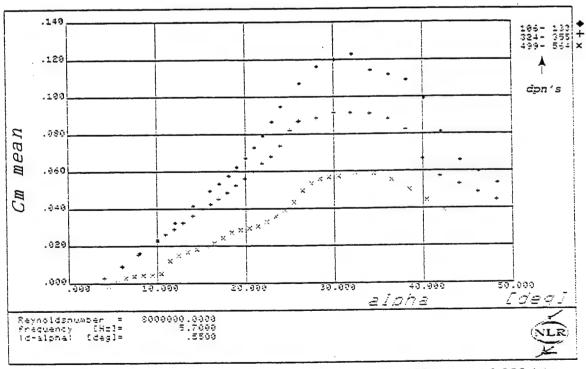


Figure B.9 Cm mean versus incidence at Mach number of 0.225 (♦), 0.600 (+) and 0.900 (x)

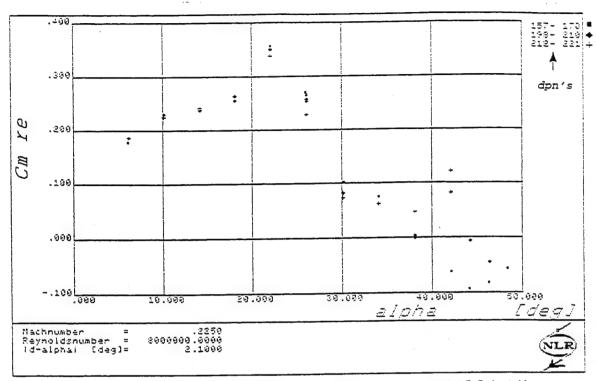


Figure B.10 Cm real versus incidence at a frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz

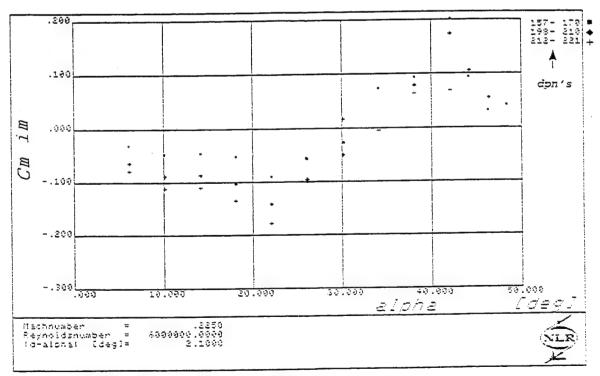


Figure B.11 Cm imaginary versus incidence at a frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz

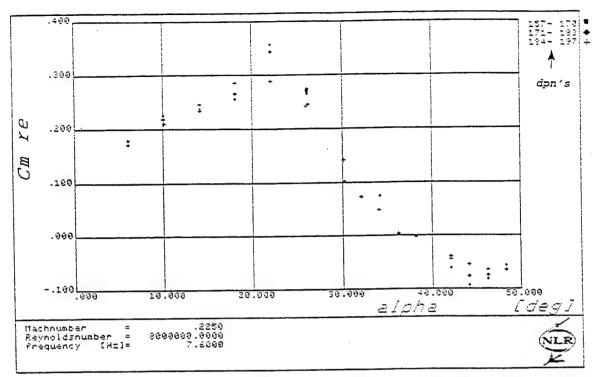


Figure B.12 Cm real versus incidence at an amplitude of 2.1 (■), 4.2 (♦) and 8.4 (+) deg

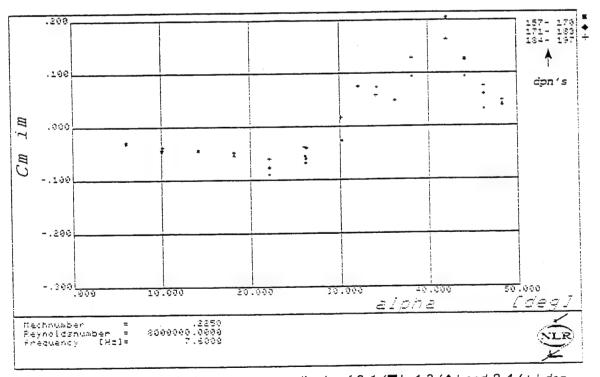


Figure B.13 Cm imaginary versus incidence at an amplitude of 2.1 (■), 4.2 (♦) and 8.4 (+) deg

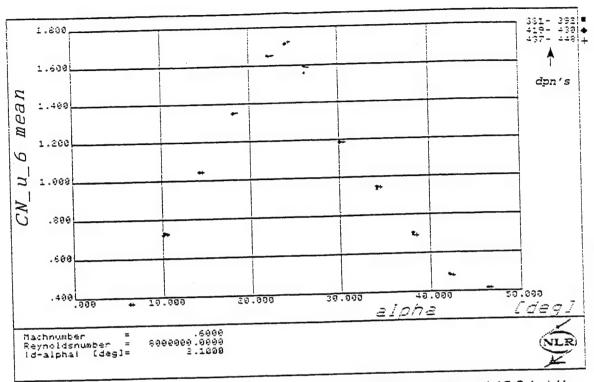


Figure B.14 CN\_u\_6 mean versus incidence at a frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz

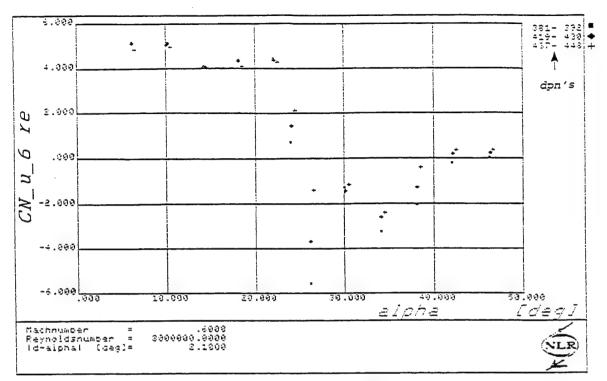


Figure B.15 CN\_u\_6 real versus incidence at a frequency of 7.6 (■), 11.4 (♦) and 15.2 (+) Hz

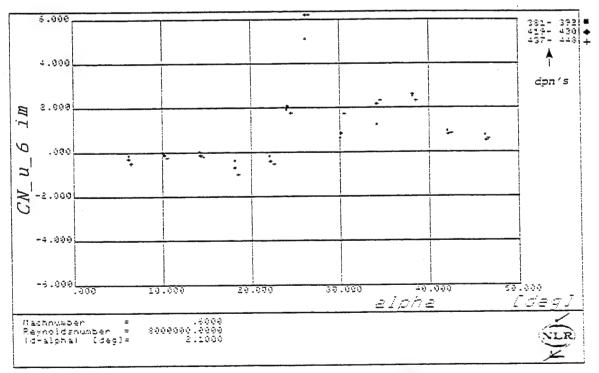


Figure B.16 CN\_u\_6 imaginary versus incidence at a frequency of 7.6 ( $\blacksquare$ ), 11.4 ( $\spadesuit$ ) and 15.2 (+) Hz

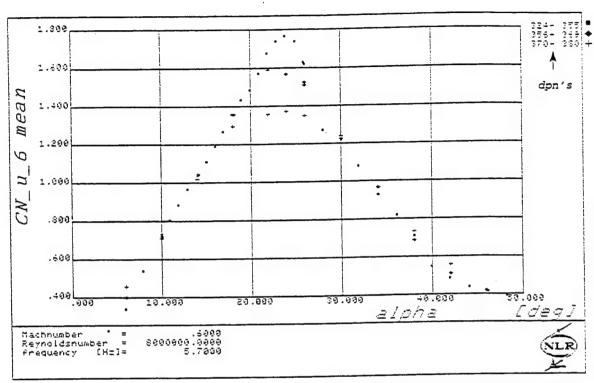


Figure B.17 CN\_u\_6 mean versus incidence at an amplitude of 0.55 (■), 4.2 (♦) and 8.4 (+) deg

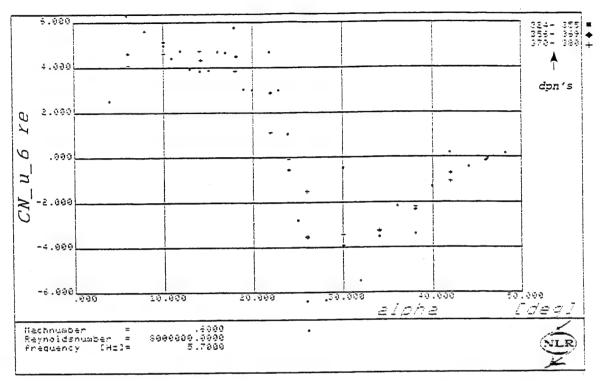


Figure B.18 CN\_u\_6 real versus incidence at an amplitude of 0.55 (■), 4.2 (♦) and 8.4 (+) deg

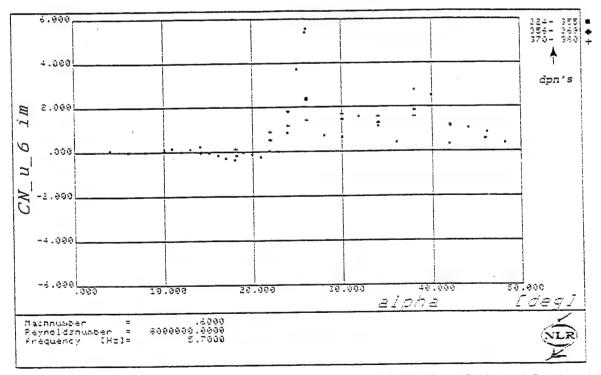


Figure B.19 CN\_u\_6 imaginary versus incidence at an amplitude of 0.55 ( $\blacksquare$ ), 4.2 ( $\spadesuit$ ) and 8.4 (+) deg

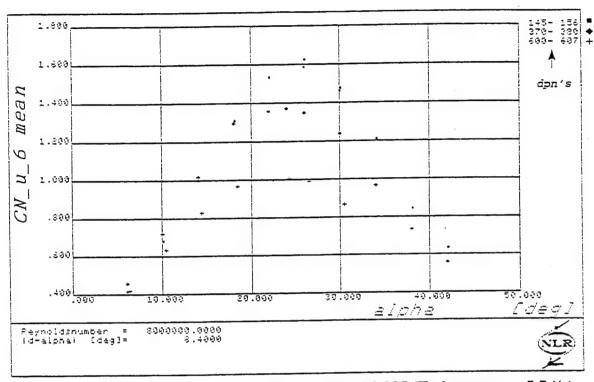


Figure 8.20 CN\_u\_6 mean versus incidence at a Mach number of 0.225 ( $\blacksquare$ ; frequency = 5.7 Hz), 0.600 ( $\spadesuit$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz) (this plot includes the three selected (see part I) data points 151, 375 and 605)

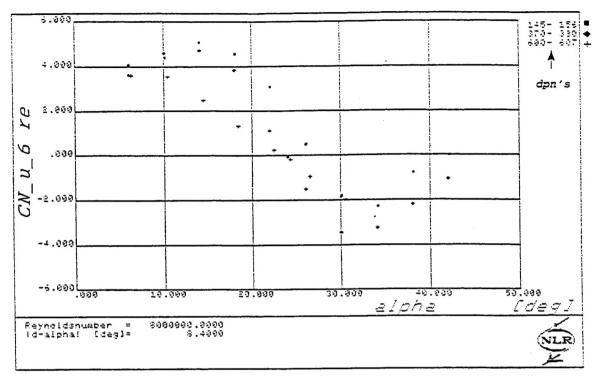


Figure B.21  $CN_u_6$  real versus incidence at a Mach number of 0.225 ( $\blacksquare$ ; frequency = 5.7 Hz), 0.600 ( $\blacklozenge$ ; frequency = 5.7 Hz) and 0.900 ( $\dotplus$ ; frequency = 7.6 Hz) (this plot includes the three selected (see part I) data points 151, 375 and 605)

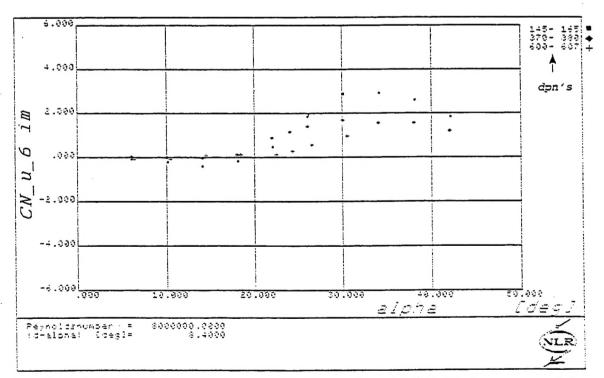


Figure B.22  $CN_u_6$  imaginary versus incidence at a Mach number of 0.225 ( $\blacksquare$ ; frequency = 5.7 Hz), 0.600 ( $\spadesuit$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz) (this plot includes the three selected (see part I) data points 151, 375 and 605)

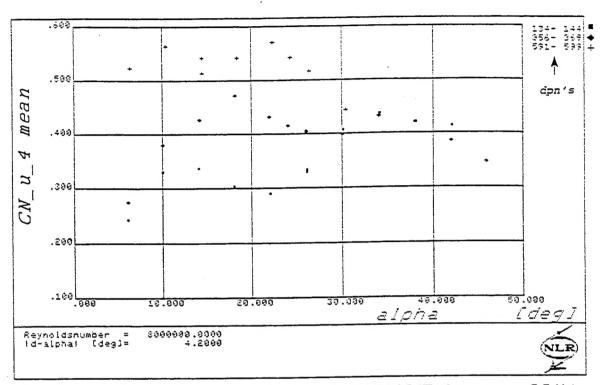


Figure B.23 CN\_u\_4 mean versus incidence at a Mach number of 0.225 ( $\blacksquare$ ; frequency = 5.7 Hz), 0.600 ( $\blacklozenge$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz) (this plot includes the two selected (see part I) data points 358 and 593)

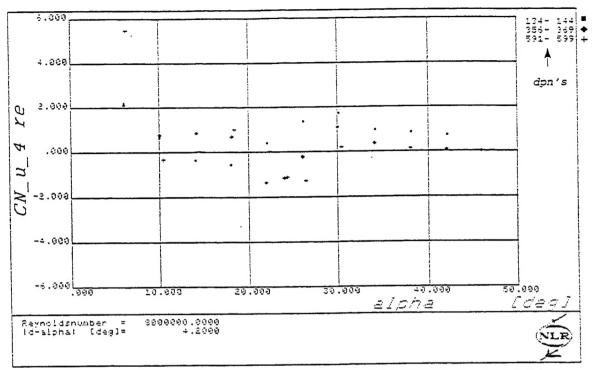


Figure 8.24  $CN_u_4$  real versus incidence at a Mach number of 0.225 ( $\blacksquare$ ; frequency = 5.7 Hz), 0.600 ( $\blacklozenge$ ; frequency = 5.7 Hz) and 0.900 ( $\dotplus$ ; frequency = 7.6 Hz)(this plot includes the two selected (see part I) data points 358 and 593)

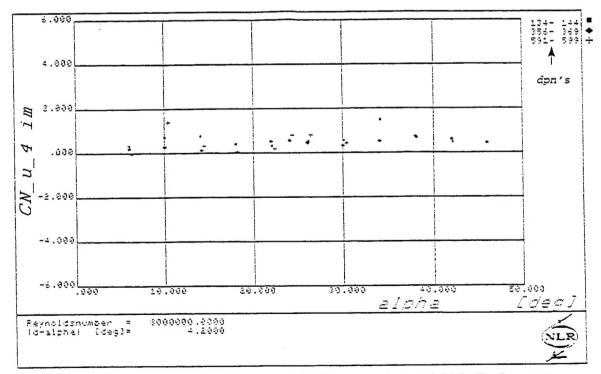


Figure B.25  $CN_u_4$  imaginary versus incidence at a123 Mach number of 0.225 ( $\blacksquare$ ; frequency = 5.7 Hz), 0.600 ( $\blacklozenge$ ; frequency = 5.7 Hz) and 0.900 (+; frequency = 7.6 Hz) (this plot includes the two selected (see part I) data points 358 and 593)

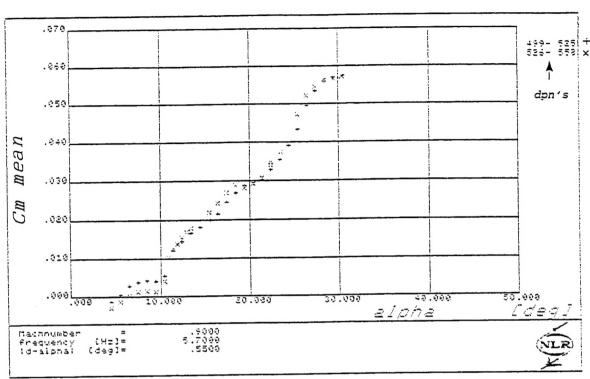


Figure B.26 Cm mean versus incidence at Reynolds number of 8.0 (+) and 14.0 (x) 10  $^{6}$